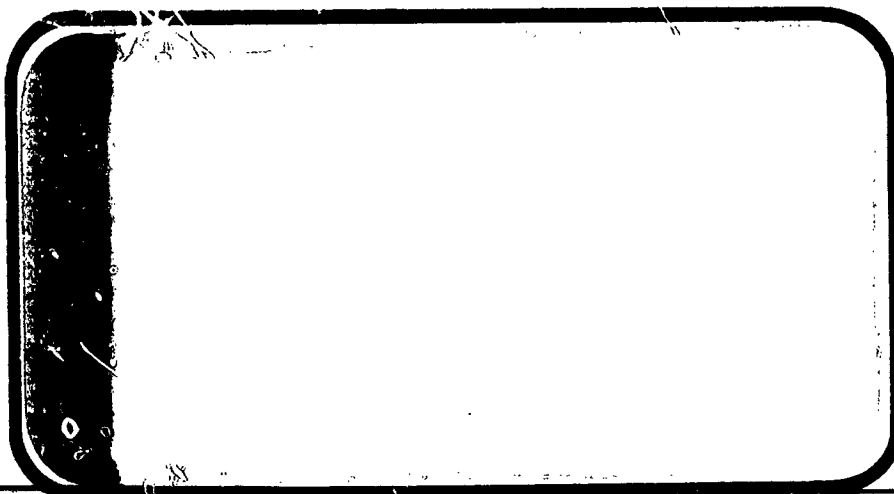


**NATIONAL AERONAUTICS AND SPACE ADMINISTRATION**



(NASA-CR-128795) WIND TUNNEL TEST OF  
THE 0.010-SCALE SPACE SHUTTLE INTEGRATED  
VEHICLE IN THE NASA-AMES 3.5 FOOT  
HYPERSONIC WIND TUNNEL (IA10) (Chrysler  
Corp.) 138 p HC \$9.00 CSCL 22B

N74-16558

Unclas  
28898

G3/31



**SPACE SHUTTLE**

**AEROTHERMODYNAMIC DATA REPORT**

**JOHNSON SPACE CENTER**

**HOUSTON, TEXAS**

**DATA MAN**agement services

SPACE DIVISION



**CHRYSLER  
CORPORATION**

December, 1973

DMS-DR-2078  
NASA CR-128,795

WIND TUNNEL TEST OF THE 0.010-SCALE  
SPACE SHUTTLE INTEGRATED VEHICLE IN THE  
NASA-AMES 3.5-FOOT HYPERSONIC WIND TUNNEL (IA10)

By

F. F. Fitzgerald and M. T. Petrozzi, Rockwell International  
J. W. Cleary and J. A. Mellenthin, NASA Ames

By

Data Management Services  
Chrysler Corporation Space Division  
New Orleans, La. 70189

for

Engineering Analysis Division  
Johnson Space Center  
National Aeronautics and Space Administration  
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: ARC 3.5-169  
NASA Series No.: IA10  
Test Dates: 1 thru 3 August 1973

FACILITY COORDINATOR:

C. R. Nysmith  
Mail Stop 229-5  
Ames Research Center  
Moffett Field, California 94035  
Phone: (415) 965-5274

PROJECT ENGINEERS:

F. F. Fitzgerald, M. T. Petrozzi  
Rockwell International Space Div.  
12214 Lakewood Blvd.  
Mail Code AC07  
Downey, California 90241  
Phone: (213) 922-4898

J. W. Cleary, J. A. Mellenthin  
NASA/Ames Research Center  
Mail Stop 229-1  
Moffett Field, California 94035  
Phone: (415) 965-6211

DATA MANAGEMENT SERVICES:

This document has been prepared by:

D. A. Sarver

M. J. Lanfranco  
Liaison Operations

B. W. Myers  
Data Operations

This document has been reviewed and is approved for release.

for N. D. Kemp  
Data Management Services

Chrysler Corporation Space Division assumes no responsibility for  
the data presented herein other than its display characteristics.

WIND TUNNEL TEST OF THE 0.010-SCALE  
SPACE SHUTTLE INTEGRATED VEHICLE IN THE  
NASA-AMES 3.5-FOOT HYPERSONIC WIND TUNNEL (1A10)

By

F. F. Fitzgerald and M. T. Petrozzi/Rockwell International  
J. W. Cleary and J. A. Mellenthin/NASA Ames

ABSTRACT

Experimental aerodynamic investigations were conducted in the NASA Ames Research Center 3.5-Foot Hypersonic Wind Tunnel from August 1, 1973 to August 3, 1973 on a 0.010-scale model of the Space Shuttle Vehicle orbiter and external tank (model no. 32 OT).

The purpose of the test was to evaluate the basic hypersonic stability characteristics of the external tank and orbiter and to define orbiter plume effects on aero characteristics using solid plumes.

The test was conducted at angles of attack from  $-10^{\circ}$  to  $30^{\circ}$  and angles of sideslip of  $-10^{\circ}$  thru  $10^{\circ}$ . Six component force data and static base pressures were recorded during the test.

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### COEFFICIENTS SCHEDULE:

A: CA, CN, CIM, vs. ALPHA

CN vs. CIM

B: CY, CBL, CYN vs. BETA

CY vs. CYN

C: CY, CBL, CYN vs. ALPHA

D: CN/A, CIM/A, XAC/L vs. ALPHA

DCY/DA, DCBLDA, DCYNDA, DCN/DA, DCIMDA, DCA/DA vs. ALPHA

E: DCY/DB, DCBLDB, DCYNDB vs. ALPHA

NOMENCLATURE  
General

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
$C_p$	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; $V/a$
p		pressure; $N/m^2$ , psf
q	Q(NSM)	dynamic pressure; $1/2\rho V^2$ , $N/m^2$ , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
V		velocity; m/sec, ft/sec
$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\psi$	PSI	angle of yaw, degrees
$\phi$	PHI	angle of roll, degrees
$\rho$		mass density; $kg/m^3$ , slugs/ft <sup>3</sup>

Reference & C.G. Definitions

c.g.		center of gravity
$l_{REF}$	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m <sup>2</sup> , ft <sup>2</sup>
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

## NOMENCLATURE (Continued)

### SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
	free stream

### Body-Axis System

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_N$	CN	normal-force coefficient; normal force/qS
$C_A$	CA	axial-force coefficient; axial force/qS
$C_Y$	CY	side-force coefficient; side force/qS
$C_m$	CLM	pitching-moment coefficient; pitching moment/qS $l$
$C_n$	CYN	yawing-moment coefficient; yawing moment/qS $l$
$C_l$	CBL	rolling-moment coefficient; rolling moment/qS $l$

### Stability-Axis System

$C_L$	CL	lift coefficient; lift/qS
$C_D$	CD	drag coefficient; drag/qS
L/D	L/D	lift-to-drag ratio; $C_L/C_D$

### ADDITIONS TO NOMENCLATURE

$C_{N_{\alpha_{local}}}$	CN/A	local normal force coefficient derivative with alpha; per degree.
$C_{m_{\alpha_{local}}}$	CLM/A	local pitching moment coefficient derivative with alpha; per degree.
	XAC/L	local longitudinal center of pressure.

# NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{Y\delta a}$	DCY/DA	side force coefficient derivative with respect to total aileron deflection. Algebraic difference of the side force coefficients of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; per degree.
$C_{l\delta a}$	DCBLDA	rolling moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree.
$C_{n\delta a}$	DCYNDA	yawing moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree
$C_{N\delta a}$	DCN/DA	normal force coefficient derivative with respect to aileron deflection. Algebraic difference of normal coefficient of two runs divided by the algebraic difference of the total aileron deflection of the runs; body axis system; per degree.
$C_{m\delta a}$	DCIMDA	pitching moment coefficient derivative with respect to aileron deflection. Algebraic difference of pitching moment coefficient of two runs divided by the algebraic difference of the total aileron deflection of the runs; body axis system; per degree.
$C_{A\delta a}$	DCA/DA	axial force coefficient derivative with respect to aileron deflection. Algebraic difference of axial force of two runs divided by the algebraic difference of total aileron deflection of the runs; body axis system; per degree.
$C_{Y\beta}$	DCY/DB	side force coefficient derivative with respect to beta. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; per degree.

# NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{l_{\beta}}$	DCBLDB	rolling moment coefficient derivative with respect to beta. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; body axis system; per degree.
$C_{n_{\beta r}}$	DCYNDB	yawing moment coefficient derivative with respect to beta. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; per degree.
$C_{N_{\delta r}}$	DCN/DR	normal force coefficient derivative with respect to rudder deflection. Algebraic difference of normal force coefficient of two runs divided by algebraic difference of rudder deflection of two runs; per degree.
$C_{Y_{\delta r}}$	DCY/DR	side force coefficient derivative with respect to rudder deflection. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{n_{\delta r}}$	DCYNDR	yawing moment coefficient derivative with respect to rudder deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{l_{\delta r}}$	DCBLDR	rolling moment coefficient derivative with respect to rudder deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{A_{\delta r}}$	DCA/DR	axial force coefficient derivative with respect to rudder deflection. Algebraic difference of axial force coefficient of two runs divided by the algebraic difference of rudder deflection of two runs; per degree.

# NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$C_{m_{\delta_r}}$	DCLMDR	pitching moment coefficient derivative with respect to rudder deflection. Algebraic difference of pitching moment coefficient of two runs divided by algebraic difference of rudder deflection of two runs; per degree.
	YAC/L	lateral center of pressure. Yawing moment coefficient derivative divided by side force coefficient derivative.
$C_{A_{\alpha=0}}$	CAAFO	axial force coefficient at zero angle of attack ( $\alpha = 0$ ).
$C_{N_{\alpha=0}}$	CNAFO	normal force coefficient at zero angle of attack ( $\alpha = 0$ ).
$C_{m_{\alpha=0}}$	CLMAFO	pitching moment coefficient at zero angle of attack ( $\alpha = 0$ ).
$\delta_{eL}$		left elevon surface deflection angle, positive deflection trailing edge down; degrees.
$\delta_{eR}$		right elevon surface deflection angle, positive deflection trailing edge down; degrees.
$\delta_e$	ELEVON	elevon, surface deflection angle, positive deflection trailing edge down; degrees, $(\delta_{eL} + \delta_{eR})/2$ .
$\delta_a$	AILRON	aileron, surface deflection angle, positive deflection trailing edge down; degrees $(\delta_{eL} + \delta_{eR})/2$ .
$\delta_r$	RUDDER	rudder, surface deflection angle, positive deflection trailing edge to the left; degrees.
$A_{BC}$		external tank balance cavity area, 3.999 in <sup>2</sup> .
$A_{ET}$		external tank base area, 4.255 in <sup>2</sup> model scale.

# NOMENCLATURE (Concluded)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$A_{ME}$		orbiter main engine nozzle base area, $2.081 \text{ in}^2$ model scale.
$A_{OB}$		orbiter base area, $2.010 \text{ in}^2$ model scale.
$A_{OM}$		OMS pod base area, $1.362 \text{ in}^2$ model scale.
$C_{A_b}$	CAB	orbiter base axial-force coefficient.
$C_{A_b ET}$	CABET	external tank base axial-force coefficient.
$C_{PBC}$	CPBC	external tank balance cavity pressure coefficient.
$C_{PET}$	CPET	external tank base pressure coefficient.
$C_{POB}$	CPOB	orbiter base pressure coefficient.
$C_{POM}$	CPOM	OMS pod base pressure coefficient.
$C_{PME}$	CPME	main engine nozzle base pressure coefficient.

## CONFIGURATIONS INVESTIGATED

The model for this test was an 0.010-scale representation of the Space Shuttle Vehicle configuration 3 Space Shuttle (model no. 32 OT). The model consisted of the orbiter, external tank and simulated engine plumes. The model was constructed of stainless steel except the orbiter body which was constructed of aluminum.

A six component 1.5-inch MK II balance was mounted in the external tank.

The model was tested with and without non-metric main engine solid exhaust plume simulators. Figure 2(c) shows the model with plume simulation installed.

The various model components tested are listed below:

$O_9 = B_{19} C_7 E_{23} F_5 M_4 N_{24} N_8 R_5 V_7 W_{107}$

$B_{19}$  = VL70 - 000139B (lines) body

$C_7$  = VL70 - 000139B (lines) canopy

$E_{23}$  = VL70 - 000139B (lines) elevon

$F_5$  = VL70 - 000139B (lines) body flap

$M_4$  = VL70 - 000139B (lines) OMS pod

$N_{24}$  = VL70 - 000140A (lines) main engine nozzle

$N_8$  = VL70 - 000140A (lines) OMS nozzle

$R_5$  = VL70 - 000139B and VL70 - 000095 rudder

$V_7$  = VL70 - 000139B (lines) vertical tail



$W_{107}$  = VL70 - 000139B (lines) wing

$T_{10}$  = VL78 - 000041 (lines) external tank

$AT_2$  = VL72 - 000089 (lines) attach structure

Base pressure data acquired during runs 1 thru 6 are invalid. Tygon tubing, used in measuring these pressures was melted by the tunnel heat during these runs. Prior to run five two external pressure tubes were added. The location of these pressure tubes is shown in figure 2(b).

Pressure tap #16 has an area value of  $4.378 \text{ in}^2$  and pressure tap #17 has an area value of  $2.010 \text{ in}^2$ . Subsequent to run 006 tygon tubing was replaced, in the orbiter, with a high temperature resistant flexible tubing. With the exception of tap #4 which was inoperative, the data acquired from that time to the end of the run series are considered reliable.

## TEST FACILITY DESCRIPTION

The NASA-Ames 3.5-Foot Hypersonic Wind Tunnel is a closed-circuit, blowdown-type tunnel capable of operating at nominal Mach numbers of 5, 7, and 10 at pressures to 1800 psia and temperatures to 3400°R for run times to four minutes. The major components of the facility include a gas storage system where the test gas is stored at 3000 psi, a storage heater filled with aluminum-oxide pebbles capable of heating the test gas to 3400°R, axisymmetric contoured nozzles with exit diameters of 42 inches for generating the desired Mach number, and a 900,000 ft<sup>3</sup> vacuum storage system which operates to pressures of 0.3 psia. The test section itself is an open-jet type enclosed within a chamber approximately 12-feet in diameter and 40-feet in length, arranged transversally to the flow direction.

A model support system is provided that can pitch models through an angle-of-attack range of -20 to +18 degrees, in a vertical plane, about a fixed point of rotation on the tunnel centerline. This rotation point is adjustable from 1 to 5 feet from the nozzle exit plane. The model normally is out of the test stream (strut centerline 37-inches from tunnel centerline) until the tunnel test conditions are established after which it is inserted. Insertion time is adjustable to as little as 1/2 second and models may be inserted at any strut angle.

A high-speed, analog-to-digital data acquisition system is used to record test data on magnetic tape. The present system is equipped to measure and record the outputs from 80 transducers in addition to 20 channels of tunnel parameters.

## DATA REDUCTION

The aerodynamic force data presented were measured by the Task Corporation 1.5- inch MK II strain gage balance and the moment data were transferred to the external tank centerline at a point 6.80 in (model scale) from the tank nose.

Base pressure axial-force coefficients were calculated for the individual regions as follows:

### 1. Orbiter Base Axial Force

$$C_{A_b} = - (A_{ME} C_{P_{ME}} + A_{OB} C_{P_{OB}} + A_{OM} C_{P_{OM}}) / S$$

### 2. External Tank Base Axial Force

$$C_{A_{b_{ET}}} = - (A_{ET} C_{P_{ET}} + A_{BC} C_{P_{BC}}) / S$$

where S is the theoretical wing area (0.269 ft<sup>2</sup> model scale).

These base pressure axial-force coefficients were calculated and plotted but were not used to correct the measured axial-force coefficient data.

The following reference dimensions were used for data reduction:

Symbol	Description	Value (model scale)	Applicable Data
$A_{ME}$	Orbiter main engine nozzle base area	2.081 in <sup>2</sup>	Pressure tap nos. 3 and 4
$A_{OB}$	Orbiter base area	2.010 in <sup>2</sup>	Pressure tap nos. 5 and 6 or 17
$A_{OM}$	OMS pod base area	1.362 in <sup>2</sup>	Pressure tap nos. 1 and 2

# DATA REDUCTION (Concluded)

<u>Symbol</u>	<u>Description</u>	<u>Value (model scale)</u>	<u>Applicable Data</u>
$A_{ET}$	ET base area	4.255 in <sup>2</sup>	Pressure tap nos. 7 and 8 or 16
$A_{BC}$	ET balance cavity area	3.990 in <sup>2</sup>	Pressure tap no. 15
S	Reference area	0.269 ft <sup>2</sup>	Force and moment data
$l_r$	Reference length	12.90 in	Moment data

Also see discussion in section "CONFIGURATIONS INVESTIGATED" concerning the use of external pressure taps 16 and 17.

**TABLE 1.**

[illegible]

TABLE 2.

[illegible]

TABLE 3. - MODEL COMPONENT DESCRIPTIONS

MODEL COMPONENT: BODY - B19

GENERAL DESCRIPTION: Fuselage, Configuration 3, per Rockwell Lines  
VL70-000139B.

NOTE: Identical to B17 except forebody.

Model Scale = .010

DRAWING NUMBER: VL70-000139B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length - IN.	<u>1290.3</u>	<u>12.903</u>
Max. Width - IN.	<u>267.6</u>	<u>2.676</u>
Max. Depth - IN.	<u>244.5</u>	<u>2.445</u>
Fineness Ratio	<u>4.82175</u>	<u>4.82175</u>
Area - FT <sup>2</sup>		
Max. Cross-Sectional	<u>386.67</u>	<u>0.03867</u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE 3. - Continued.

MODEL COMPONENT: Canopy - C7

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

Model Scale = .010

DRAWING NUMBER VL70-000139

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length ( $X_0 = 433$ to $X_0 = 670$ ) - in. FS	<u>237</u>	<u>2.37</u>
Max Width	<u>                    </u>	<u>                    </u>
Max Depth ( $Z_0 =$ to $Z_0 = 501$ ) - in FS	<u>                    </u>	<u>                    </u>
Fineness Ratio	<u>                    </u>	<u>                    </u>
Area		
Max Cross-Sectional	<u>                    </u>	<u>                    </u>
Planform	<u>                    </u>	<u>                    </u>
Wetted	<u>                    </u>	<u>                    </u>
Base	<u>                    </u>	<u>                    </u>



TABLE 3. - Continued.

MODEL COMPONENT: ELEVON - E23GENERAL DESCRIPTION: Configuration 3 per W107 Rockwell LinesVL70-000139B, data for (1) of (2) sidesModel Scale = .010DRAWING NUMBER: VL70-000139B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>205.52</u>	<u>0.02055</u>
Span (equivalent) - IN.	<u>353.34</u>	<u>3.5334</u>
Inb'd equivalent chord	<u>114.78</u>	<u>1.1478</u>
Outb'd equivalent chord	<u>55.00</u>	<u>0.550</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.208</u>	<u>.208</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.24</u>	<u>-10.24</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line) - FT <sup>3</sup>	<u>1548.07</u>	<u>0.00155</u>
Product of Area Moment		

TABLE 3. - Continued.

MODEL COMPONENT: F5 Body Flap

GENERAL DESCRIPTION: 3 Configuration per Rockwell Lines VL70-000139

Scale Model = .010

DRAWING NUMBER

VL70-000139

DIMENSION:

FULL SCALE

MODEL SCALE

Length - in

84.70

0.8470

Max Width - in

267.6

2.676

Max Depth

Fineness Ratio

Area - Ft<sup>2</sup>

Max Cross-Sectional

Planform

Wetted

Base

142.00

0.0142

38.0958

0.00381

TABLE 3. - Continued.

MODEL COMPONENT: OVS Pod - M<sub>4</sub>

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

NOTE: M<sub>4</sub> identical to M<sub>3</sub>, except intersection to fuselage.

Model Scale = .010.

DRAWING NUMBER VL70-000139

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length - IN	<u>346.0</u>	<u>3.460</u>
Max Width - IN	<u>108.0</u>	<u>1.080</u>
Max Depth - IN	<u>113.0</u>	<u>1.130</u>
Fineness Ratio	<u>          </u>	<u>          </u>
Area - FT <sup>2</sup>	<u>          </u>	<u>          </u>
Max Cross-Sectional	<u>          </u>	<u>          </u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE 3. - Continued..

MODEL COMPONENT: Attach Structure - AT<sub>2</sub>GENERAL DESCRIPTION: Attach Structure for Orbiter-Tank Configuration 3 perRockwell Lines VL70-000139B, VL78-000041

MODEL SCALE = 0.010

DRAWING NO. SS-A00060DIMENSIONS:FULL SCALEMODEL SCALEFORWARD ATTACH POINTS

## Orbiter to Tank

Number of Struts

Diameter in.

Location in.

 $X_o$  $X_T$ 

1

25.000

1

0.250

382.000

1077.000

3.820

10.770

## Orbiter to SRB

Number of Struts

Diameter in.

Location in.

 $X_o$  $X_s$ 

## Tank to SRB

Number of Struts

Diameter in.

Location in.

 $X_T$  $X_s$ AFT ATTACH POINTS

## Orbiter to Tank

Number of Struts /Side

Diameter in.

Location in.

 $X_o$  $X_T$ 

3

15.000

3

0.150

1292, 1308, 1308

1859, 2061, 2061

12.92, 13.08, 13.08

18.59, 20.61, 20.61

## Orbiter to SRB

Number of Struts

Diameter in.

Location in.

 $X_o$  $X_s$ 

## Tank to SRB

Number of Struts

Diameter in.

Location in.

 $X_T$  $X_s$

TABLE 3. - Continued.

MODEL COMPONENT: NOZZLES - N 8GENERAL DESCRIPTION: Basic OMS nozzle of Configuration 2A per Rockwell LinesVL70-008306 and VL70-000089"B". Intersection of nozzle exit plane andnozzle centerline at  $X_0 = 1570.75$ ,  $Y_0 = \pm 99.25$ ,  $Z_0 = 507.25$ MODEL SCALE = .010DRAWING NO. VL70-008306, VL70-000089"B", SS-A00092

<u>DIMENSIONS</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Mach No. _____		
Length ~ in.		
Gimbal Point to Exit Plane	_____	_____
Throat to Exit Plane	_____	_____
Diameter ~ in.		
Exit	<u>50.00</u>	<u>0.500</u>
Throat	<u>N/A</u>	<u>N/A</u>
Inlet	<u>28.00</u>	<u>0.280</u>
Area ~ ft <sup>2</sup> /Nozzle		
Exit	<u>13.635</u>	<u>0.00136</u>
Throat	_____	_____
Gimbal Point (station) ~ in.		
X	<u>1518.0</u>	<u>15.18</u>
Y	<u>±88.0</u>	<u>±0.88</u>
Z	<u>492.0</u>	<u>4.92</u>
Null Position ~ deg.		
Pitch	<u>15°49'</u>	<u>15°49'</u>
Yaw (Outb'd)	<u>±12°17'</u>	<u>±12°17'</u>

TABLE 3. - Continued.

MODEL COMPONENT: MPS NOZZLES - N 24GENERAL DESCRIPTION: Configuration 3A MPS NozzlesMODEL SCALE = .010DRAWING NO. VL70-000140A, VL70-005030A

<u>DIMENSIONS</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Mach No. _____		
Length~in.		
Gimbal Point to Exit Plane	_____	_____
Throat to Exit Plane	_____	_____
Diameter~in.		
Exit	<u>91.000</u>	<u>0.910</u>
Throat	_____	_____
Inlet	_____	_____
Area~ft <sup>2</sup> . /Nozzle		
Exit	<u>45.16585</u>	<u>0.00453</u>
Throat	_____	_____
Gimbal Point (station)~in.		
Upper Nozzle		
X	<u>14.45</u>	<u>14.45</u>
Y	<u>0</u>	<u>0</u>
Z	<u>4.43</u>	<u>4.43</u>
Lower Nozzles		
X	<u>14.68.16996</u>	<u>14.68170</u>
Y	<u>153.00000</u>	<u>0.53000</u>
Z	<u>342.63988</u>	<u>3.42640</u>
Null Position~deg.		
Upper Nozzle		
Pitch	<u>16°</u>	<u>16°</u>
Yaw	<u>0°</u>	<u>0°</u>
Lower Nozzles		
Pitch	<u>10°</u>	<u>10°</u>
Yaw (outb'd)	<u>3.5°</u>	<u>3.5°</u>

TABLE 3. - Continued.

MODEL COMPONENT: Solid Plume - PL 1

GENERAL DESCRIPTION: SSME simulated plumes from N24 nozzles to  
represent all 3 engines at M= 5.5 during exit trajectory

MODEL SCALE = .010

DRAWING NUMBER: \_\_\_\_\_

COORDINATES:

Ratio of local plume radius  
to nozzle exit plane internal  
radius

Ratio of local axial distance  
from nozzle exit plane to nozzle  
exit plane internal radius

1.053  
1.943  
2.772  
3.497  
4.450  
5.421  
5.905  
6.389  
7.321  
7.861  
8.136  
8.672  
8.937  
9.204  
9.464

0.057  
1.122  
2.250  
3.341  
4.912  
6.642  
7.566  
8.529  
10.496  
11.699  
12.330  
13.602  
14.367  
14.912  
15.569

DIMENSIONS:  
Nozzle Exit Radius, in.

FULL SCALE  
45.2

MODEL SCALE  
0.452

TABLE 3. - Continued.

MODEL COMPONENT: EXTERNAL TANK - T10

GENERAL DESCRIPTION: External Oxygen Hydrogen Tank, 3 Configuration,  
per Rockwell Lines VL78-000041 and VL72-000088

Model Scale = .010

DRAWING NUMBER

VL72-000088

VL78-000041

DIMENSION:

FULL SCALE

MODEL SCALE

Length - IN (Nose @  $X_T = 309$ )

1865

18.65

Max Width (Dia) - IN.

324

3.24

Max Depth

-

-

Fineness Ratio

5.75617

5.75617

Area -  $FT^2$

Max Cross-Sectional

572.555

0.05726

Planform

Wetted

Base

WP of Tank Centerline ( $X_T$ ) IN.

400.0

4.00



TABLE 3. - Continued.

MODEL COMPONENT: RUDDER - R5GENERAL DESCRIPTION: 2A, 3 and 3A Configuration per Rockwell LinesVL70-000095Model Scale = .010DRAWING NUMBER: VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>106.38</u>	<u>0.01064</u>
Span (equivalent) - IN.	<u>201.0</u>	<u>2.010</u>
Inb'd equivalent chord	<u>91.585</u>	<u>0.91585</u>
Outb'd equivalent chord	<u>50.833</u>	<u>0.50833</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83</u>	<u>34.83</u>
Tailing Edge	<u>26.25</u>	<u>26.25</u>
Hingeline	<u>34.83</u>	<u>34.83</u>
Area Moment (Normal to hinge line)- FT <sup>3</sup>	<u>526.13</u>	<u>0.00053</u>
Product of Area and Mean Chord		

TABLE 3. - Continued.

MODEL COMPONENT: VERTICAL - V7GENERAL DESCRIPTION: Centerline vertical tail, double-edge airfoil with rounded leading edge.NOTE: Same as V5, but with manipulator housing removed.Model Scale = .010

DRAWING NUMBER:

VL70-000139DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area (Theo) Ft <sup>2</sup>	<u>425.92</u>	<u>0.04259</u>
Planform		
Span (Theo) In	<u>315.72</u>	<u>3.157</u>
Aspect Ratio	<u>1.675</u>	<u>1.675</u>
Rate of Taper	<u>0.507</u>	<u>0.507</u>
Taper Ratio	<u>0.404</u>	<u>0.404</u>
Sweep Back Angles, degrees		
Leading Edge	<u>45.000</u>	<u>45.000</u>
Trailing Edge	<u>26.249</u>	<u>26.249</u>
0.25 Element Line	<u>41.130</u>	<u>41.130</u>
Chords:		
Root (Theo) WP	<u>268.50</u>	<u>2.6850</u>
Tip (Theo) WP	<u>108.47</u>	<u>1.0847</u>
MAC	<u>199.81</u>	<u>1.9981</u>
Fus. Sta. of .25 MAC	<u>1463.50</u>	<u>14.6350</u>
W. P. of .25 MAC	<u>635.522</u>	<u>6.3552</u>
B. L. of .25 MAC	<u>0.00</u>	<u>0.00</u>
Airfoil Section		
Leading Wedge Angle Deg	<u>10.000</u>	<u>10.000</u>
Trailing Wedge Angle Deg	<u>14.920</u>	<u>14.920</u>
Leading Edge Radius	<u>2.0</u>	<u>0.020</u>
Void Area - Ft <sup>2</sup>	<u>13.17</u>	<u>0.00132</u>
Blanketed Area	<u>0.00</u>	<u>0.00</u>

TABLE 3. - Concluded.

MODEL COMPONENT: WING-W 107GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VI70-000139BNOTE: Same as W103, except cuff, airfoil and incidence angle.

Model Scale = .010

TEST NO.DWG. NO. VI70-000139BDIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATAArea (Theo.)  $\text{Ft}^2$ 

Planform

2690.00

0.2690

Span (Theo) In.

936.68

9.3668

Aspect Ratio

2.265

2.265

Rate of Taper

1.177

1.177

Taper Ratio

0.200

0.200

Dihedral Angle, degrees (@ TE of Elevon)

3.500

3.500

Incidence Angle, degrees

0.500

0.500

Aerodynamic Twist, degrees

+3.000

+3.000

Sweep Back Angles, degrees

+3.000

+3.000

Leading Edge

45.000

45.000

Trailing Edge

-10.24

-10.24

0.25 Element Line

35.209

35.209

Chords:

Root (Theo) B.P.O.O.

689.24

6.8924

Tip, (Theo) B.P.

137.85

1.3785

MAC

474.81

4.7481

Fus. Sta. of .25 MAC

1136.89

11.3689

W.P. of .25 MAC

299.20

2.9920

B.L. of .25 MAC

182.13

1.8213

EXPOSED DATAArea (Theo)  $\text{Ft}^2$ 

1752.29

0.17523

Span, (Theo) In. BP108

720.68

7.2068

Aspect Ratio

2.058

2.058

Taper Ratio

0.2451

0.2451

Chords

Root BP108

562.40

5.6240

Tip 1.00  $\frac{b}{2}$ 

137.85

1.3785

MAC

393.03

3.9303

Fus. Sta. of .25 MAC

1185.31

11.8531

W.P. of .25 MAC

300.20

3.0020

B.L. of .25 MAC

251.76

2.5176

Airfoil Section (Rockwell Mod NASA)

XXXX-64

Root  $\frac{b}{2}$  =

0.10

0.10

Tip  $\frac{b}{2}$  =

0.12

0.12

Data for (1) of (2) Sides

Leading Edge Cuff  $\text{Ft}^2$ 

118.333

0.01183

Planform Area  $\text{Ft}^2$ 

500

5.000

Leading Edge Intersects Fus M. L. @ Sta

1083.4

10.834

Leading Edge Intersects Wing @ Sta

**Notes:**

1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrow
2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

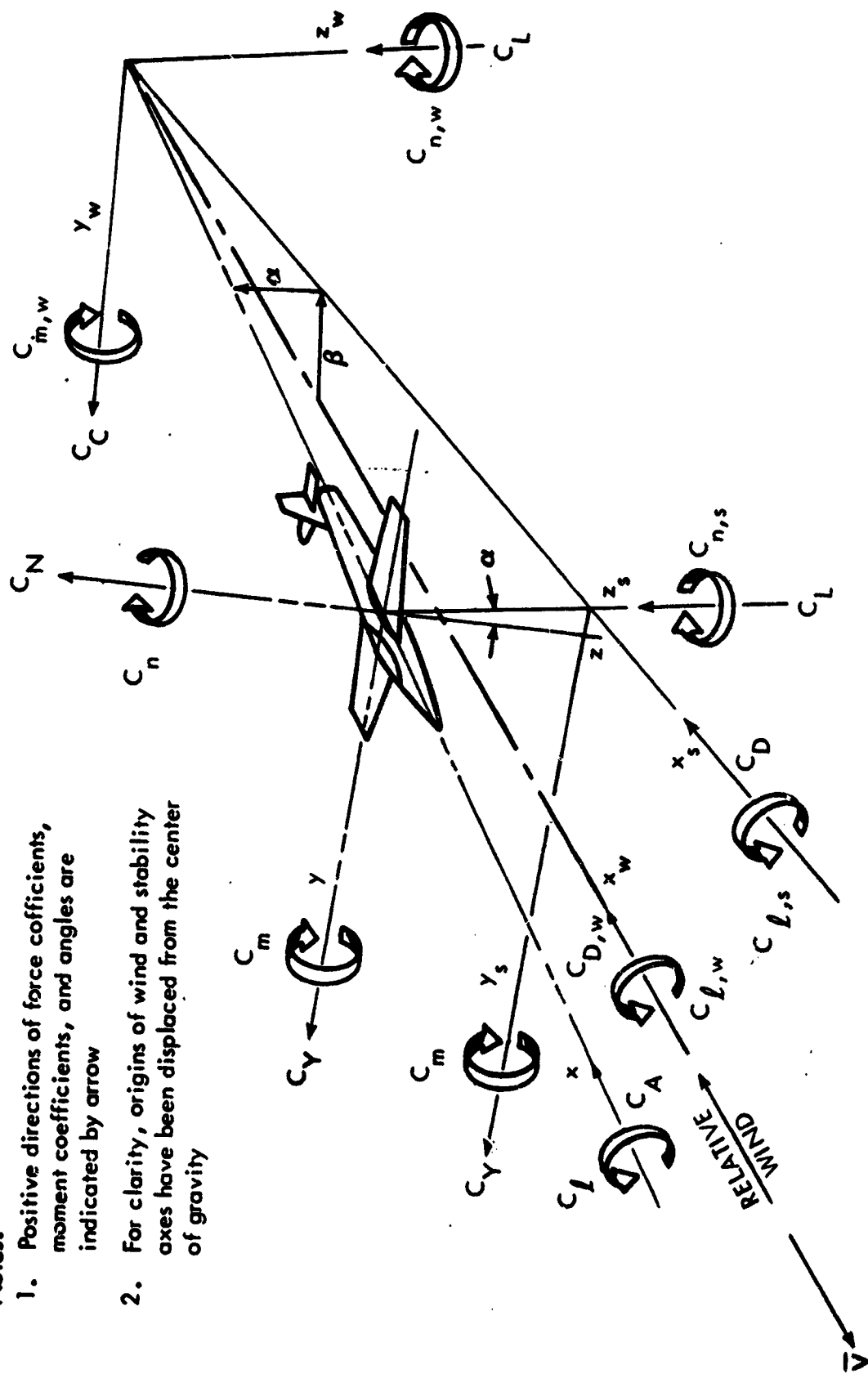


Figure 1. - Axis Systems.

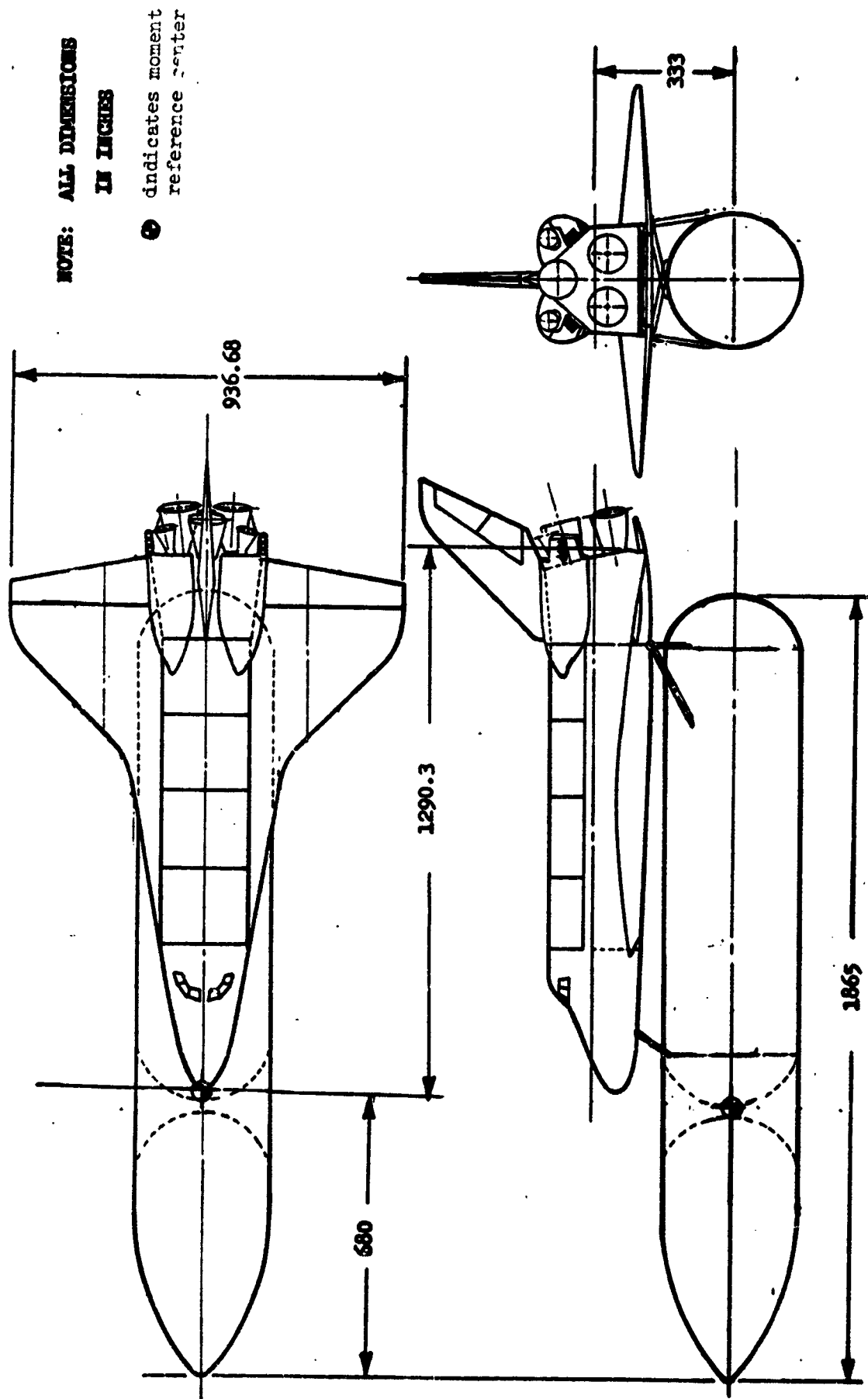
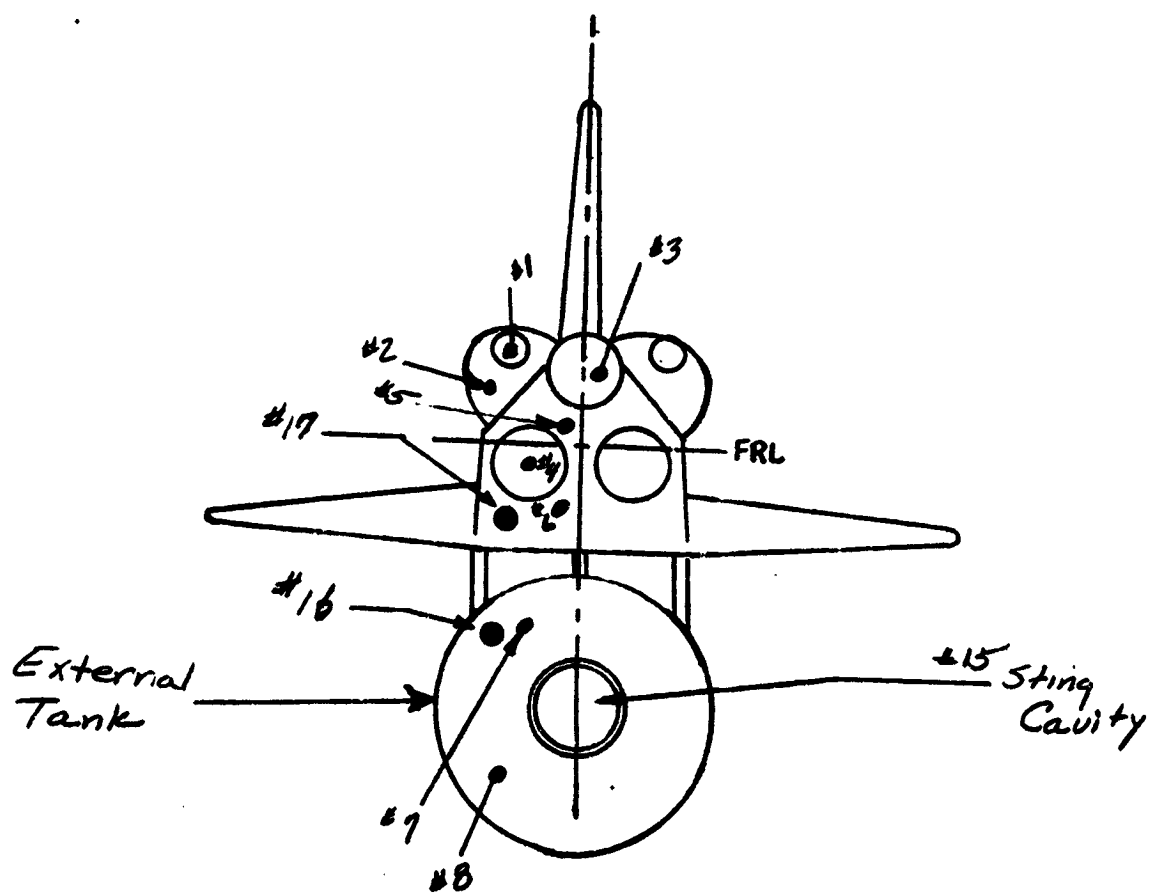


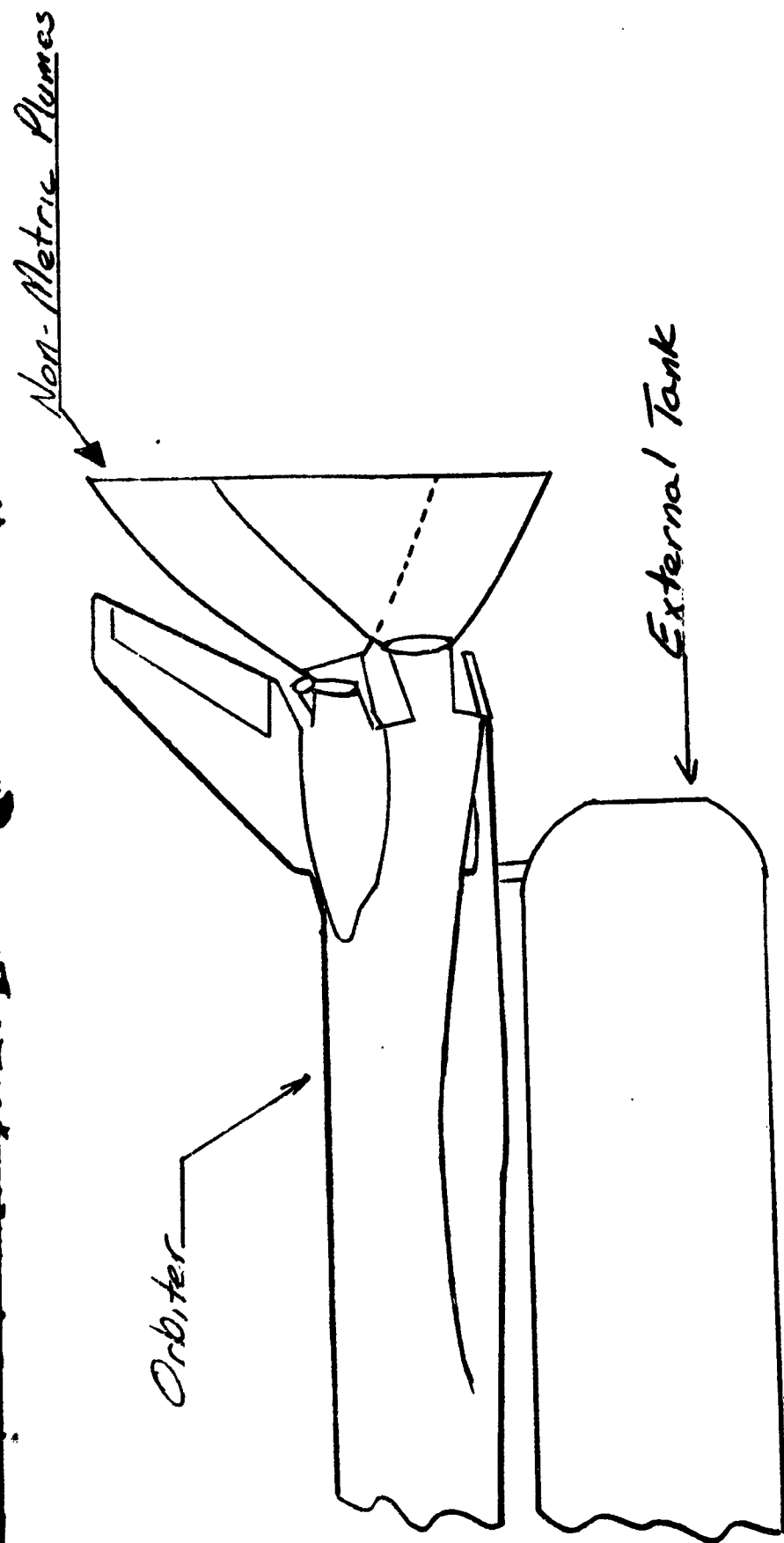
Figure 2. Model Sketches.

a. Orbiter Tank General Arrangement, O<sub>9</sub>T<sub>10</sub>



b. Base Pressure Orifice Locations

Figure 2. - Continued.



C. Plume Simulators

Figure 2. - Concluded.



Figure 3. - Model installation photograph.



DATA FIGURES

DATA SET SYMBOL: (587010)  
(587011)

CONFIGURATION DESCRIPTION: AYES 3.5-169 1A10 09 T10 AT2 PLUME ON  
AYES 3.5-169 1A10 09 T10 AT2 PLUME OFF

BETA: .000  
AILRON: .000  
ELEVON: .000  
RUDDER: .000

REFERENCE INFORMATION:  
SREF: 2690.0000 SQ.FT.  
LREF: 1290.0000 IN.  
BREF: 936.6800 IN.  
XPRP: 1076.4800 IN.  
YPRP: 400.0000 IN.  
ZPRP: 400.0000 IN.  
SCALE: .0100

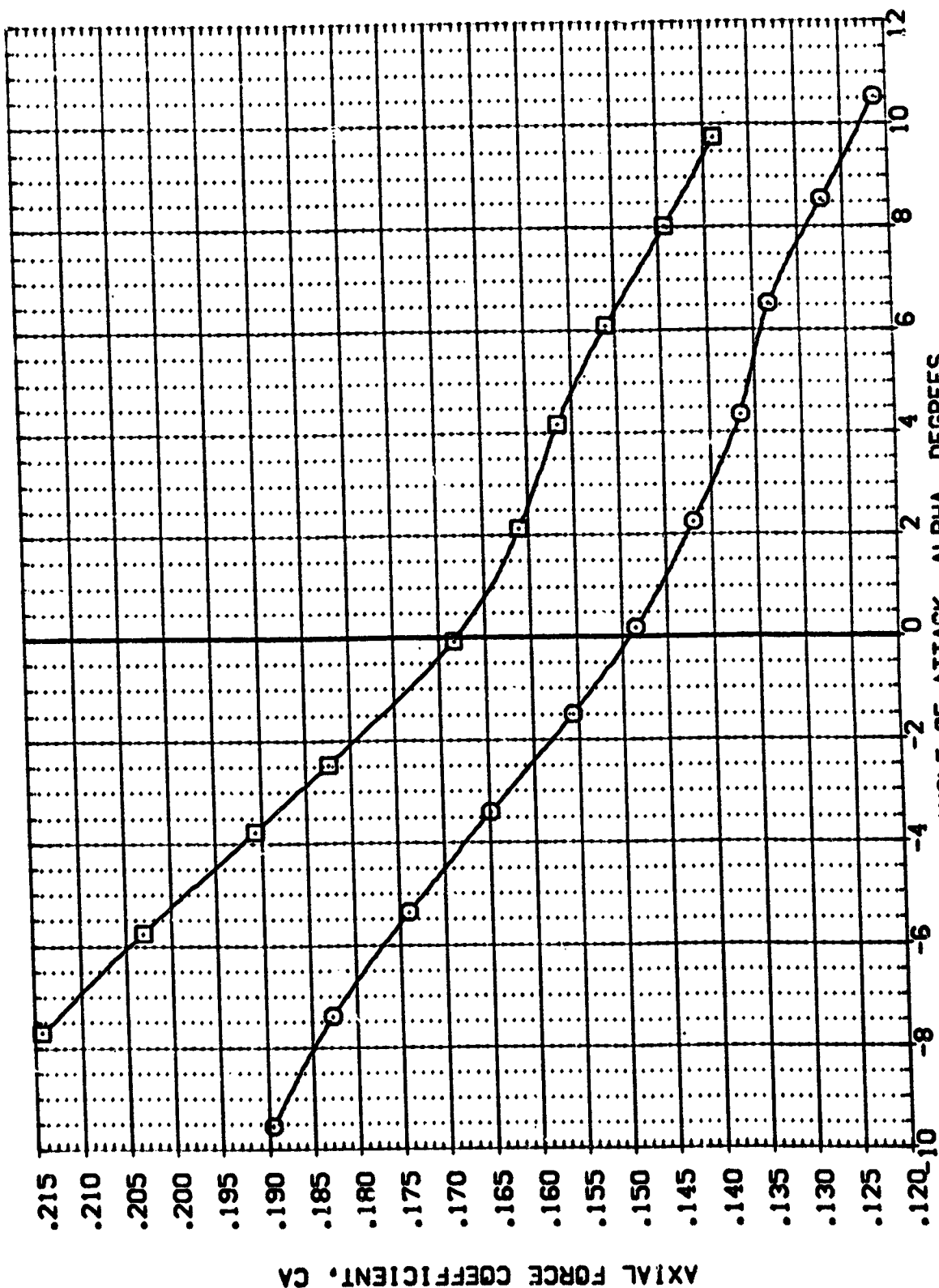


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 5.30

**NORMAL FORCE COEFFICIENT, CN**



**CAJMACH = 5.30**

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION
(587010)	AVES 3.5-169 1A10 09 T10 AT2 PLANE ON	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(087011)	AVES 3.5-169 1A10 09 T10 AT2 PLANE OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
						SREF 936.6800 IN.
						XMRP 1076.4800 IN.
						YMRP .0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

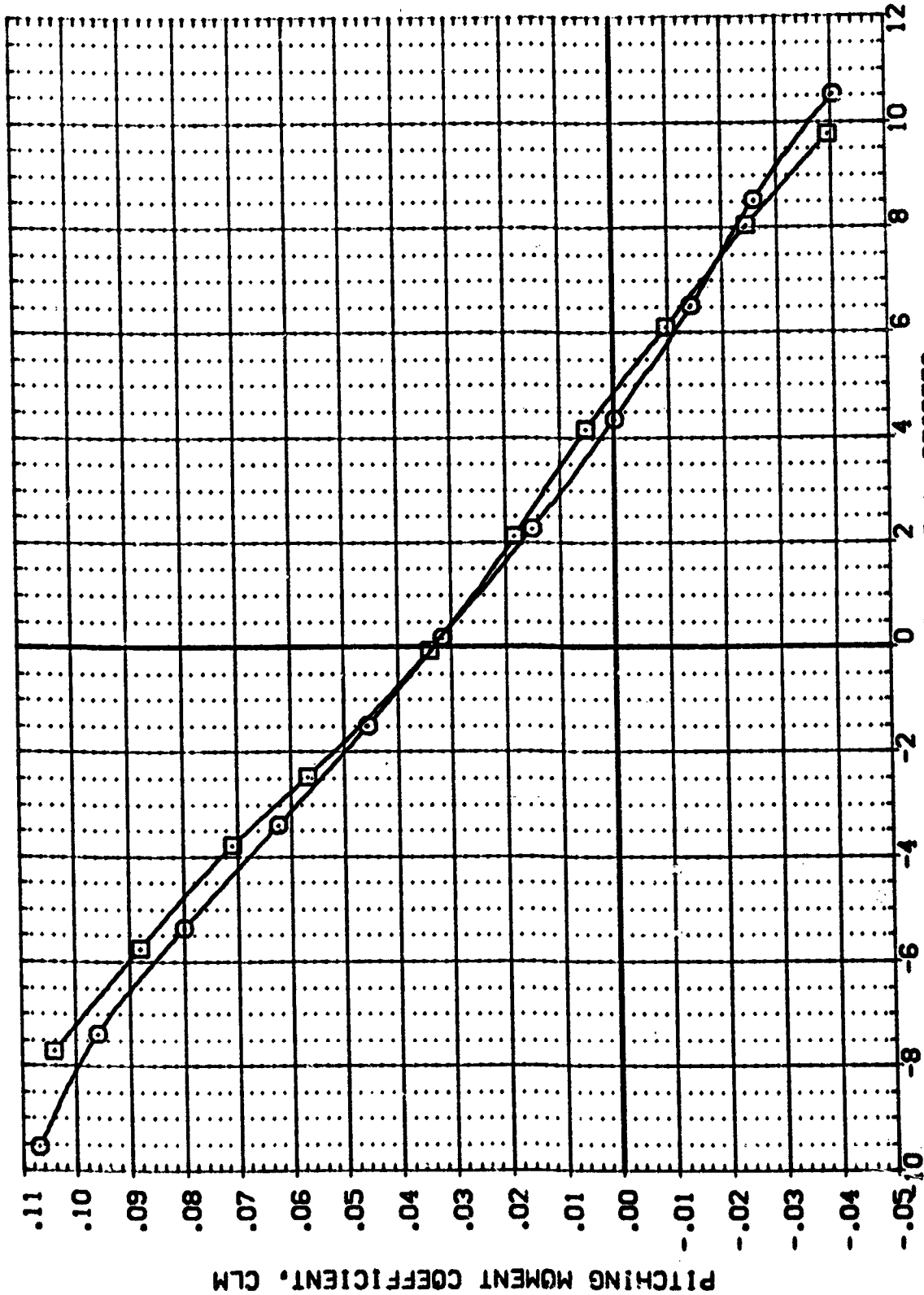


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 5.30

DATA SET SYMBOL: (SB7010) (DB7011)  
 CONFIGURATION DESCRIPTION: AVES 3.5-169 1A1/2 08 T10 AT12 PLUVE ON  
 AVES 3.5-169 1A1/2 08 T10 AT12 PLUVE OFF

BETA .000 .000  
 AILRON .000 .000  
 ELEVEN .000 .000  
 RUDDER .000 .000  
 REFERENCE INFORMATION  
 SREF 2190.0000 50.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

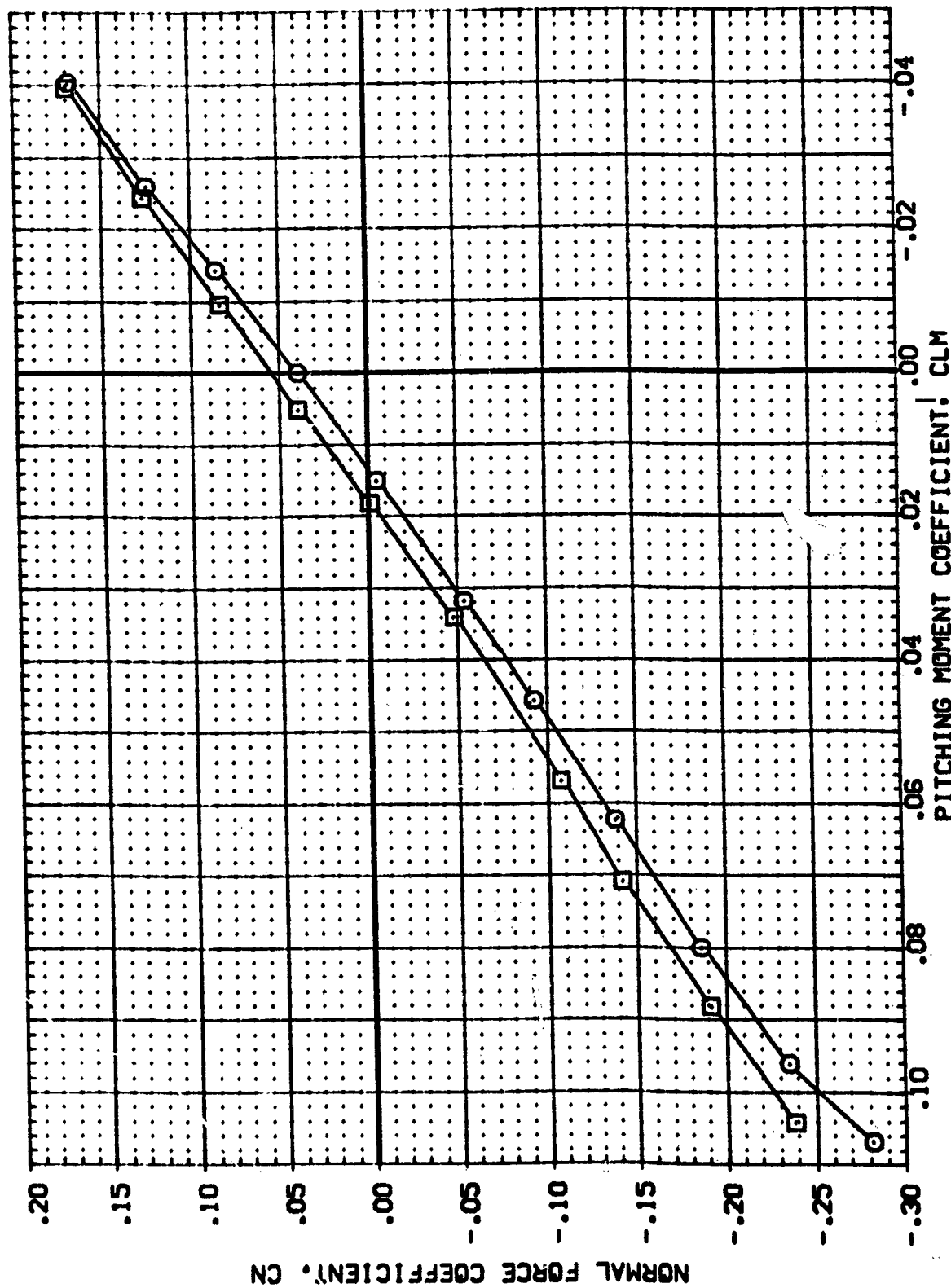


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(M)MACH = 5.30

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		BETA		AILLON		ELEVON		RUDDER		REFERENCE INFORMATION	
(197010)	□	AVES 3.5-189	IA10	09	T10	AT2	PLUVE	ON	.000	.000	.000	SREF	2690.0000
(197001)	○	AVES 3.5-189	IA10	09	T10	AT2	PLUVE	OFF	.000	.000	.000	LREF	1290.0000
												BREF	936.6800
												XMRP	1076.4800
												YMRP	.0000
												ZMRP	400.0000
												SCALE	.0100

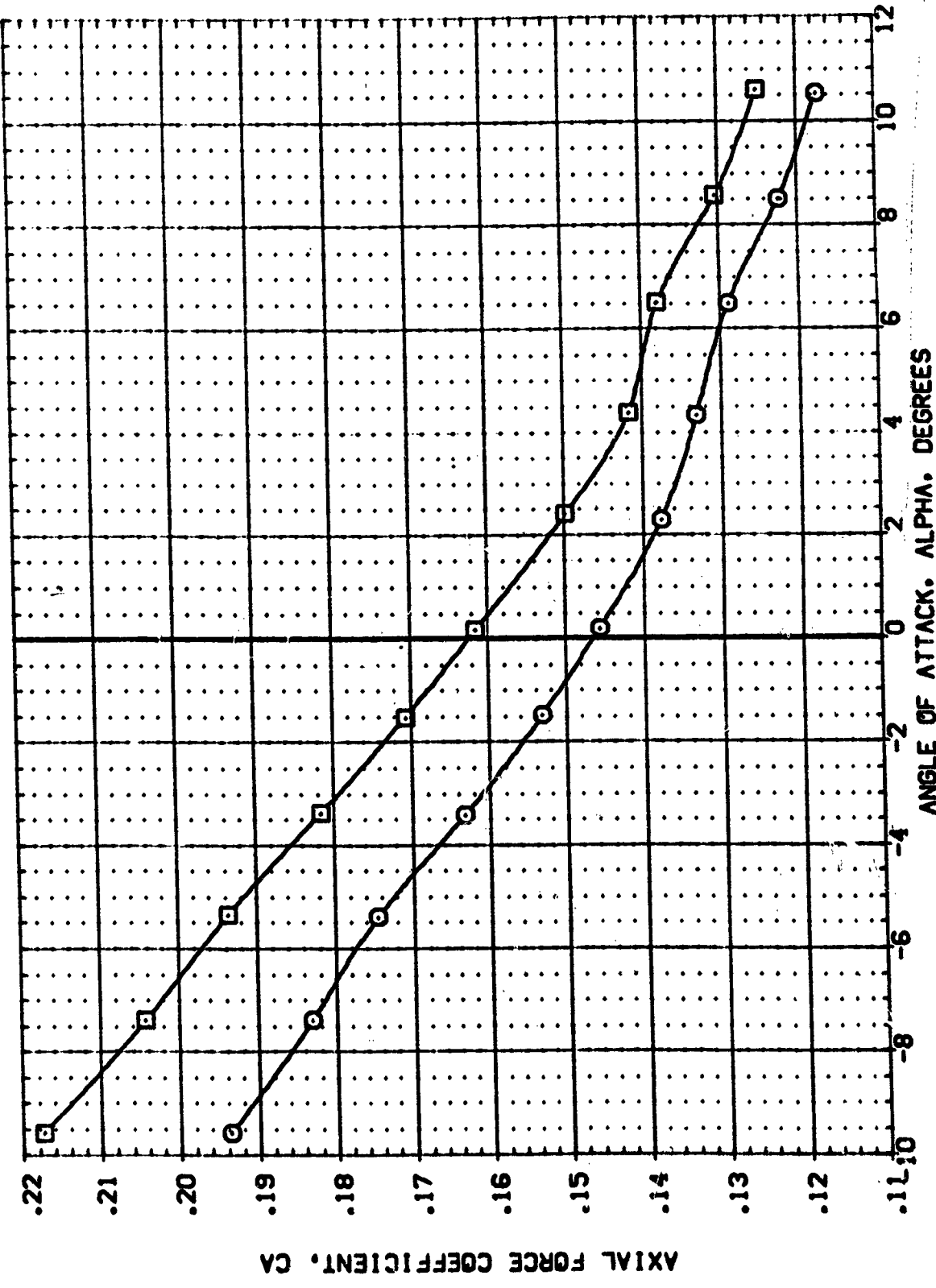


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(M)MACH = 7.32

DATA SET SYMBOL: (TB7010) (RB7001)

CONFIGURATION DESCRIPTION: AYES 3.5-169 1A10 09 T10 AT2 PLANE ON AYES 3.5-169 1A10 09 T10 AT2 PLANE OFF

BETA: .000 .000 .000 .000

AILRON: .000 .000 .000 .000

ELEVON: .000 .000 .000 .000

RUDDER: .000 .000 .000 .000

REFERENCE INFORMATION:

SREF	2690.0000	IN.
LREF	1290.0000	IN.
BREF	936.6900	IN.
XMRP	1076.4800	IN.
YMRP	400.0000	IN.
ZMRP	400.0000	IN.
SCALE	.0100	

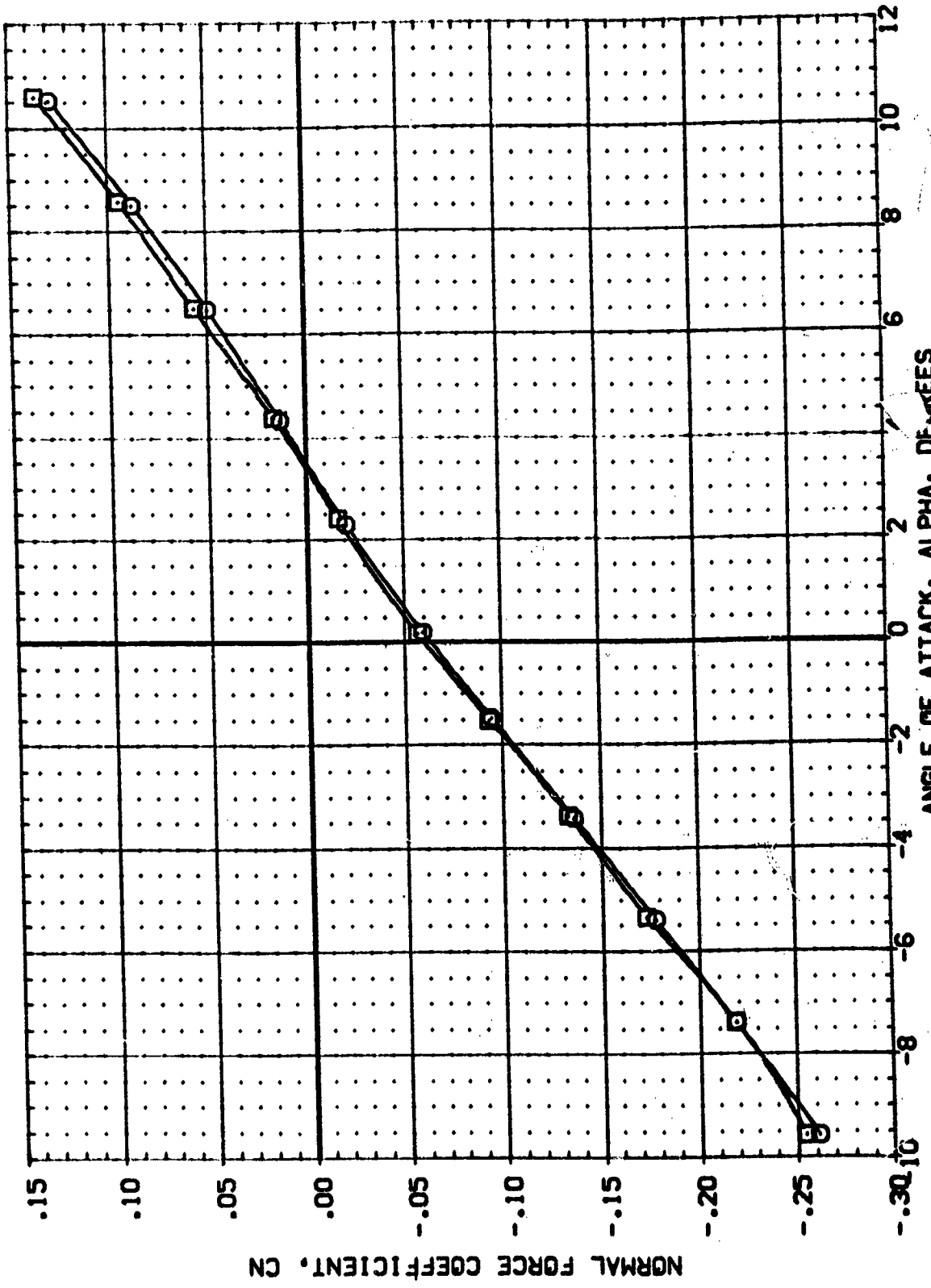


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(187010)	AVES 3-5-169 1A10 09 T10 AT2 PLUPE ON	.000	.000	.000	.000	SREF 2690.0000 SO.FT.
(187001)	AVES 3-5-169 1A10 09 T10 AT2 PLUPE OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						XMRP 1076.4800 IN.
						YMRP 400.0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

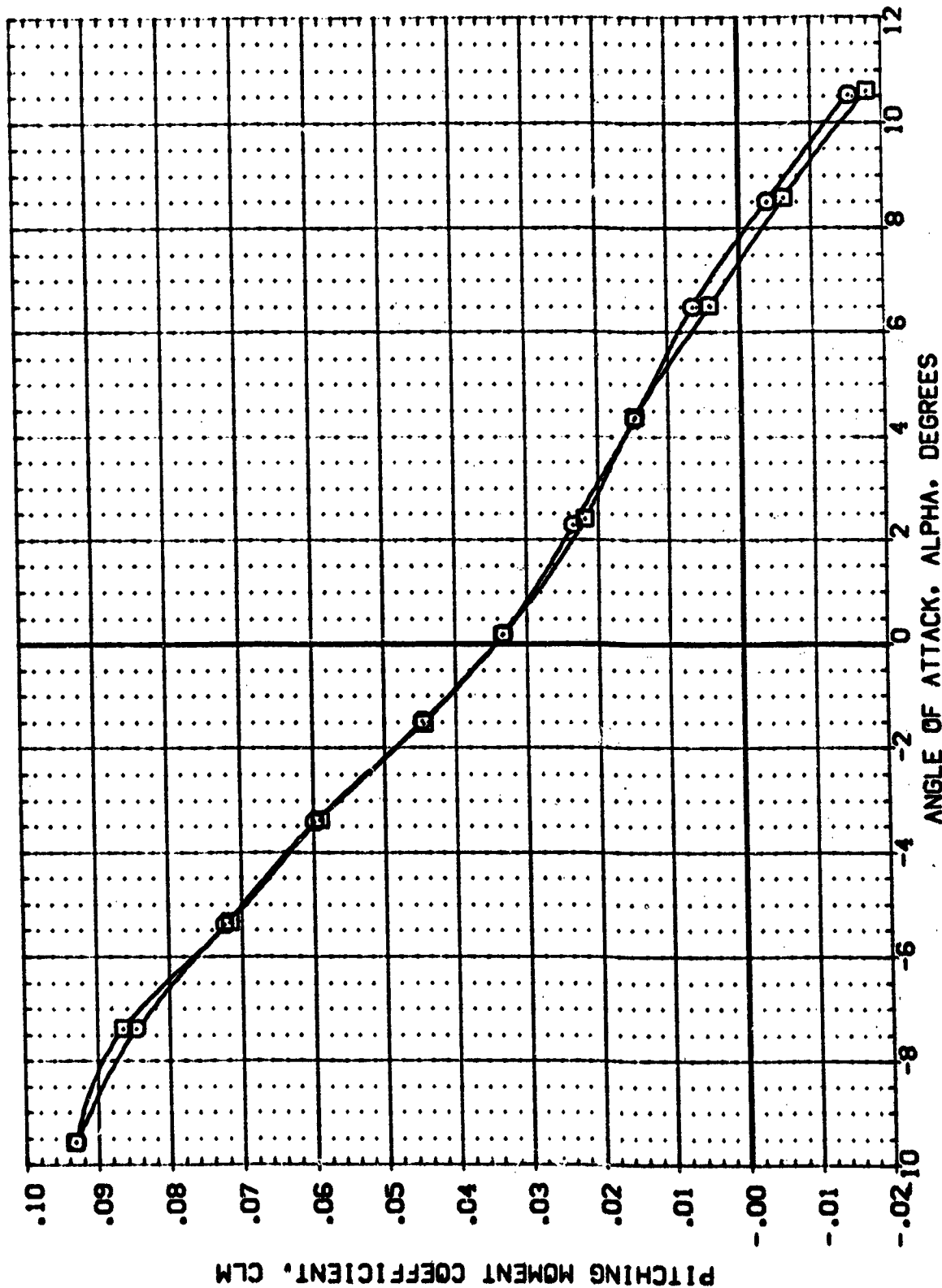


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 7.32



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(T87010)	AVES 3.5-169 1A10 09 T10 AT2 PLUVE ON	.000	.000	.000	.000	SREF 2690.0000 50.FT.
(R87001)	AVES 3.5-169 1A10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						XPMP 1076.4800 IN.
						YMMP .0000 IN.
						ZMMP 400.0000 IN.
						SCALE .0100

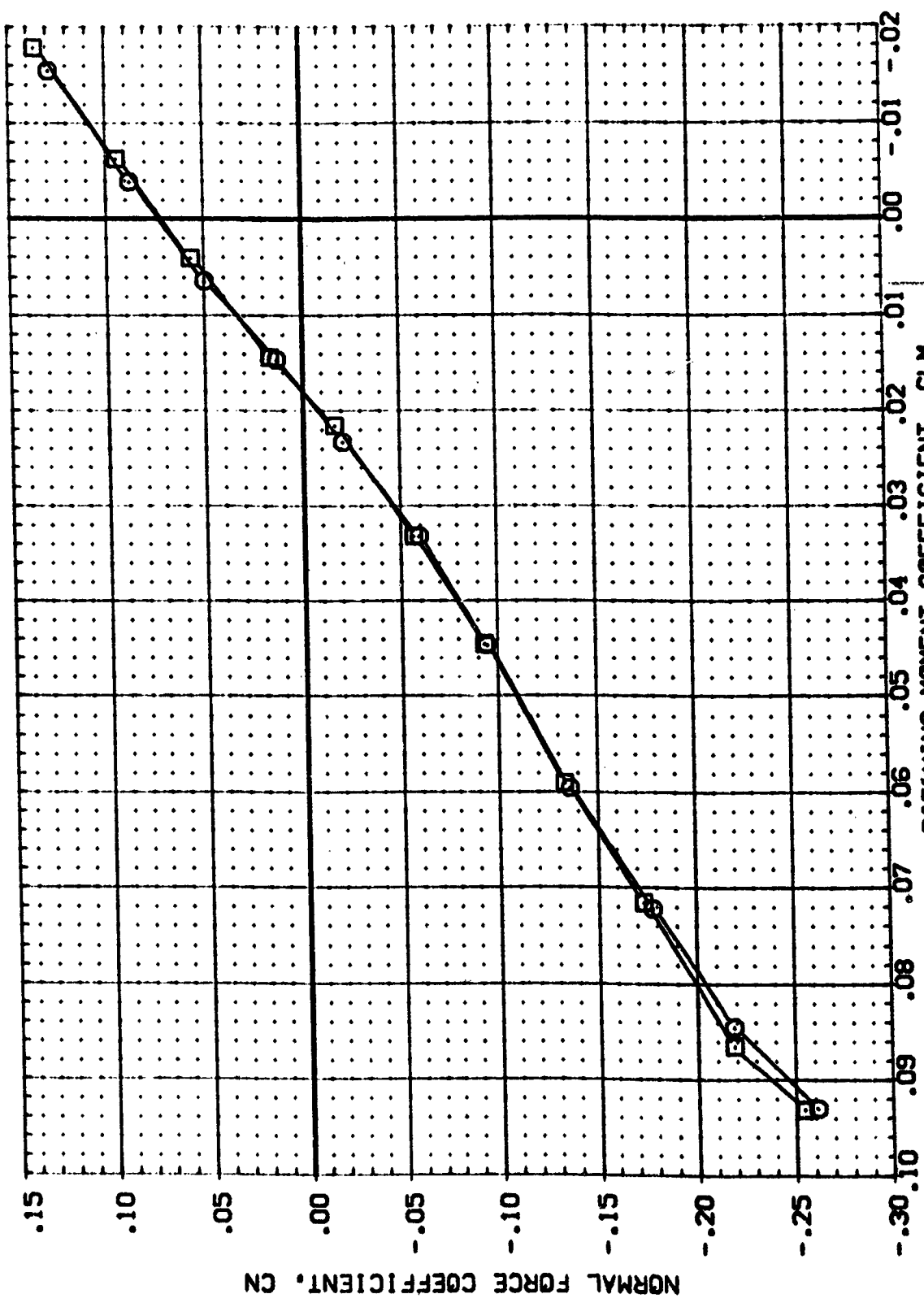


FIG. 4 EFFECTS OF SOLID PLUMES ON LONGITUDINAL CHARACTERISTICS.

(A)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION	
(R87009)	AVES 3.5-169 1A10 09 T10 AT2 PLUVE ON	.000	.000	.000	.000	SREF	2690.0000 SQ.FT.
(R87006)	AVES 3.5-169 1A10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	LREF	1290.0000 IN.
						SREF	936.6800 IN.
						XMRP	1076.4800 IN.
						YMRP	.0000 IN.
						ZMRP	400.0000 IN.
						SCALE	.0100

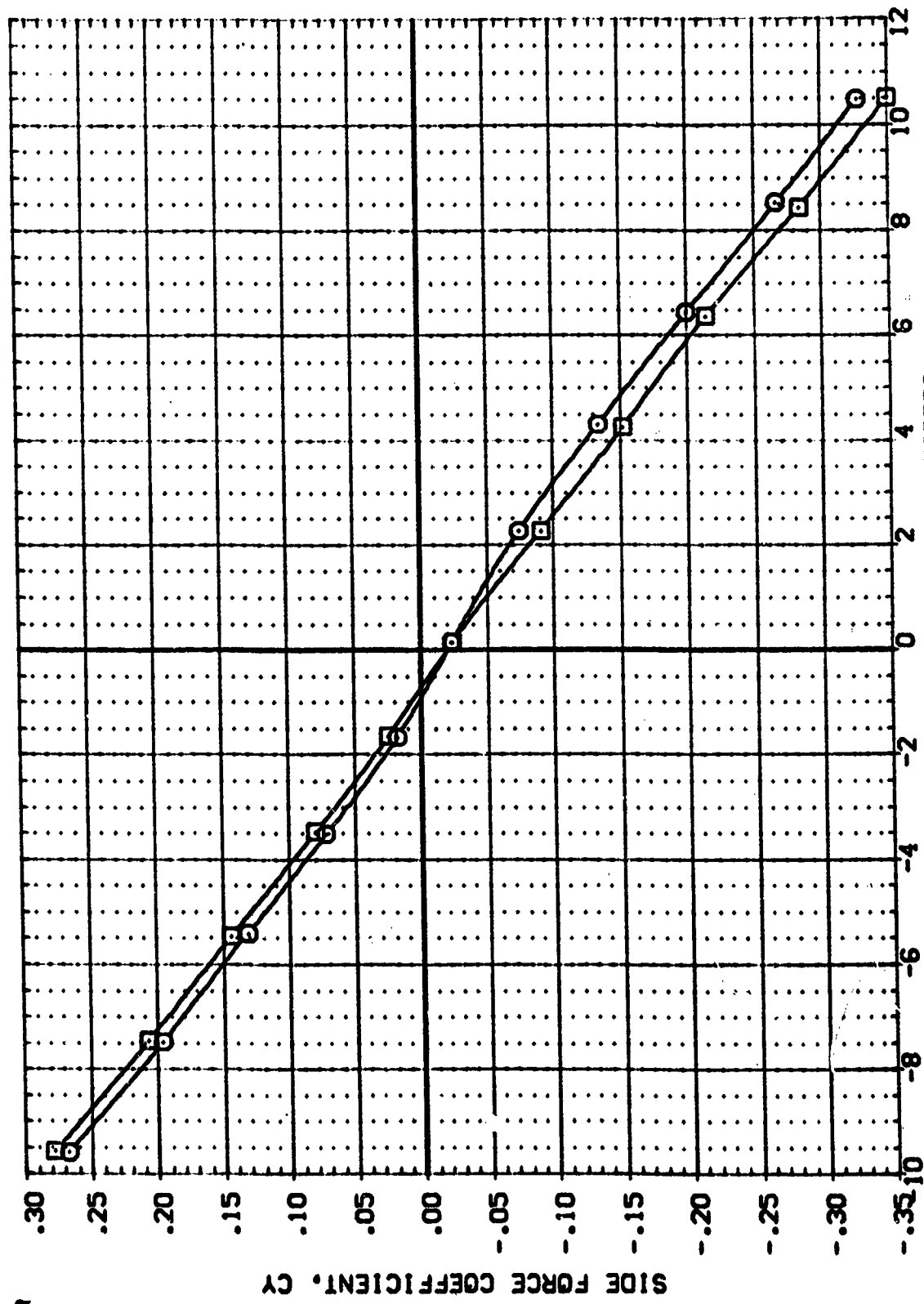


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87009)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE ON	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(R87006)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	LREF 1790.0000 IN.
						BREF 936.6800 IN.
						XMRP 1076.4800 IN.
						YMRP .0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

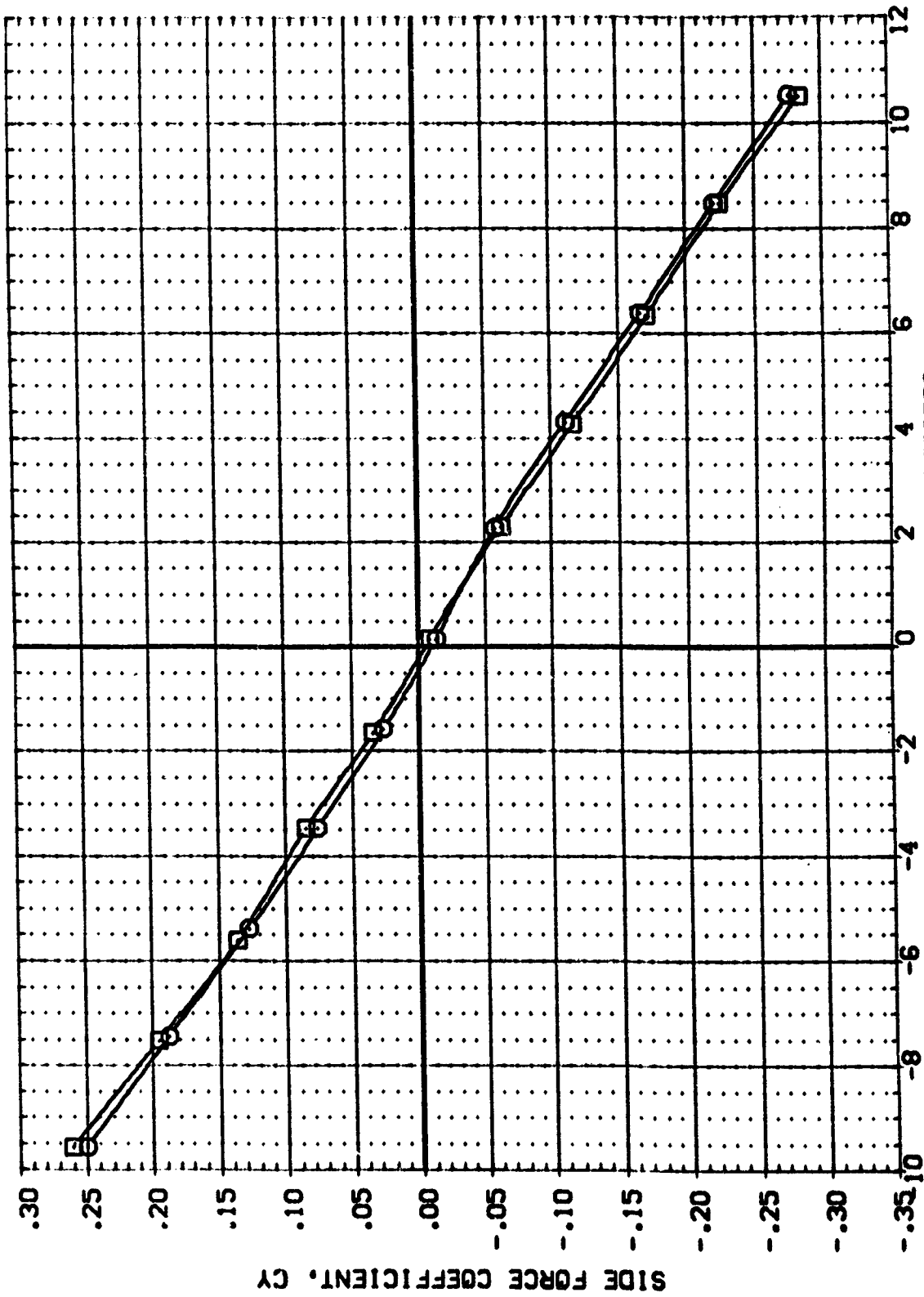


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (R87029) ☐ AVES 3.5-169 1A10 09 T10 AT2 PLUME ON  
 (R87036) ☐ AVES 3.5-169 1A10 09 T10 AT2 PLUME OFF

ALPHA AILRON ELEVON RUDDER REFERENCE INFORMATION  
 .000 .000 .000 SREF 2690.0000 50.FT.  
 .000 .000 .000 LREF 1290.0000 IN.  
 YPRP 936.6800 IN.  
 ZPRP 1076.4800 IN.  
 400.0000 IN.  
 SCALE .0100

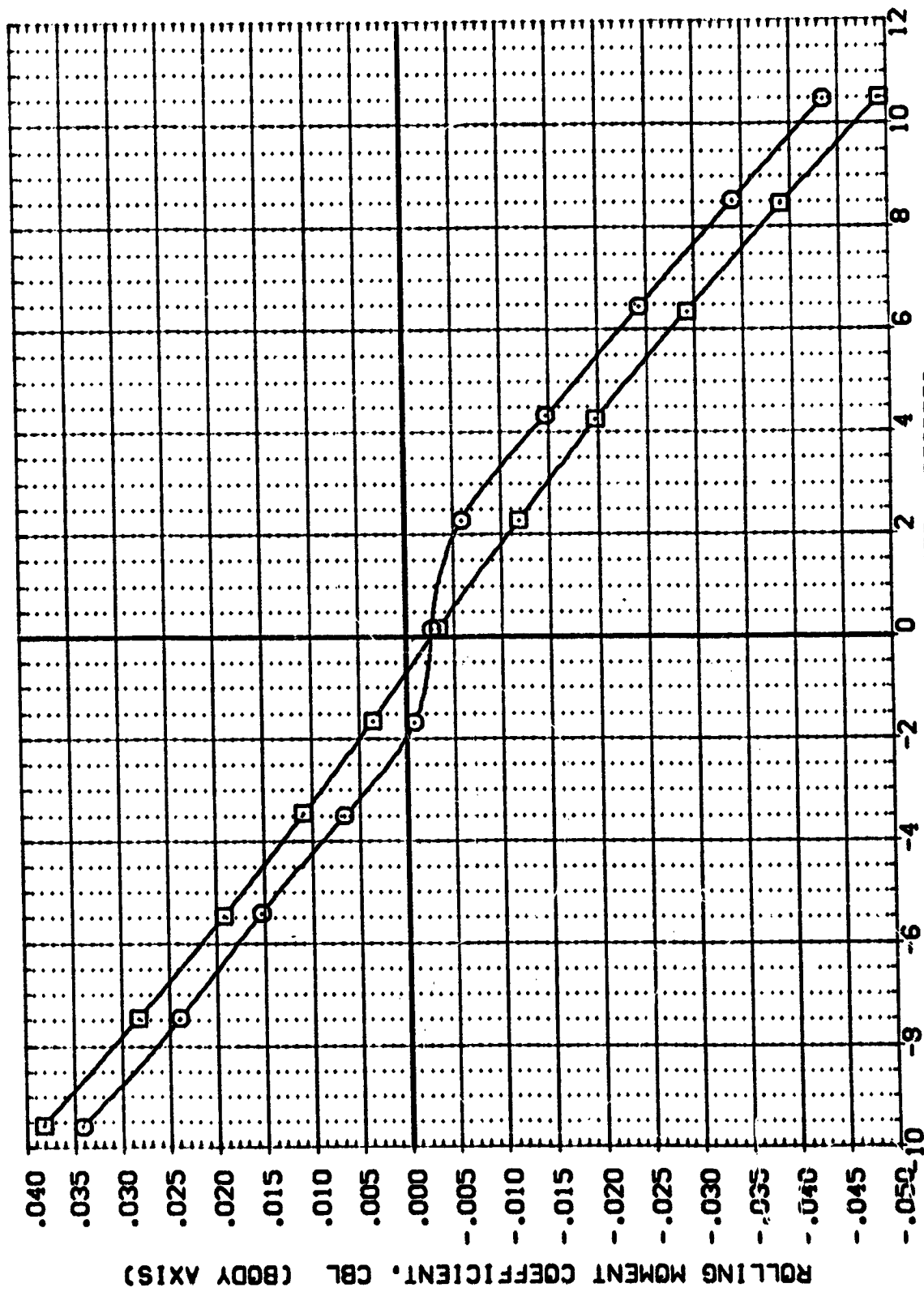


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		ALPHA		AILRON		ELEVON		RUDDER		REFERENCE INFORMATION	
(R87009)	AVES 3.5-169	IA10	09 T10 AT2 PLUME ON	.000	.000	.000	.000	.000	.000	.000	.000	SREF	2690.0000
(R87009)	AVES 3.5-169	IA10	09 T10 AT2 PLUME OFF	.000	.000	.000	.000	.000	.000	.000	.000	LREF	1290.0000
												BREF	936.6800
												YMRP	1076.4800
												ZMRP	400.0000
												SCALE	.0100

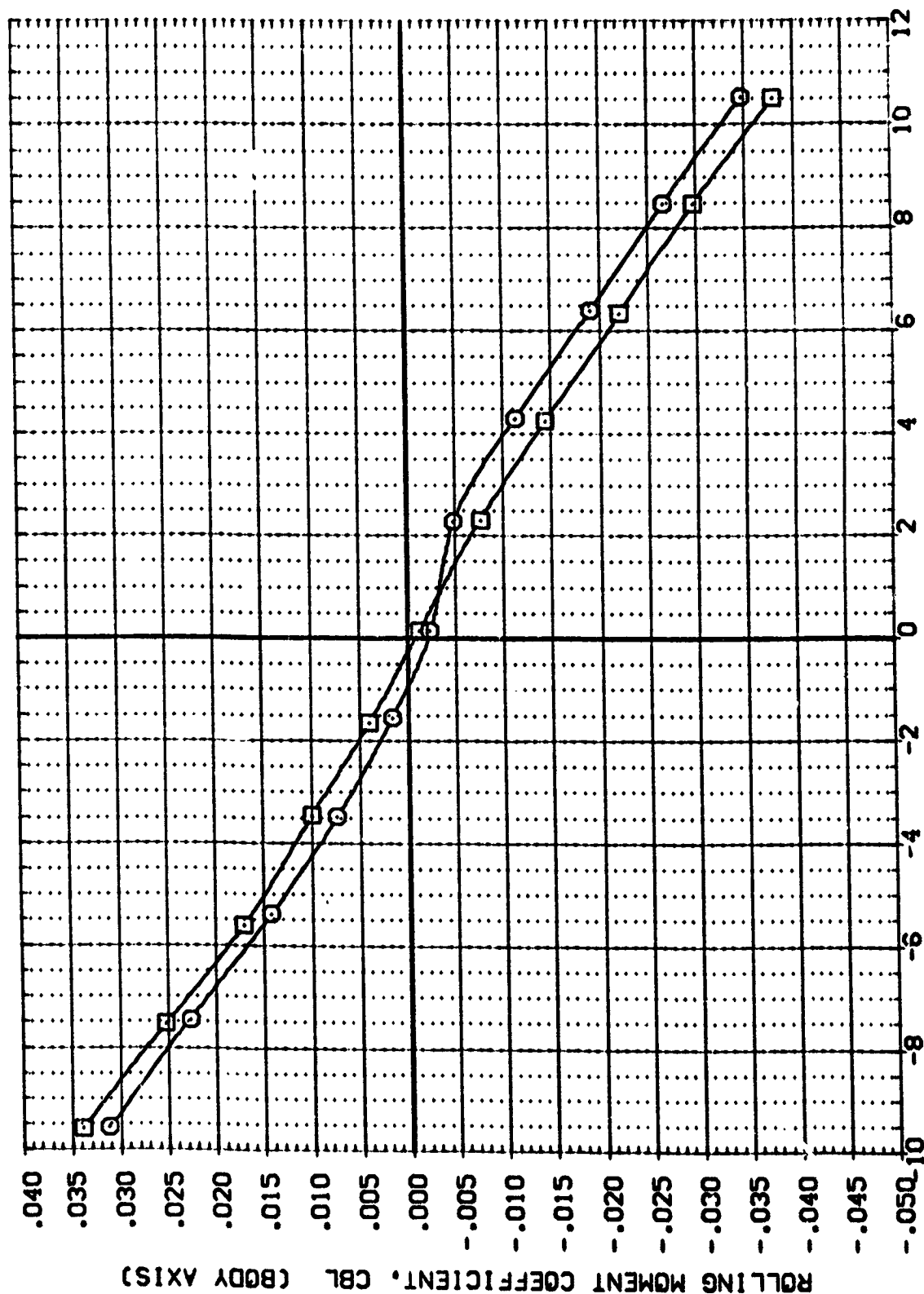


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
(RB7009)	□	AMES 3.5-169	1A10 09 T10 AT2 PLUME ON	SREF	2690.0000 SO.FT.
(RB7006)	○	AMES 3.5-169	1A10 09 T10 AT2 PLUME OFF	LREF	1280.0000 IN.
				BREF	936.6800 IN.
				XMRP	1076.4800 IN.
				YMRP	.0000 IN.
				ZMRP	400.0000 IN.
				SCALE	.0100

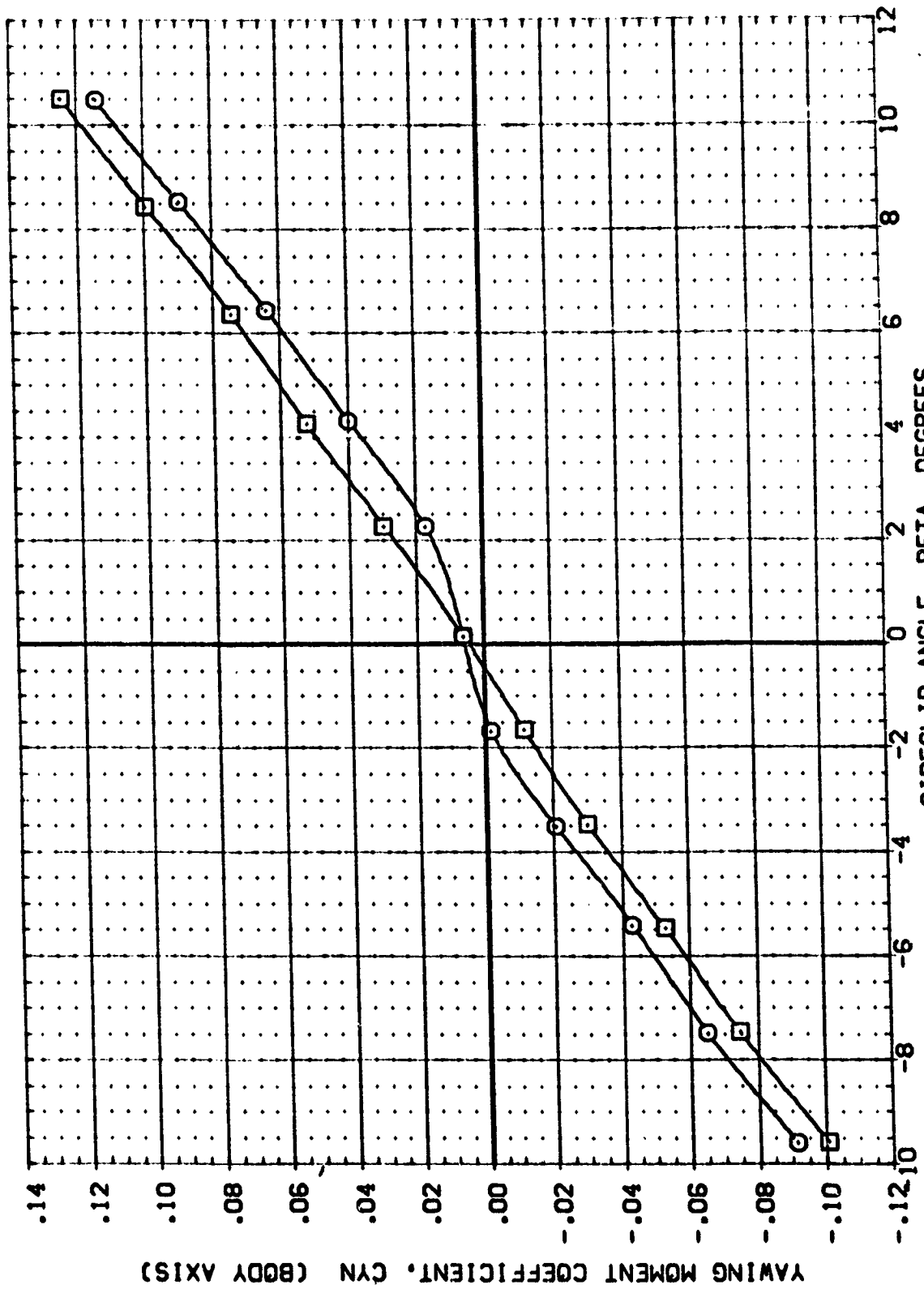


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
(RB7009)	□	AMES 3.5-169 1A10	09 T10 AT2 PLUME ON	SREF	2690.0000
(RB7006)	○	AMES 3.5-169 1A10	09 T10 AT2 PLUME OFF	LREF	1290.0000
				BREF	936.6800
				XMRP	1076.4800
				YMRP	.0000
				ZMRP	400.0000
				SCALE	.0100

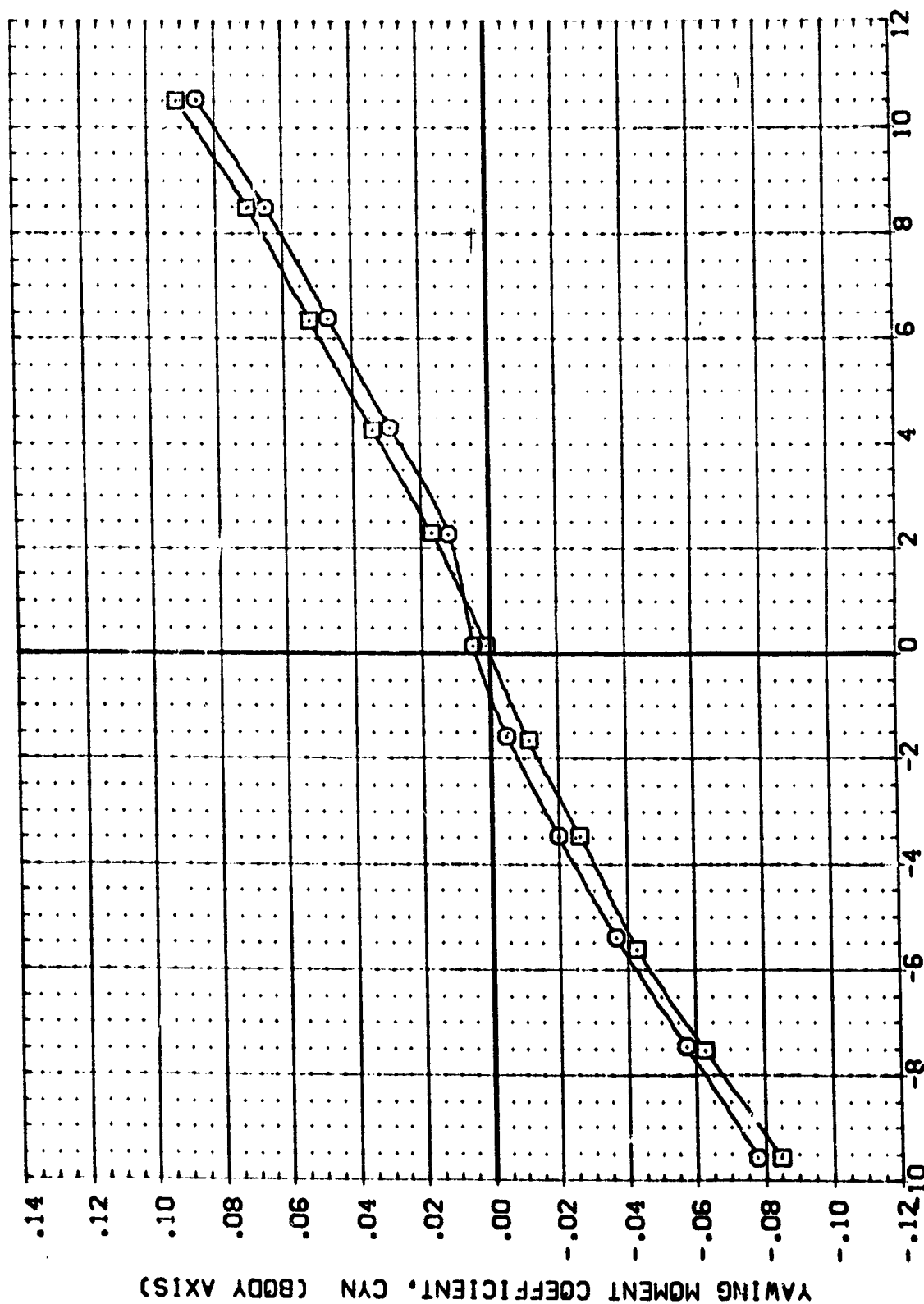


FIG. 5 EFFECTS OF SOLID PLUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		ALPHA		AILRON		ELEVON		RUDDER		REFERENCE INFORMATION	
(R87009)	□	AVES 3.5-169	IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	.000	.000	.000	.000	SREF	2690.0000
(R87006)	□	AVES 3.5-169	IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	.000	.000	.000	.000	.000	LREF	1290.0000
												BREF	936.6800
												XMRP	1076.4800
												YMRP	.0000
												ZMRP	400.0000
												SCALE	.0100
													SO.FT.
													IN.
													IN.
													IN.
													IN.

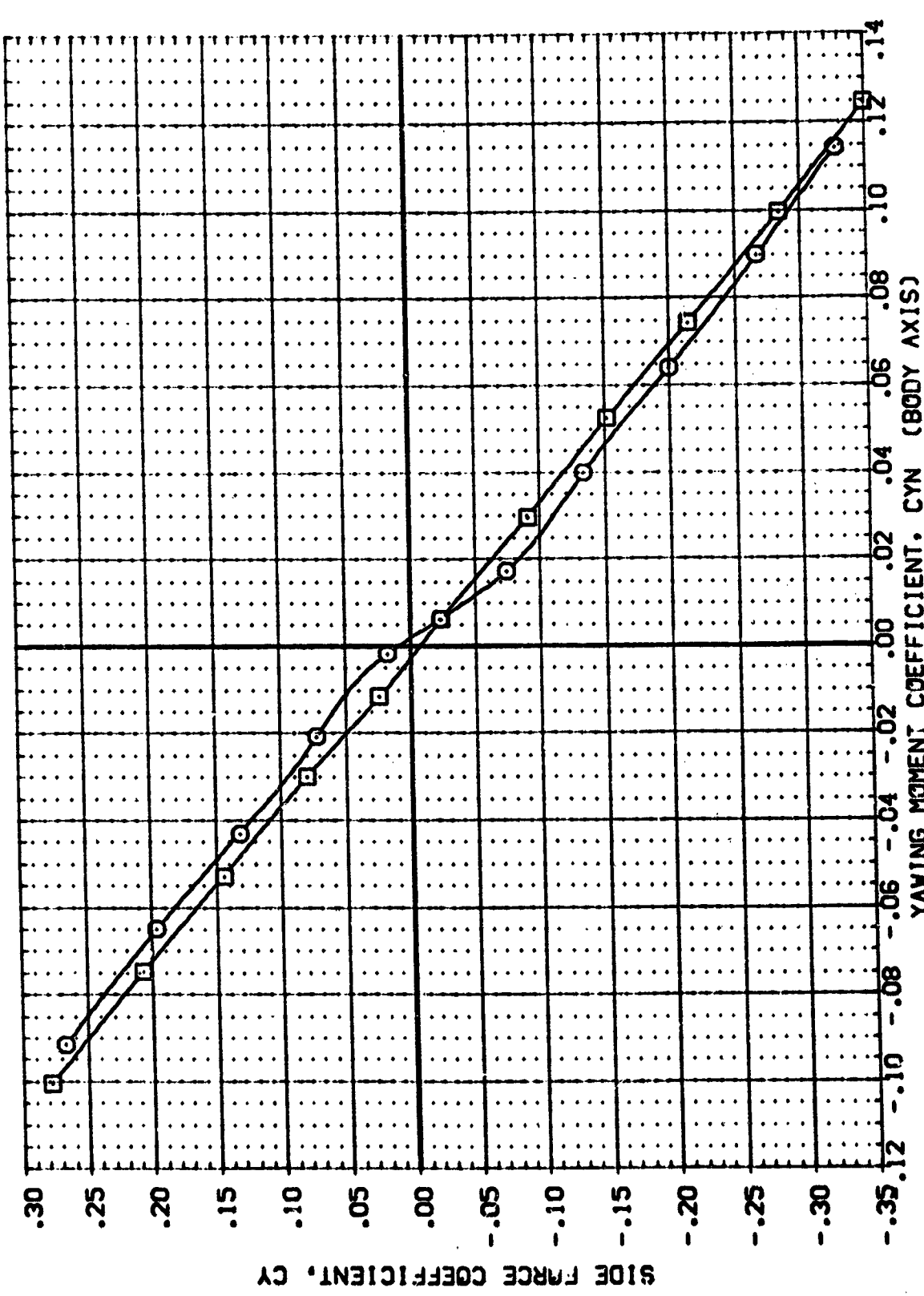


FIG. 5 EFFECTS OF SOLID PUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA-SET SYMBOL: (R87009) (R87006) CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 09 T10 AT2 PLUVE ON AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

REFERENCE INFORMATION: SREF 2690.0000 SD.FT. LREF 1290.0000 IN. BREF 936.6800 IN. XTRP 1076.4800 IN. YTRP 400.0000 IN. ZTRP 400.0000 IN. SCALE .0100

ALPHA .000 ELEVON .000 RUDDER .000

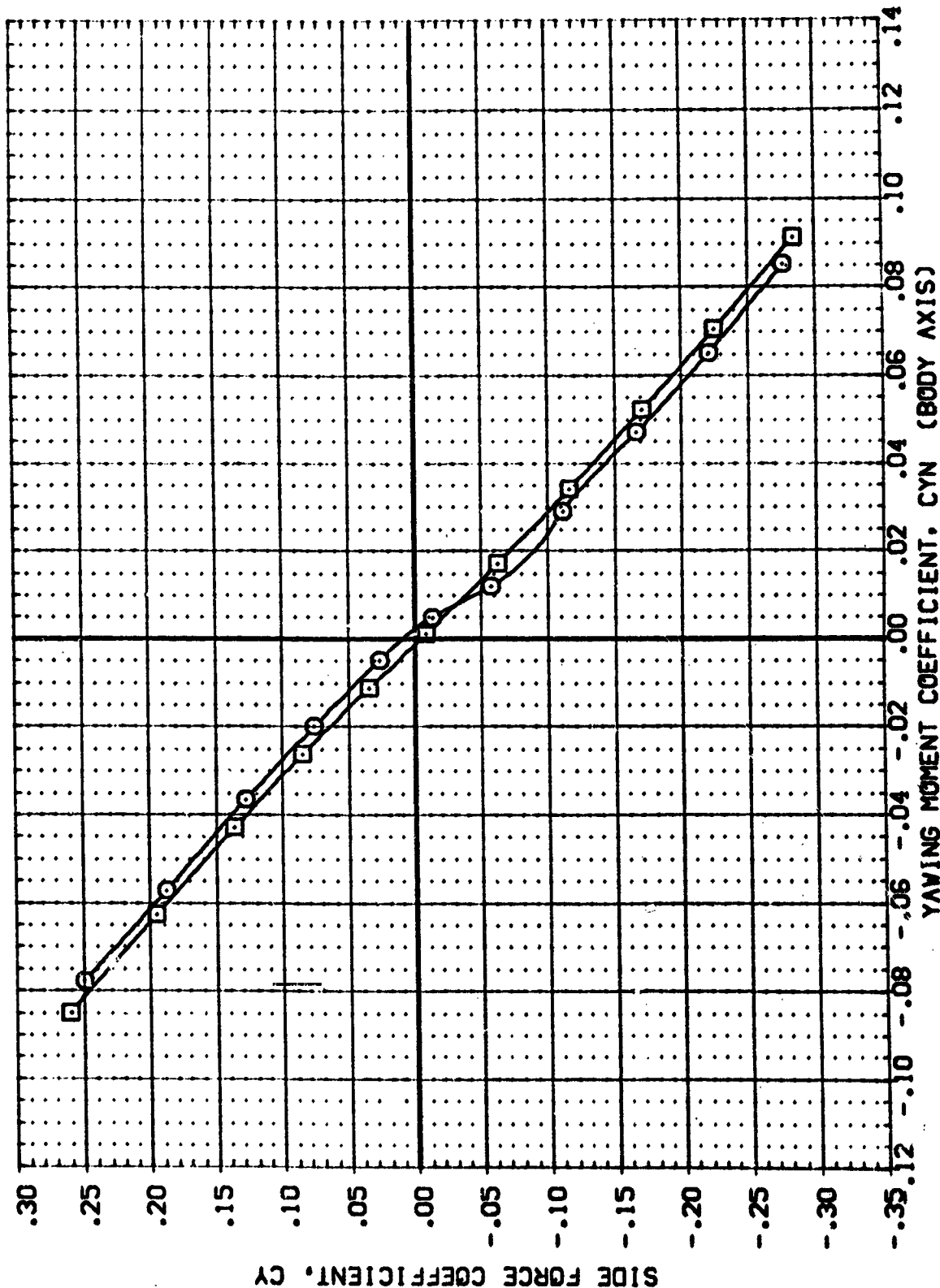


FIG. 5 EFFECTS OF SOLID PLOUMES ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	ANES 3.5-169 IA10 09 T10 AT2 PLUVE ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	ANES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	ANES 3.5-169 IA10 09 T10 AT2 PLUVE ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)	ANES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
						YMRP .0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

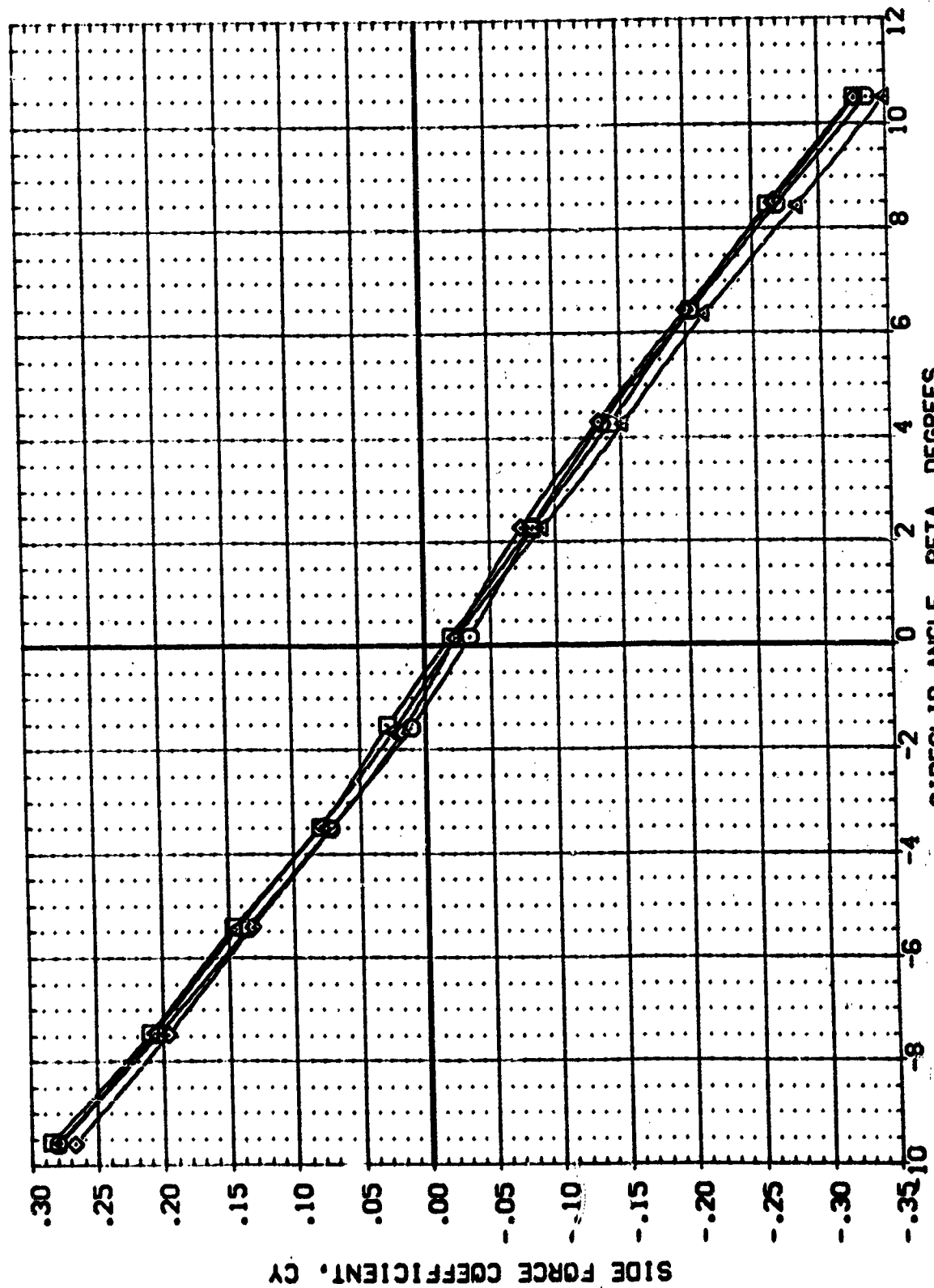


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(A)MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	.000	VMRP 1076.4800 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

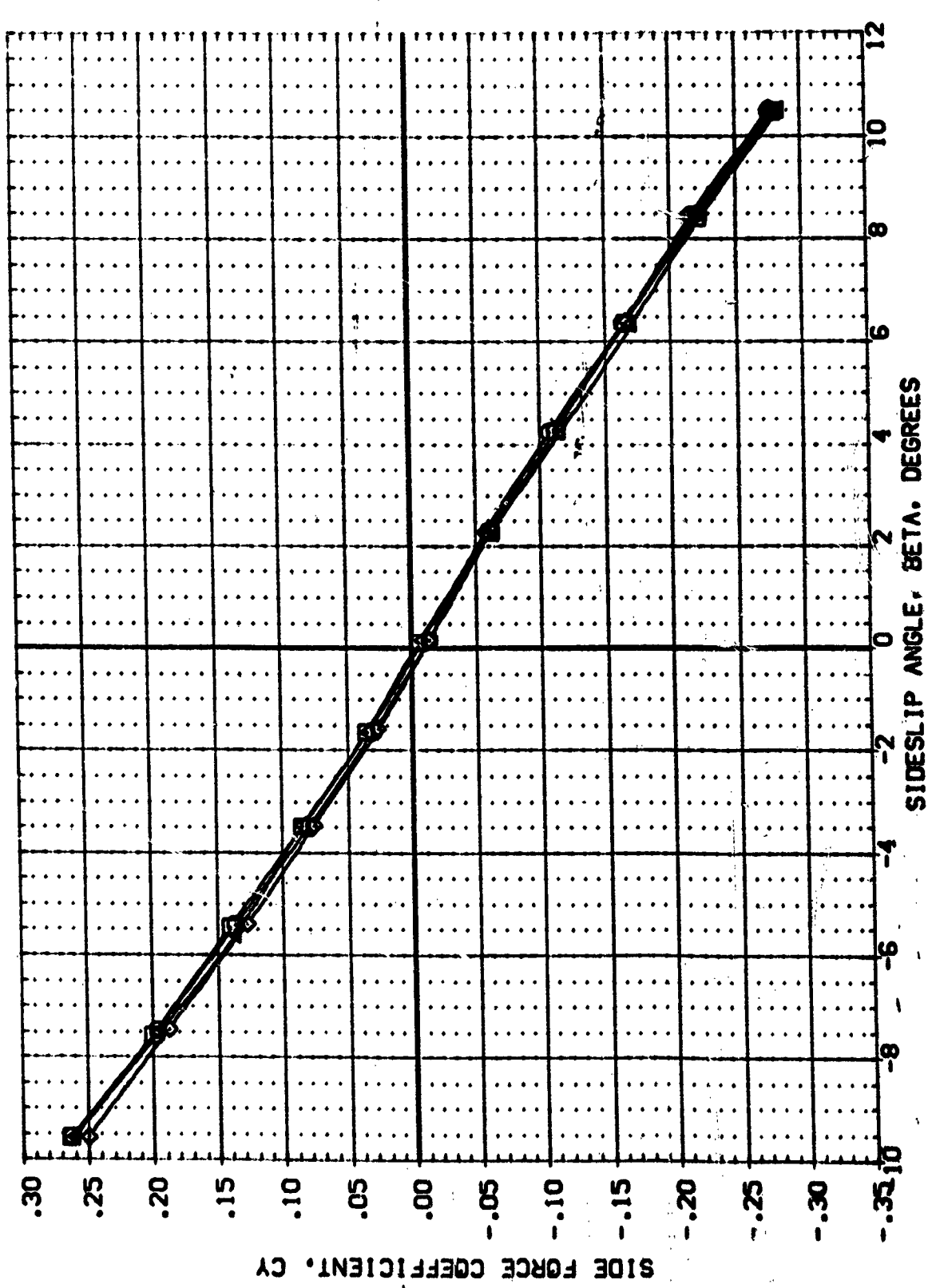


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBO	CONFIGURATION DESCRIPTION	ALPHA	AT10	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	YMRP 1076.4800 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

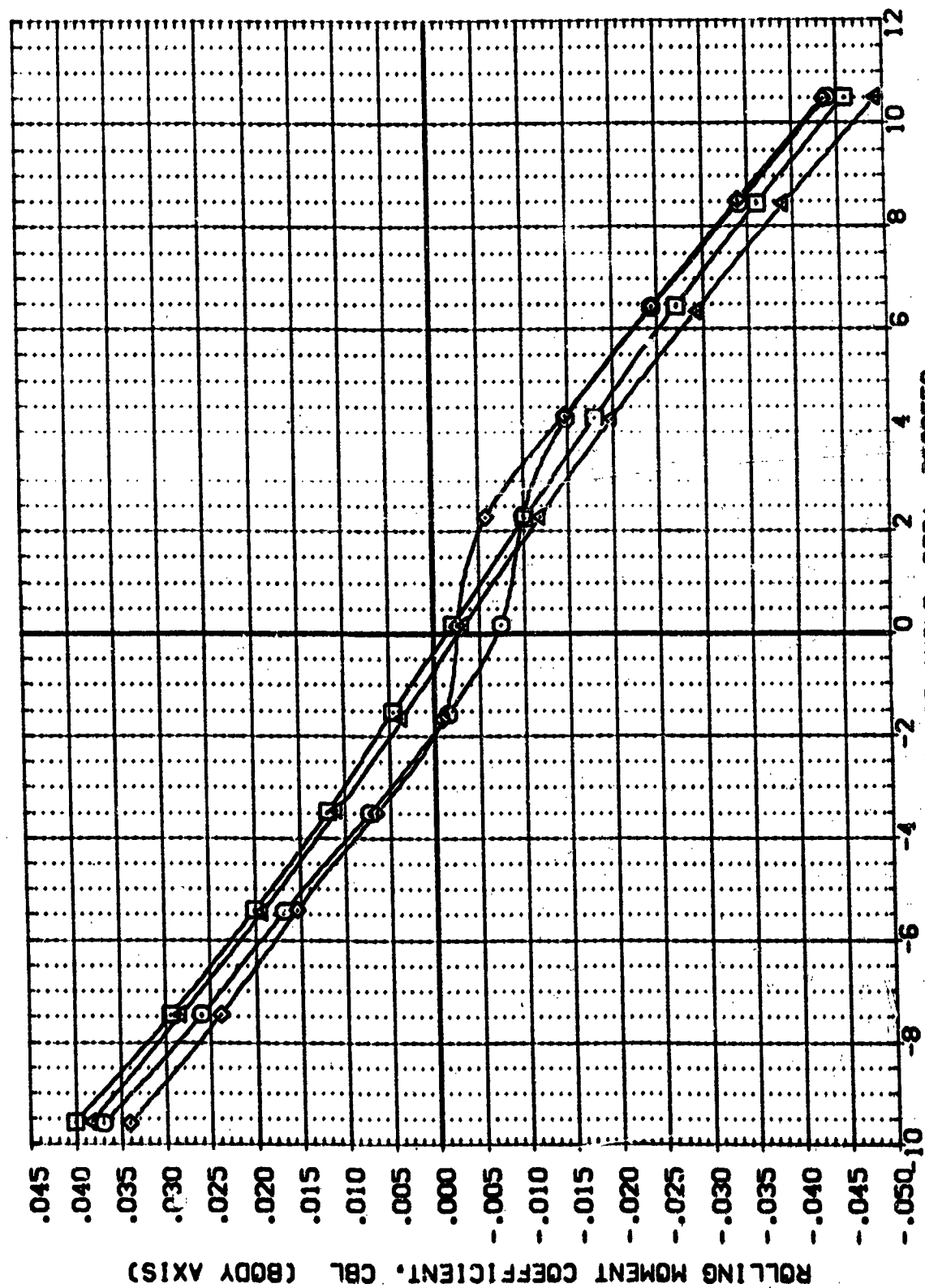


FIG. 6 EFFECTS OF SOLID PLOWES ON RUDDER EFFECTIVENESS.

(A)MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	AVES 3.5-169 IA10 OS T10 AT2 PLUVE ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	AVES 3.5-169 IA10 OS T10 AT2 PLUVE ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)	AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF	.000	.000	.000	.000	XREF 1076.4800 IN.
						YREF 400.0000 IN.
						ZREF 400.0000 IN.
						SCALE .0100

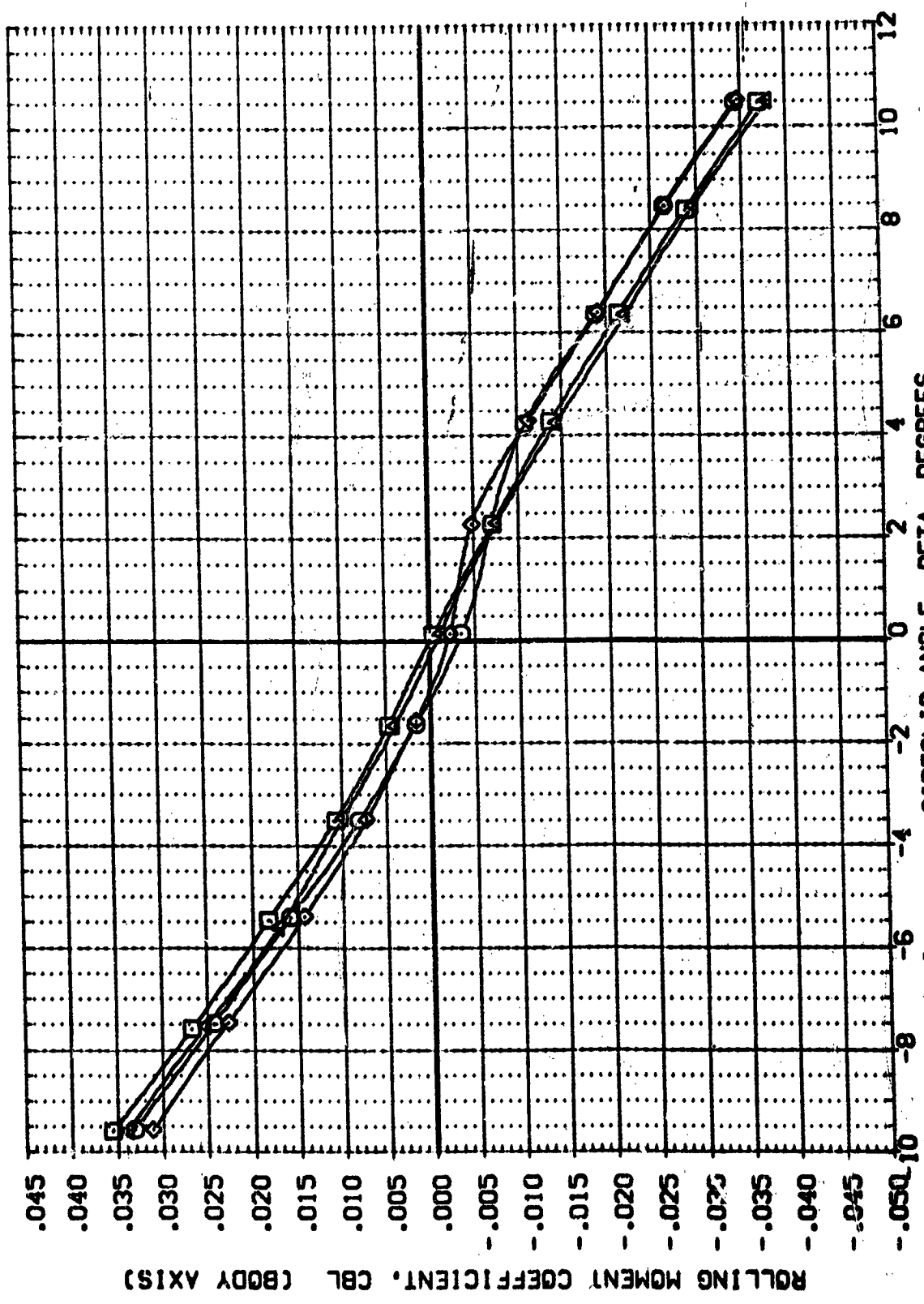


FIG. 6 EFFECTS OF SOLID PLOMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION				REFERENCE INFORMATION	
(R7008)	AVES 3.5-169	IA10	CS T10	AT2	PLUE	ON	SREF 2690.0000
(R7007)	AVES 3.5-169	IA10	CS T10	AT2	PLUE	OFF	LREF 1290.0000
(R7009)	AVES 3.5-169	IA1C	CS T10	AT2	PLUE	ON	BREF 936.6900
(R7006)	AVES 3.5-169	IA10	CS T10	AT2	PLUE	OFF	YREF 1076.4800
							ZREF 400.0000
							SCALE .0100

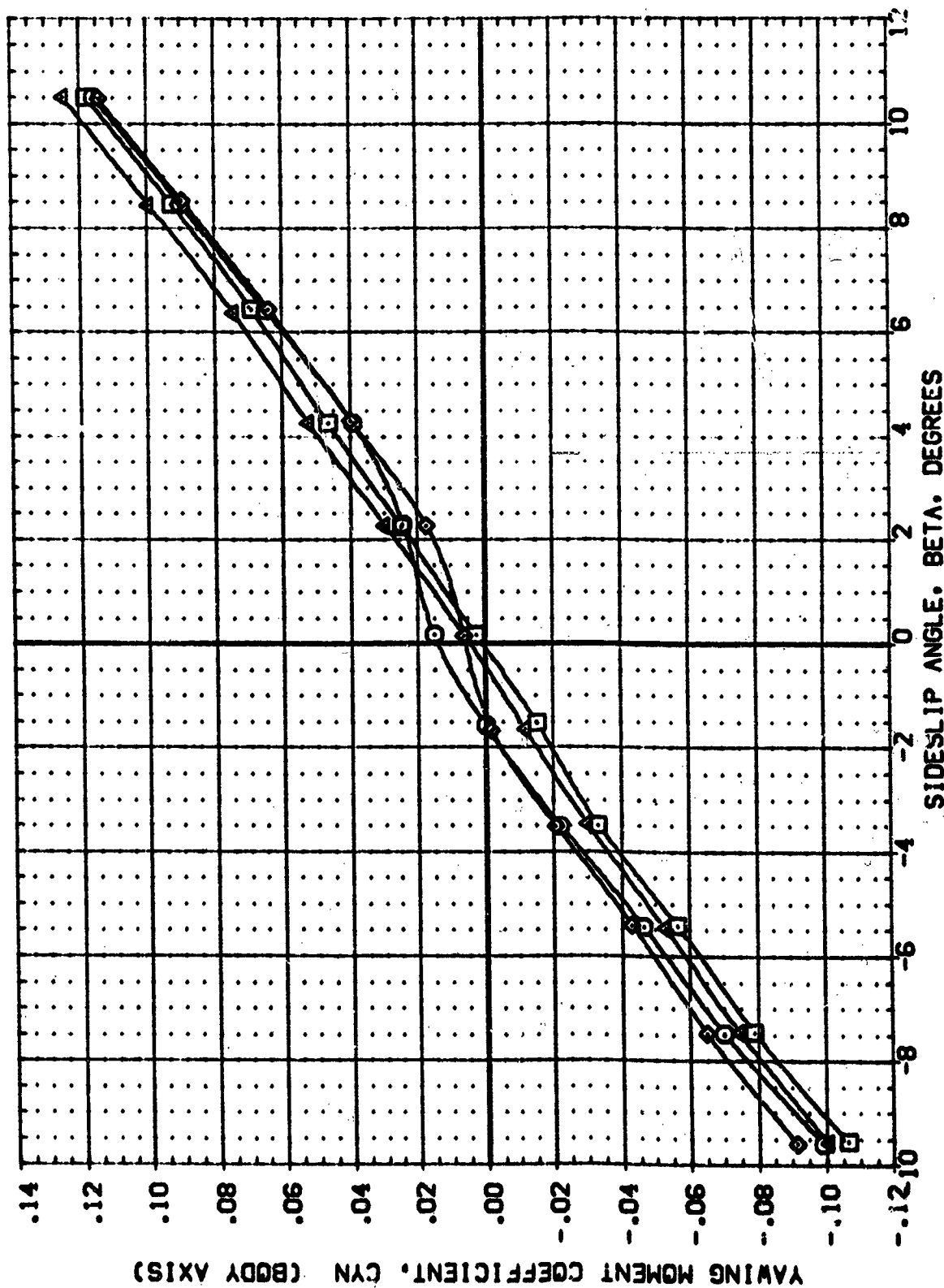


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(RB7008)	AVES 3.5-169 IA10 OS T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(RB7007)	AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(RB7009)	AVES 3.5-169 IA10 OS T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(RB7006)	AVES 3.5-169 IA10 OS T10 AT2 PLUME OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
						YMRP .0000 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

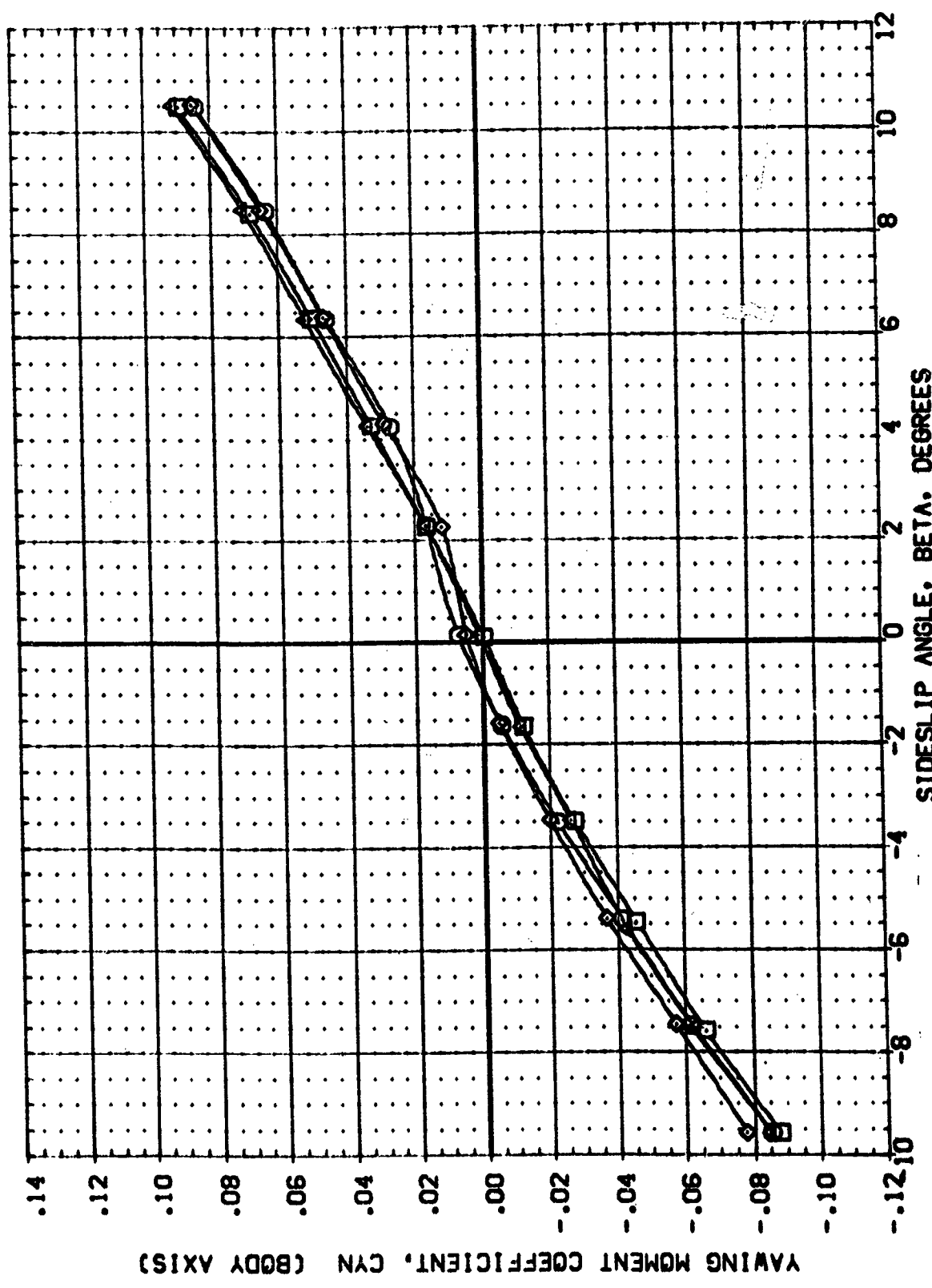


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87008)	□	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	○	AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87009)	×	AVES 3.5-169 IA10 09 T10 AT2 PLUME ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87006)		AVES 3.5-169 IA10 09 T10 AT2 PLUME OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
							YMRP .0300 IN.
							ZMRP 400.0300 IN.
							SCALE .0100

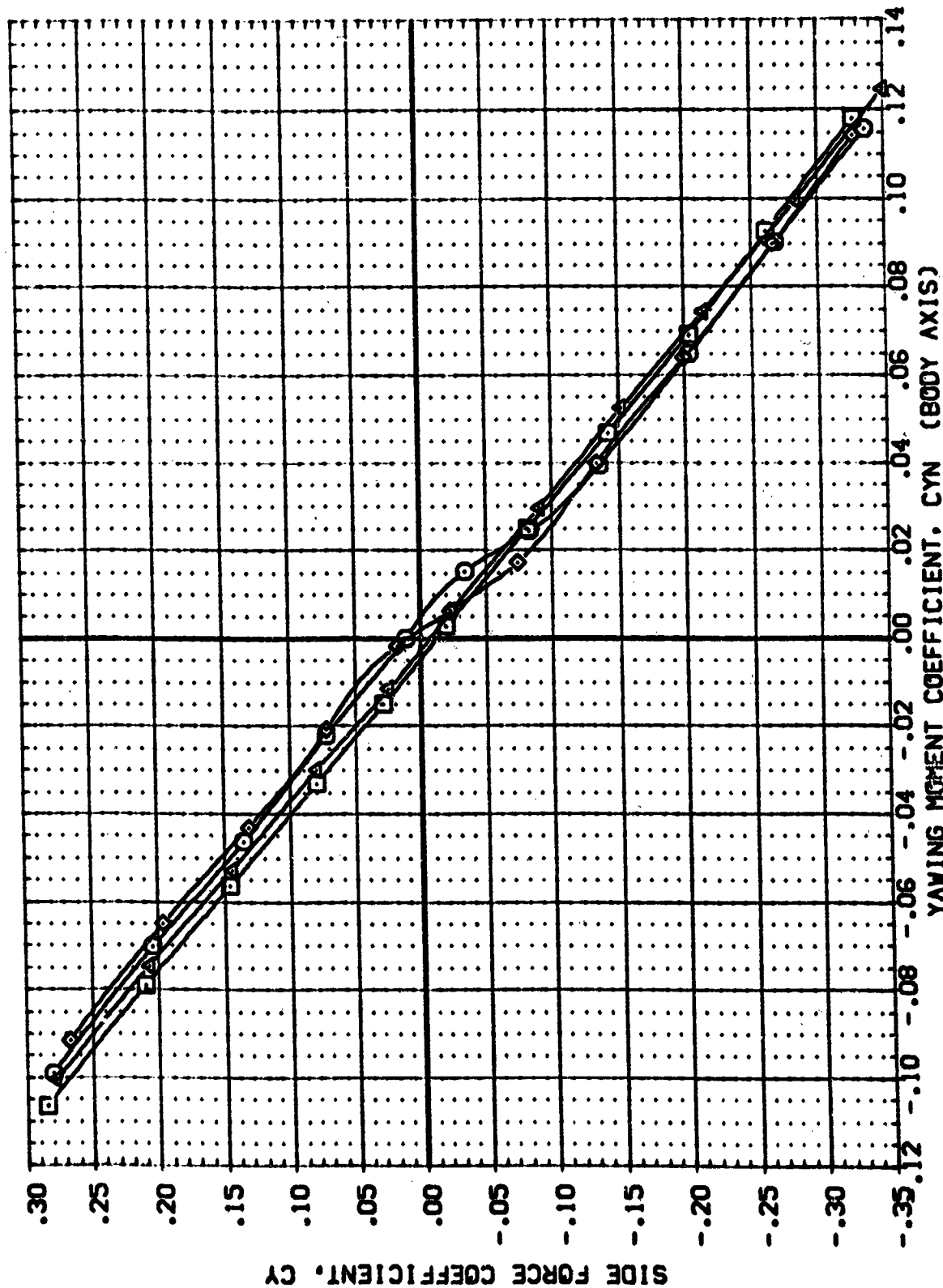


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

[A]MACH = 5.26



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87006)	AVES 3.5-169 IA10 CS T10 AT2 PLUVE ON	.000	.000	.000	10.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 CS T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
(R87008)	AVES 3.5-169 IA10 CS T10 AT2 PLUVE ON	.000	.000	.000	.000	BREF 936.6800 IN.
(R87009)	AVES 3.5-169 IA10 CS T10 AT2 PLUVE OFF	.000	.000	.000	.000	XMRP 1076.4800 IN.
(R87006)						ZMRP 400.0000 IN.
						SCALE .0100

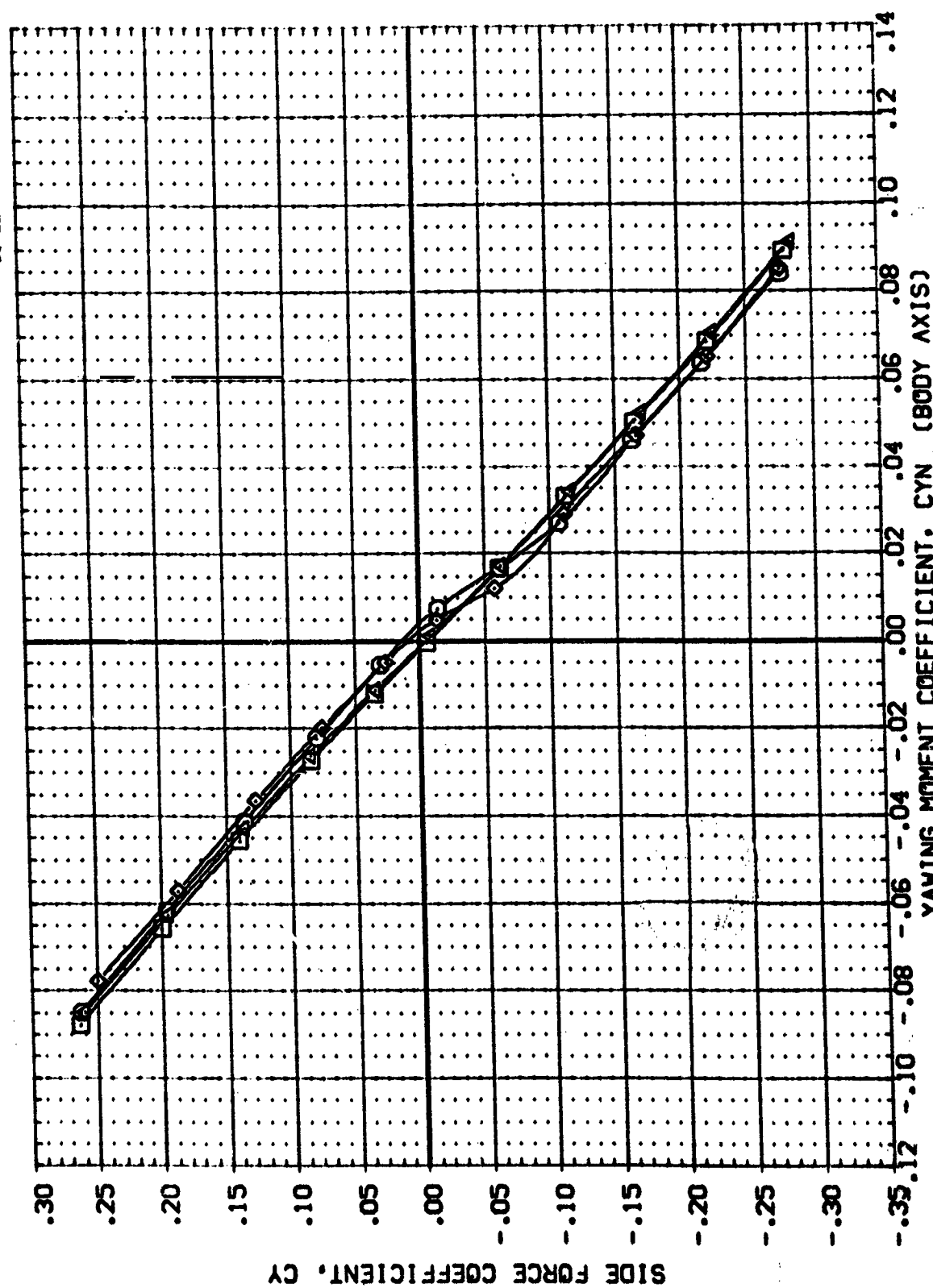


FIG. 6 EFFECTS OF SOLID PLUMES ON RUDDER EFFECTIVENESS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		BETA		AILRON		ELEVON		RUDDER		REFERENCE INFORMATION	
(087011)	□	AVES 3-5-169	IA10	09	110	AT2	PLUVE	OFF	.000	.000	.000	SREF	2690.0000
(157005)	×	AVES 3-5-169	IA10	09	110	AT2	PLUVE	OFF	.000	.000	.000	LREF	1290.0000
(157001)		DATA NOT AVAILABLE		.000	.000	.000	.000	.000	.000	.000	.000	BREF	936.6800
(157005)		DATA NOT AVAILABLE		.000	.000	.000	.000	.000	.000	.000	.000	XPRP	1076.4800
				.000	.000	.000	.000	.000	.000	.000	.000	YPRP	400.0000
												ZPRP	400.0000
												SCALE	.0100

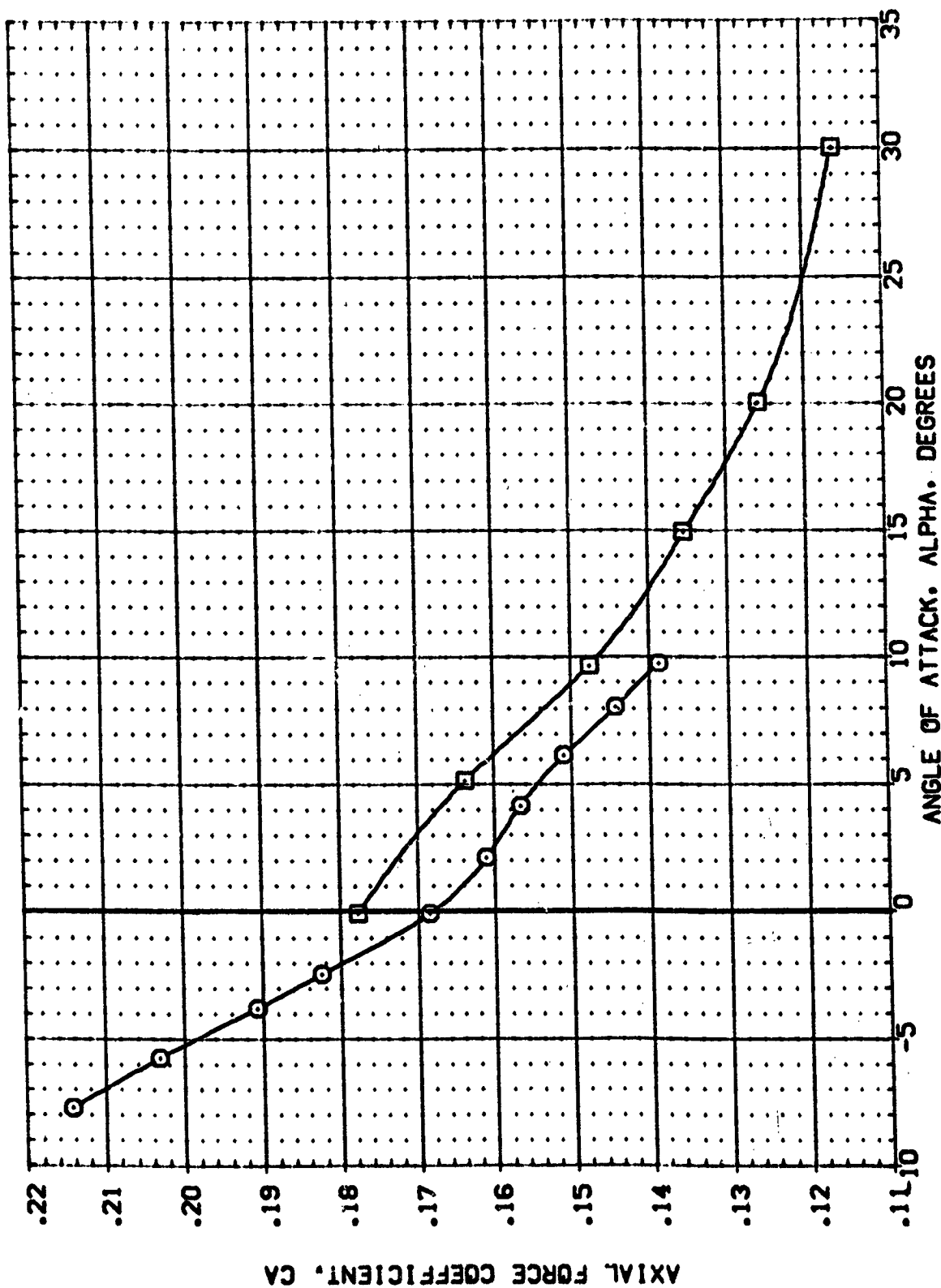


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A) MACH = 5.26

DATA NOT AVAILABLE	09 T10 AT2 PLUME OFF
AMES 3.5-169 1A10	09 T10 AT2 PLUME OFF
AMES 3.5-169 1A10	09 T10 AT2 PLUME OFF

DATA SET SYMBOL

REFERENCE INFORMATION	SO. FT.
SREF	2690.0000
REF	1290.0000
BREF	936.6800
XPRP	1076.4800
YPRP	.0000
ZPRP	400.0000
SCALE	.0100

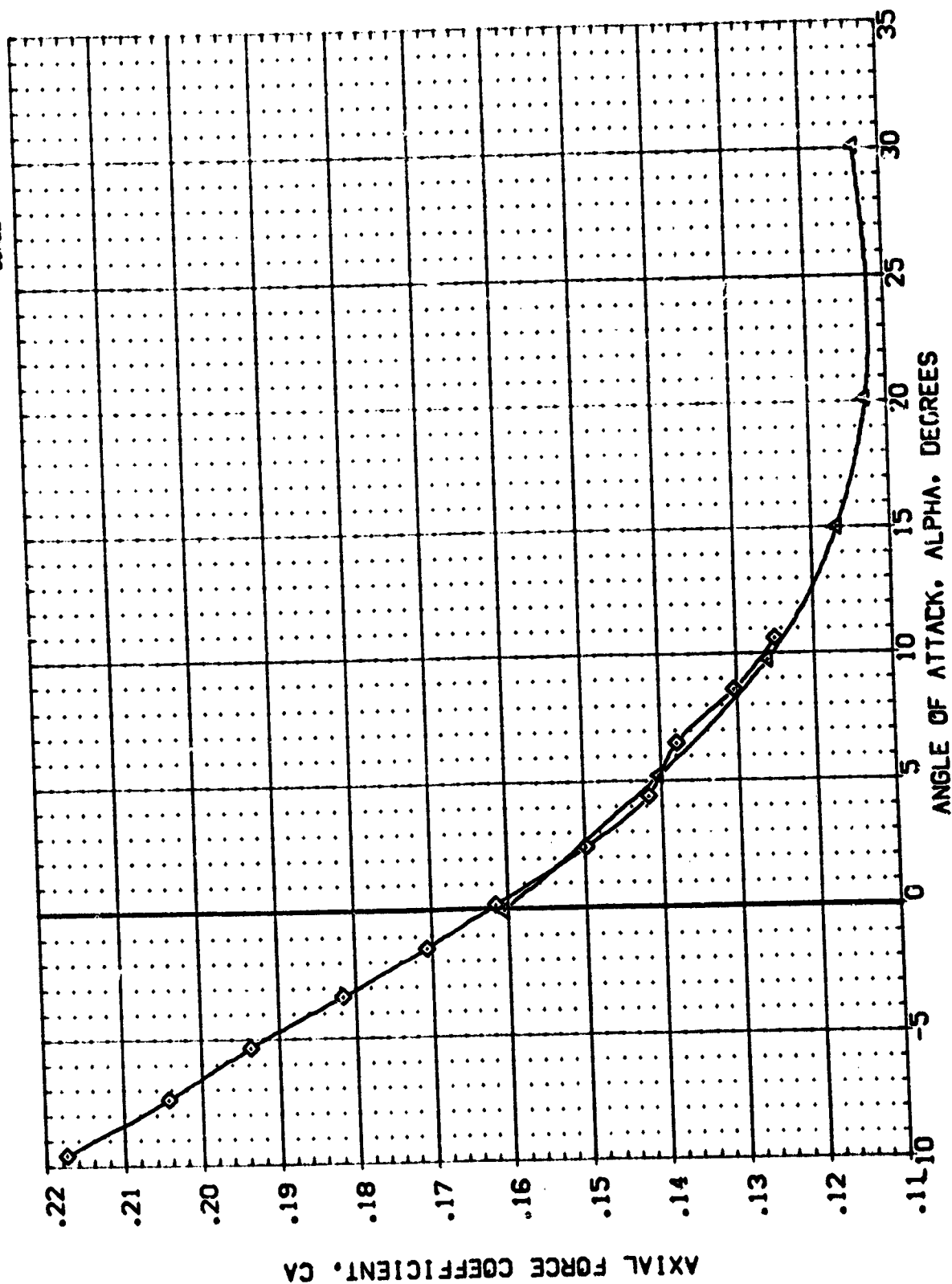


FIG. 7 EFFECT OF DIFFERENTIAL ELEVEN DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(087011)	AVES 3-5-169 1A10 09 T10 AT2 PLANE OFF	.000	.000	.000	.000	SREF 2630.0000 50.FT.
(087005)	AVES 3-5-169 1A10 09 T10 AT2 PLANE OFF	.000	10.000	.000	.000	LREF 1250.0000 IN.
(087001)	DATA NOT AVAILABLE	.000	10.000	.000	.000	BREF 936.6800 IN.
(187005)	DATA NOT AVAILABLE	.000	10.000	.000	.000	XREF 1076.4800 IN.
						YREF 400.0000 IN.
						ZREF 400.0000 IN.
						SCALE .0100

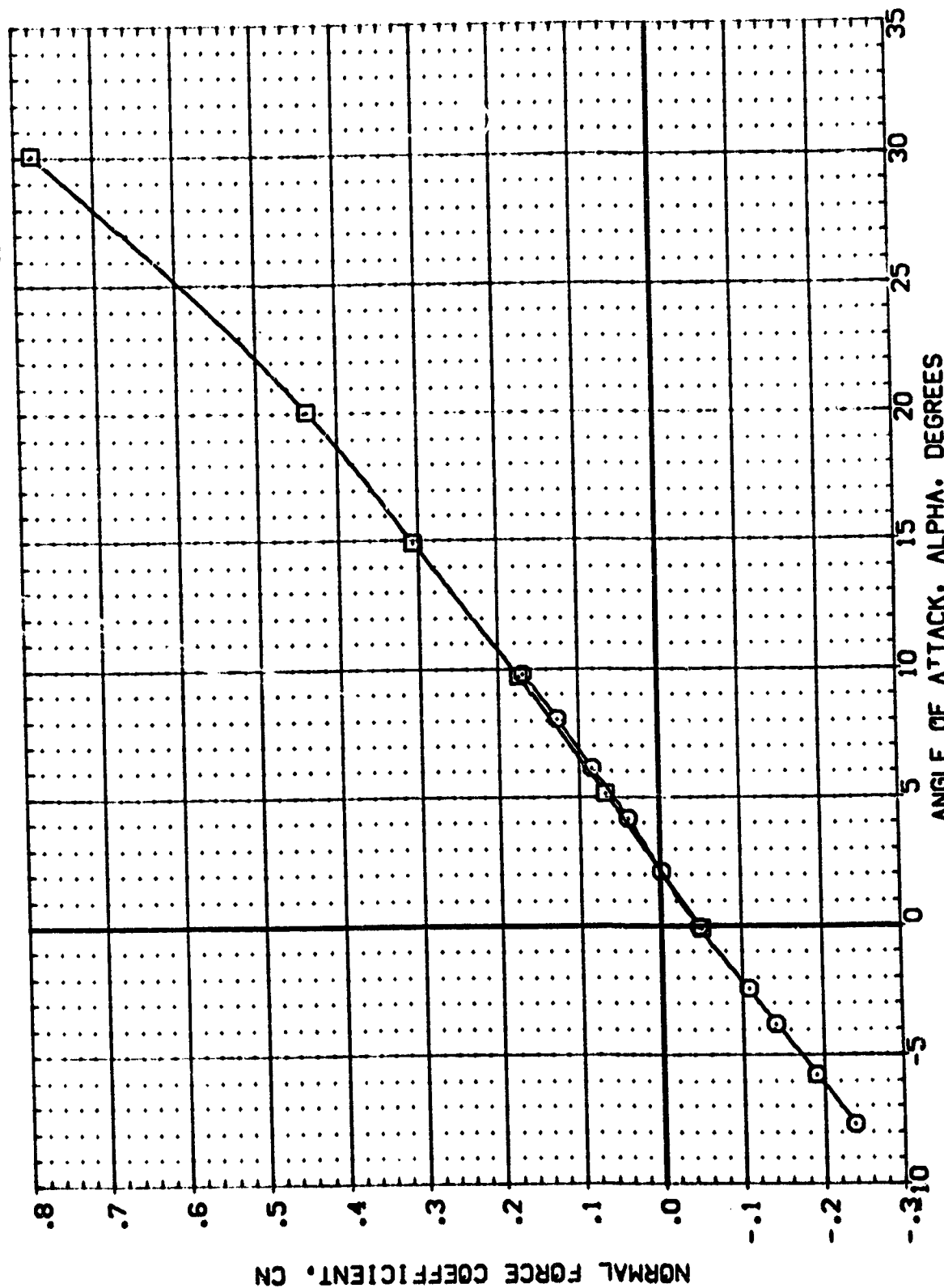


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(087011)	DATA NOT AVAILABLE	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(087005)	DATA NOT AVAILABLE	.000	.000	.000	.000	LREF 1250.0000 IN.
(087001)	AVES 3.5-169 1A10 09 T10 AT2 PLANE OFF	.000	10.000	.000	.000	BREF 936.6800 IN.
(187005)	AVES 3.5-169 1A10 09 T10 AT2 PLANE OFF	.000	10.000	.000	.000	YMRP 1076.4800 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

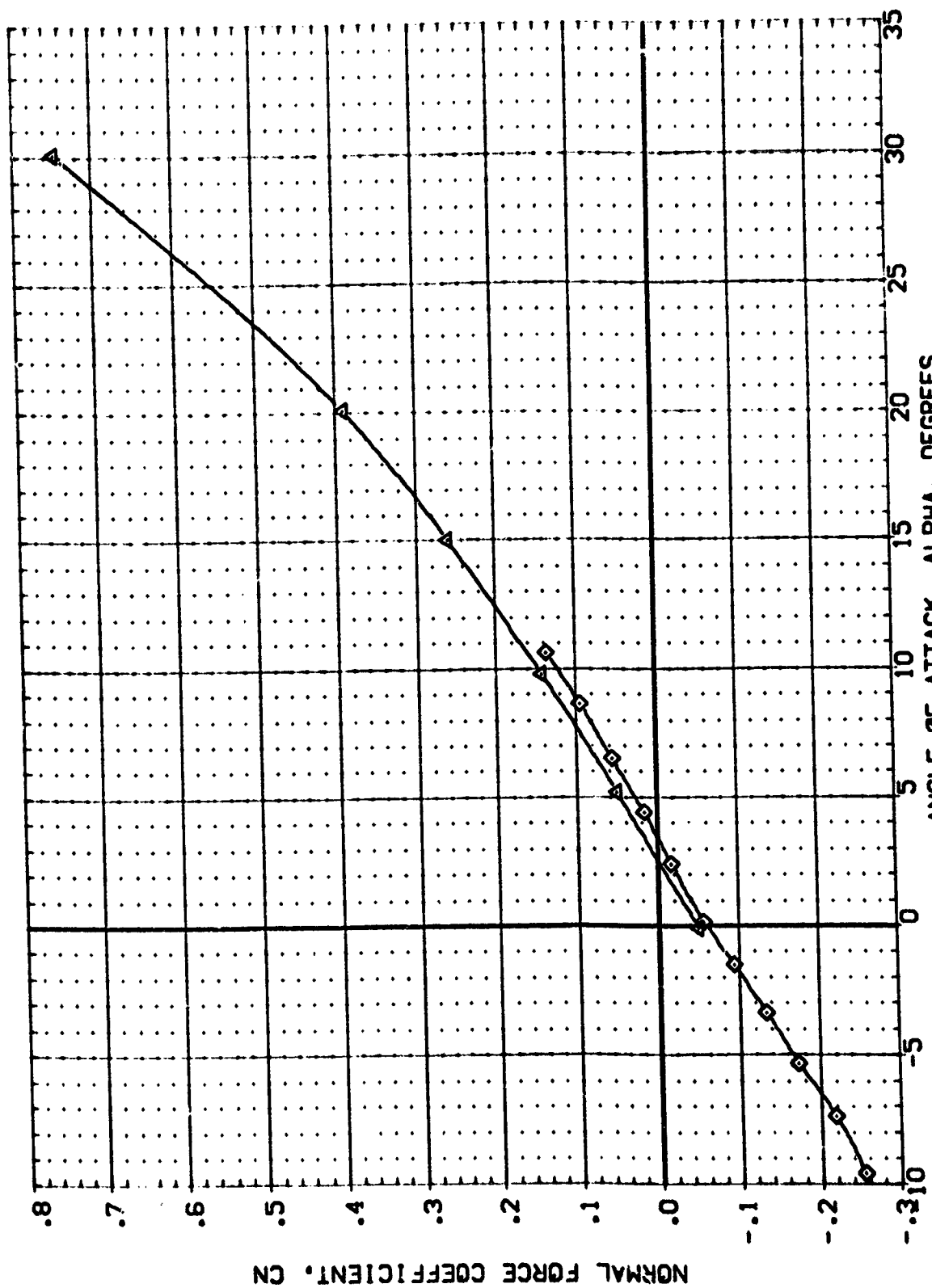


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL    CONFIGURATION DESCRIPTION  
 (DB7011)    AYES 3-169 IAI0 09 T10 AT2 PLANE OFF  
 (SB7005)    AYES 3-169 IAI0 09 T10 AT2 PLANE OFF  
 (RB7001)    DATA NOT AVAILABLE  
 (TB7005)    DATA NOT AVAILABLE

BETA    AILRON    ELEVON    RUDDER    REFERENCE INFORMATION  
 .000    .000    .000    .000    SREF 2690.0000 SO.FT.  
 .000    .000    .000    .000    LREF 1790.0000 IN.  
 .000    .000    .000    .000    BREF 536.6600 IN.  
 .000    .000    .000    .000    XMRP 1576.4600 IN.  
 .000    .000    .000    .000    YMRP 400.0000 IN.  
 .000    .000    .000    .000    ZMRP 400.0000 IN.  
 SCALE .0100

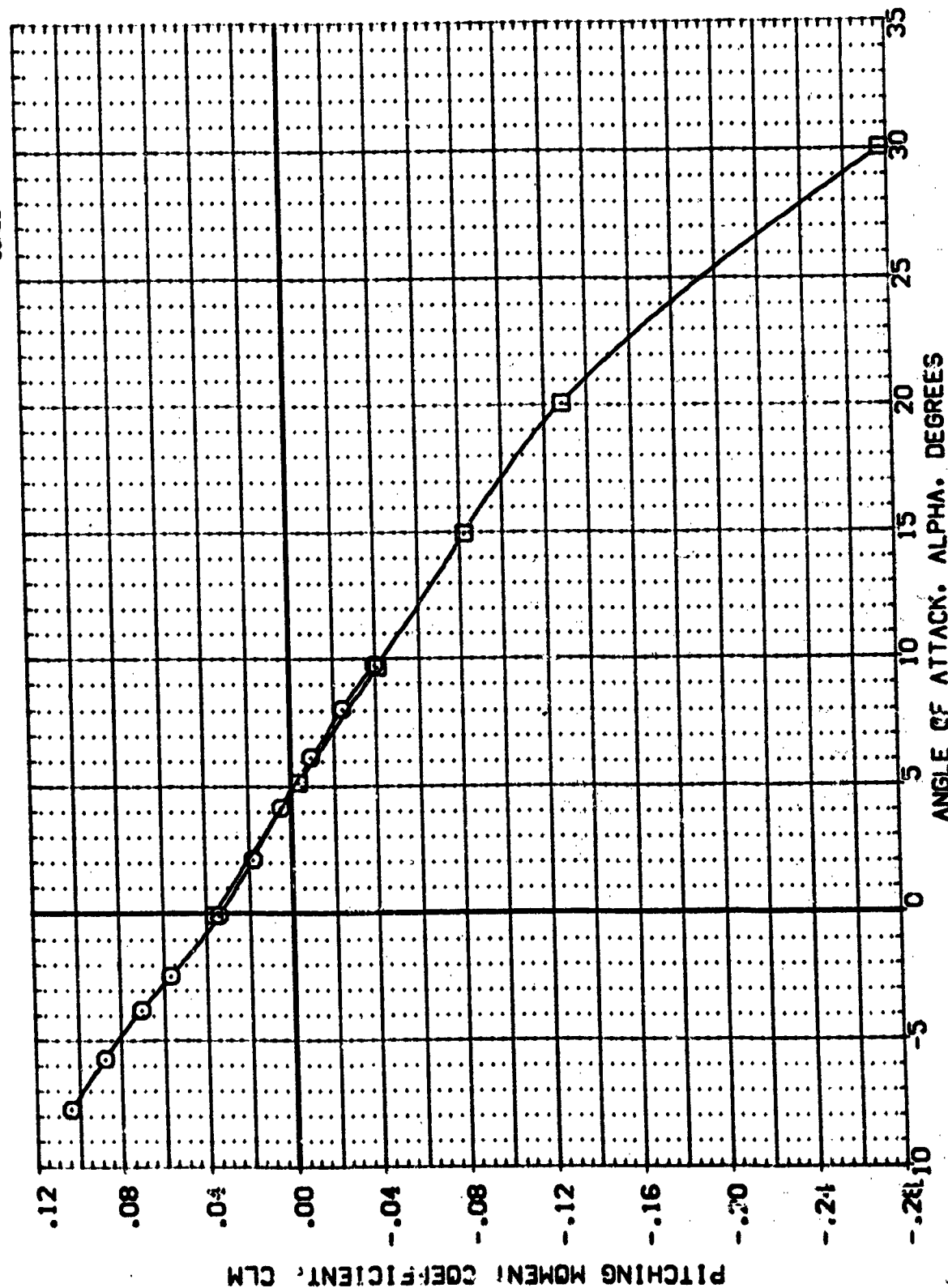


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(MACH = 5.26)

(DB701) DATA NOT AVAILABLE 09 110 AT2 PLANE OFF  
 (SB702) DATA NOT AVAILABLE 09 110 AT2 PLANE OFF  
 (RB703) AVES 3.5-169 1A10 09 110 AT2 PLANE OFF  
 (TB704) AVES 3.5-169 1A10 09 110 AT2 PLANE OFF

SREF 2690.0000 50.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6900 IN.  
 XMRP 1076.4800 IN.  
 YMRP 400.0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

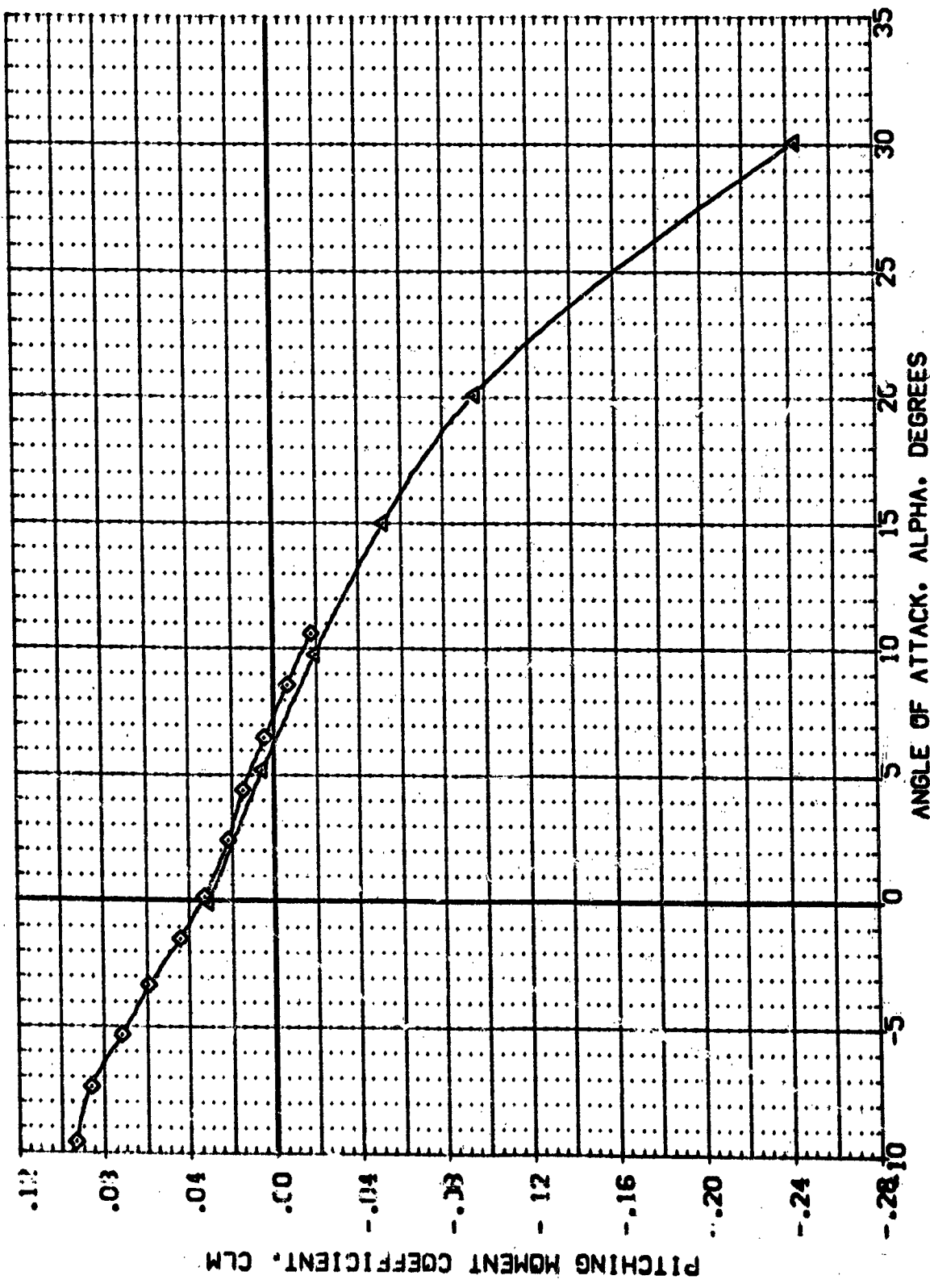


FIG. 7 EFFECT OF DIFFERENTIAL ELEVEN DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL: (187011) (187005) (187001) (187005)

CONFIGURATION DESCRIPTION: ARES 3.5-189 1A10 08 T10 AT2 PLUKE OFF  
 ARES 3.5-189 1A10 08 T10 AT2 PLUKE OFF  
 DATA NOT AVAILABLE

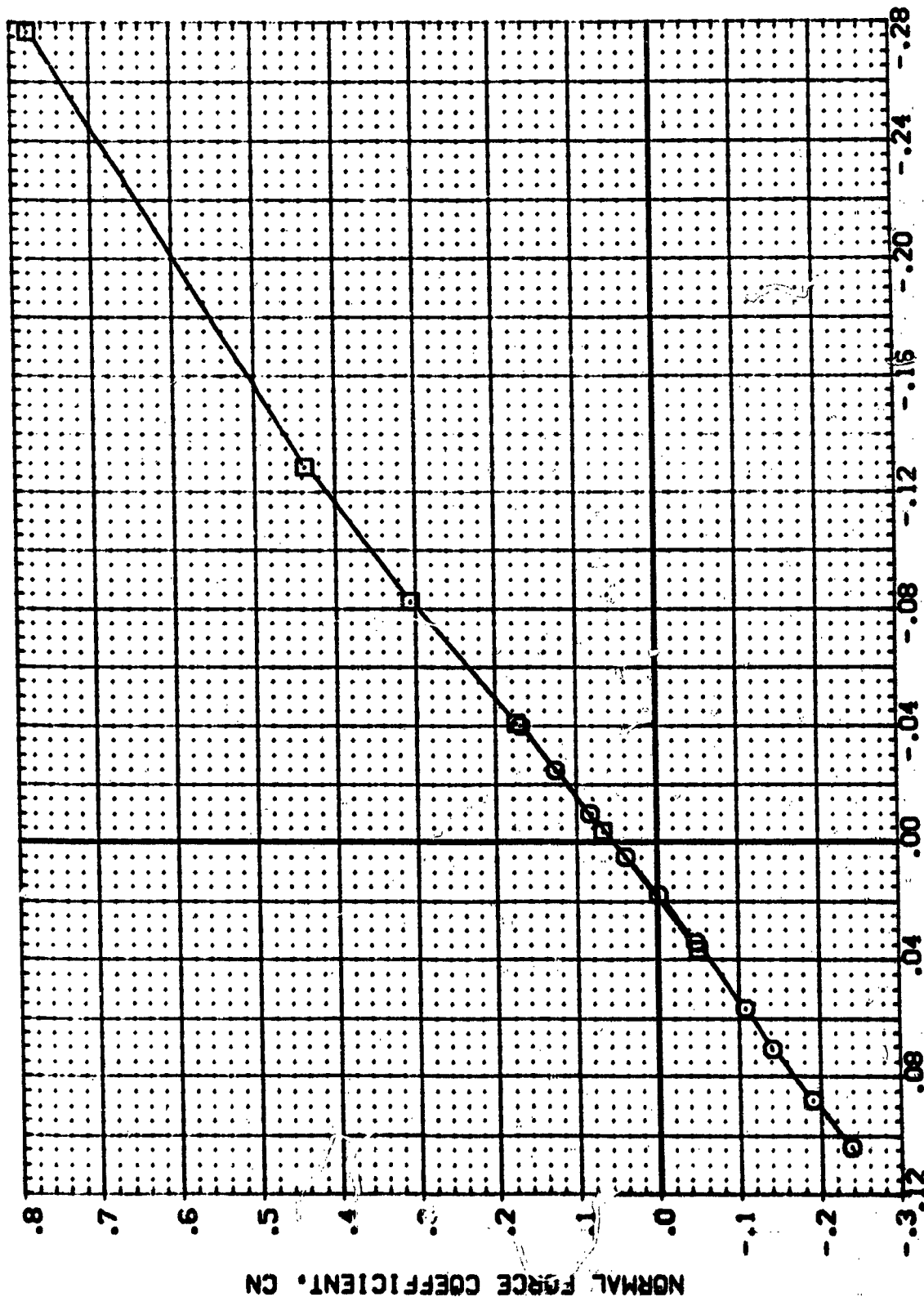
BETA: .000 .000 .000 .000

AIRLON: .000 10.000 10.000

ELEVON: .000 .000 .000 .000

RUDDER: .000 .000 .000 .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100



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FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26



# DATA SET SYMBOL CONFIGURATION DESCRIPTION

(187011) DATA NOT AVAILABLE  
 (187005) DATA NOT AVAILABLE  
 (187001) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 (187005) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF

BETA .000 .000 .000  
 AILRON .000 .000 .000  
 ELEVON .000 .000 .000  
 RUDDER .000 .000 .000  
 REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 YPRP 1076.4800 IN.  
 ZPRP 400.0000 IN.  
 SCALE .0100

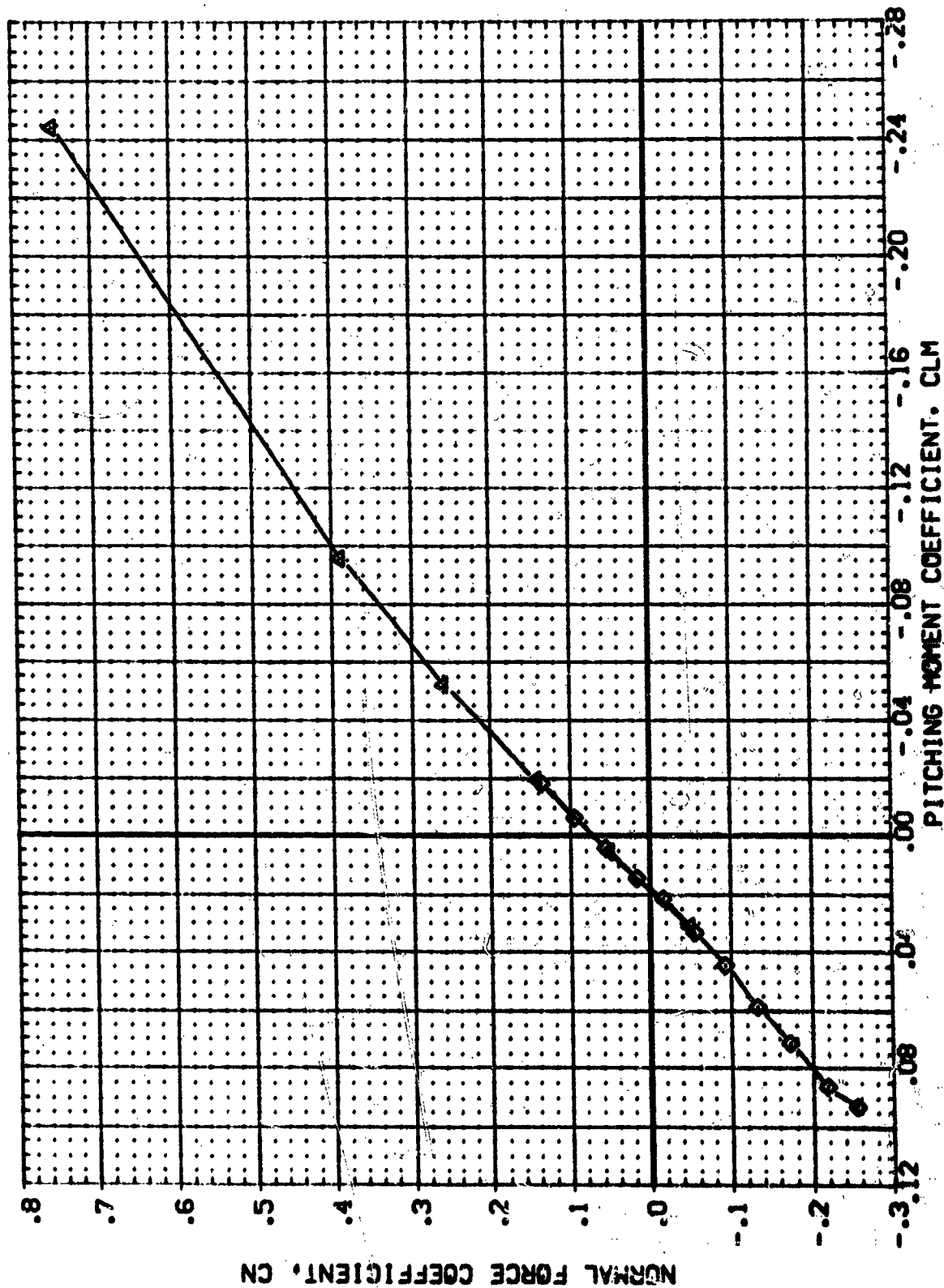


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32



# DATA SET SYMBOL CONFIGURATION DESCRIPTION

(087011) DATA NOT AVAILABLE  
 (587005) DATA NOT AVAILABLE  
 (887001) AYES 3.5-169 IA10 09 T10 AT2 PLUVE OFF  
 (187005) AYES 3.5-169 IA10 09 T10 AT2 PLUVE OFF

BETA AILRON ELEVON RUDDER REFERENCE INFORMATION  
 .000 .000 .000 SREF 2650.0000 SQ.FT.  
 .000 10.000 .000 LREF 1250.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 10.000 .000 XMRP 1076.4800 IN.  
 .000 .000 .000 YMRP 400.0000 IN.  
 .000 .000 .000 ZMRP 400.0000 IN.  
 SCALE .0100

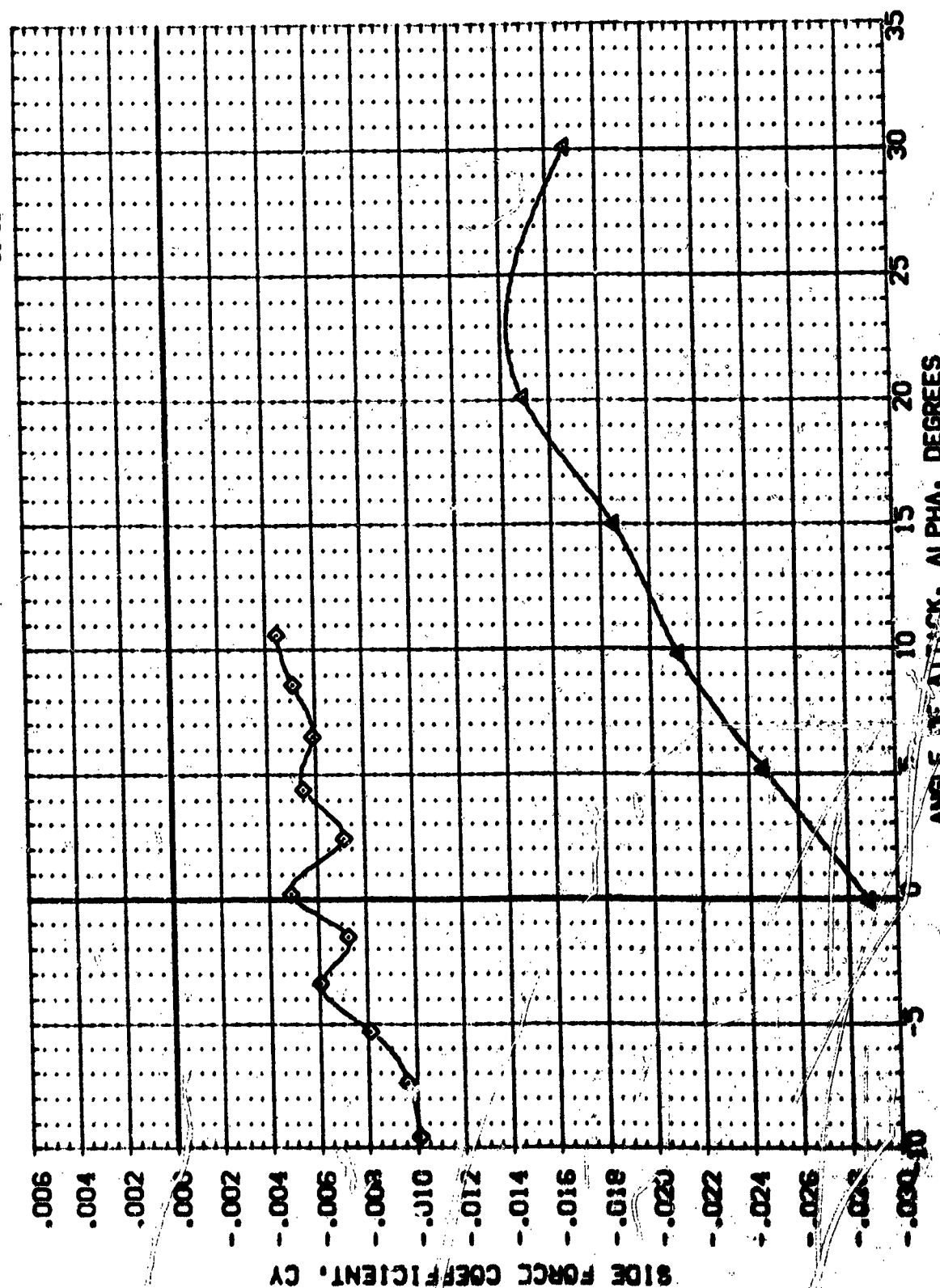


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

8 MARCH 7 32

DATA SET SYMBOL    CONFIGURATION DESCRIPTION    BETA    AILSON    ELEVON    RUDDER    REFERENCE INFORMATION

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILSON	ELEVON	RUDDER	REFERENCE INFORMATION
(087011)	WES 3-169 IAI0 08 T10 AT2 PLUPE OFF	.000	.000	.000	.000	SREF 2680.0000 SD.FT.
(S87005)	WES 3-5-169 IAI0 08 T10 AT2 PLUPE OFF	.000	.000	.000	.000	LREF 1290.0000 IN.
(R87001)	DATA NOT AVAILABLE	.000	.000	.000	.000	BREF 936.6800 IN.
(T87005)	DATA NOT AVAILABLE	.000	.000	.000	.000	YARP 1076.4800 IN.
						ZARP 400.0000 IN.
						SCALE .0100

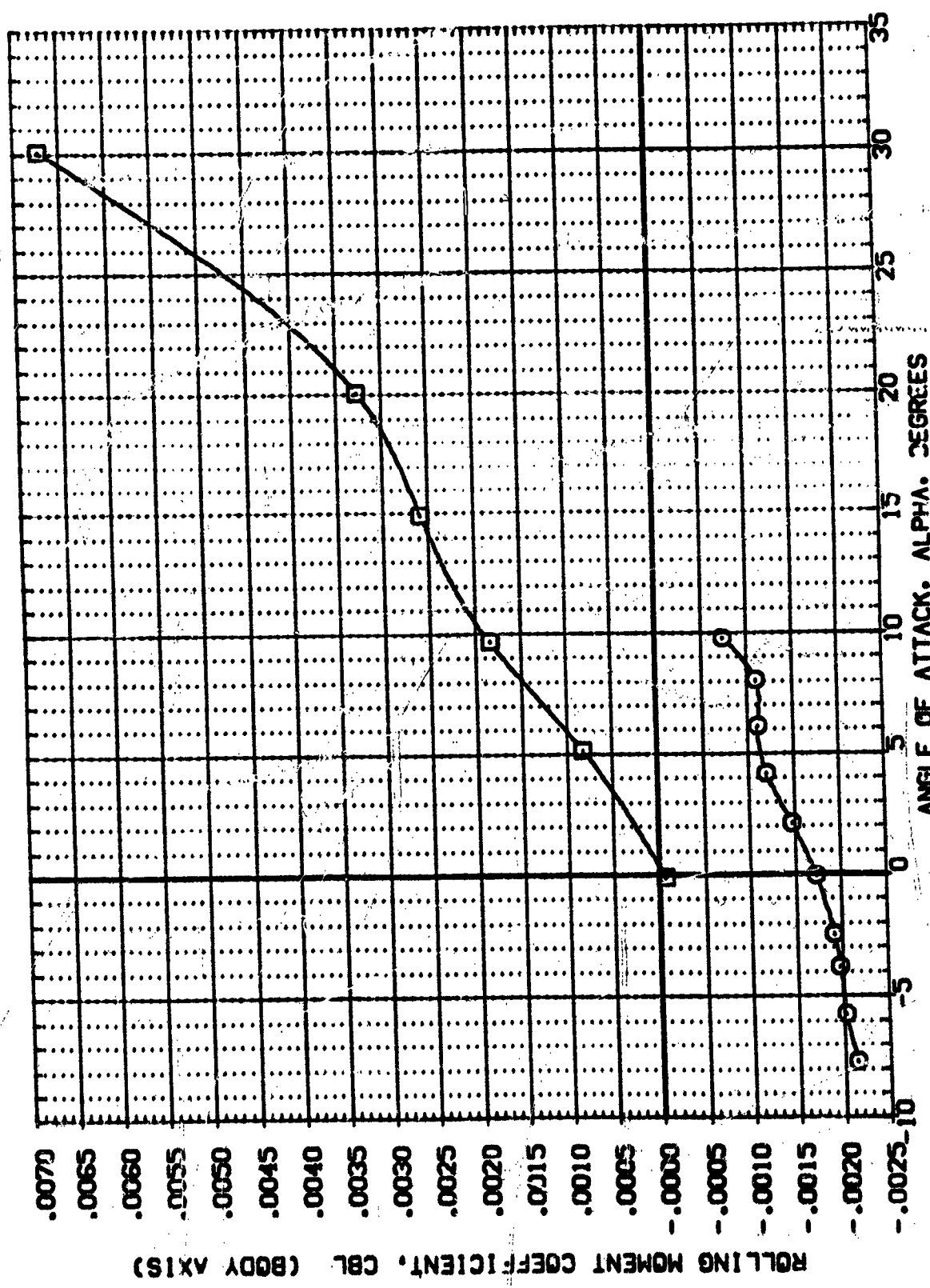


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

DATA SET SYMBOL:   
 (DB7011)   
 (SB7005)   
 (RB7001)   
 (TB7005)   
 CONFIGURATION DESCRIPTION:   
 DATA NOT AVAILABLE   
 AYES 3-5-169 IAI0   
 AYES 3-5-169 IAI0   
 CS T10 AT2 PLUVE OFF   
 CS T10 AT2 PLUVE OFF   
 REFERENCE INFORMATION:   
 SREF 2650.0000 SO.FT.   
 LRET 1250.0000 IN.   
 BREF 936.6800 IN.   
 XREF 1076.4800 IN.   
 YREF 400.0000 IN.   
 ZREF 400.0000 IN.   
 SCALE .0100   
 BETA .000   
 AIRCEN .000   
 ELEVON .000   
 RUDDER .000

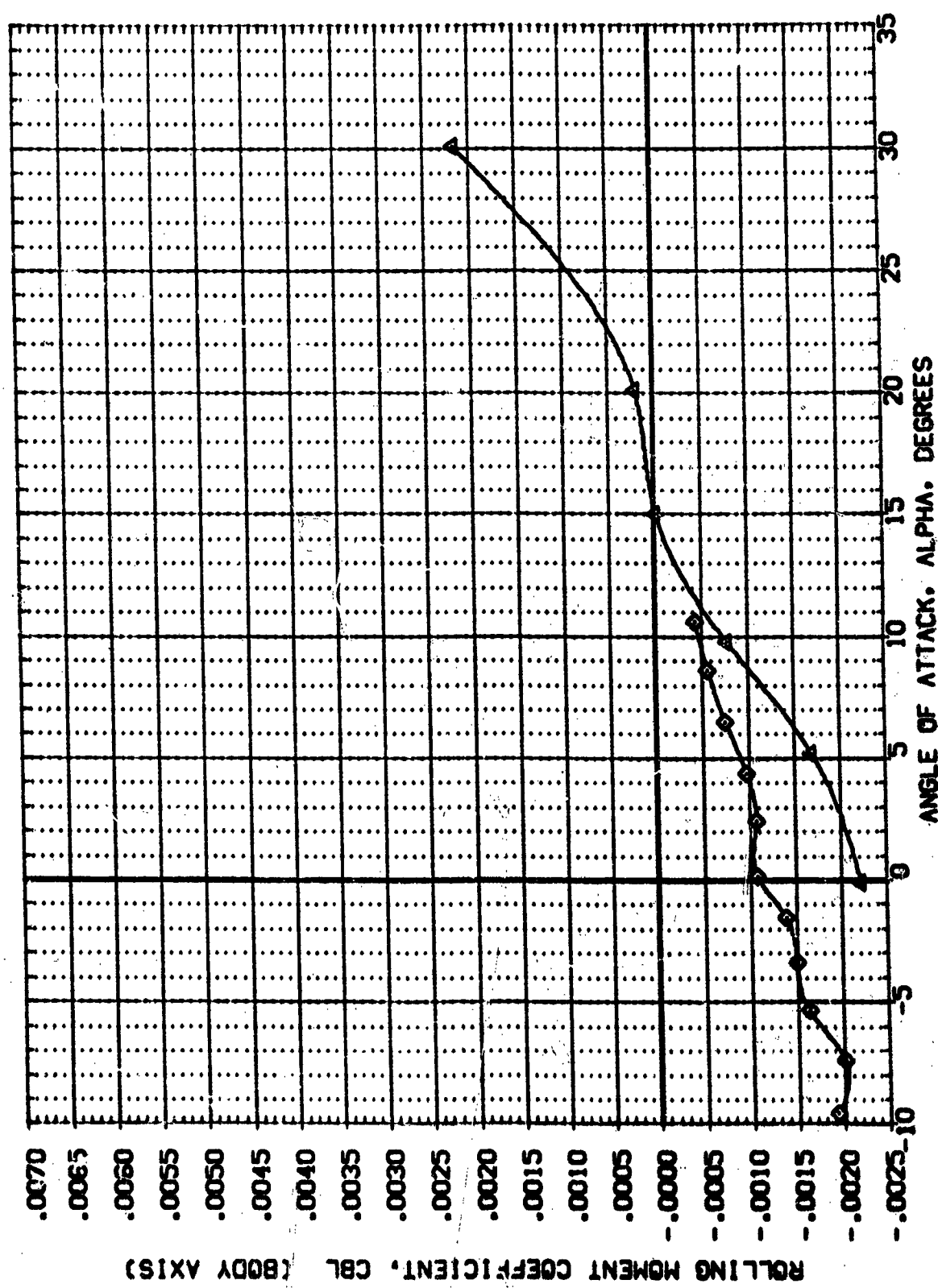


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION

Q AVES 3.5-169 1A10 09 T10 AT2 PLUVE OFF  
 (087011)  
 X AVES 3.5-169 1A10 09 T10 AT2 PLUVE OFF  
 (587005)  
 X DATA NOT AVAILABLE  
 (187001)  
 X DATA NOT AVAILABLE  
 (787005)

BETA AIRLON ELEVON RUDDER REFERENCE INFORMATION SO.FT.  
 .000 .000 .000 SREF 2680.0000 IN.  
 .000 .000 .000 LREF 1290.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 .000 .000 XMRP 1076.4800 IN.  
 .000 .000 .000 YMRP 400.0000 IN.  
 .000 .000 .000 ZMRP 400.0000 IN.  
 SCALE .0100

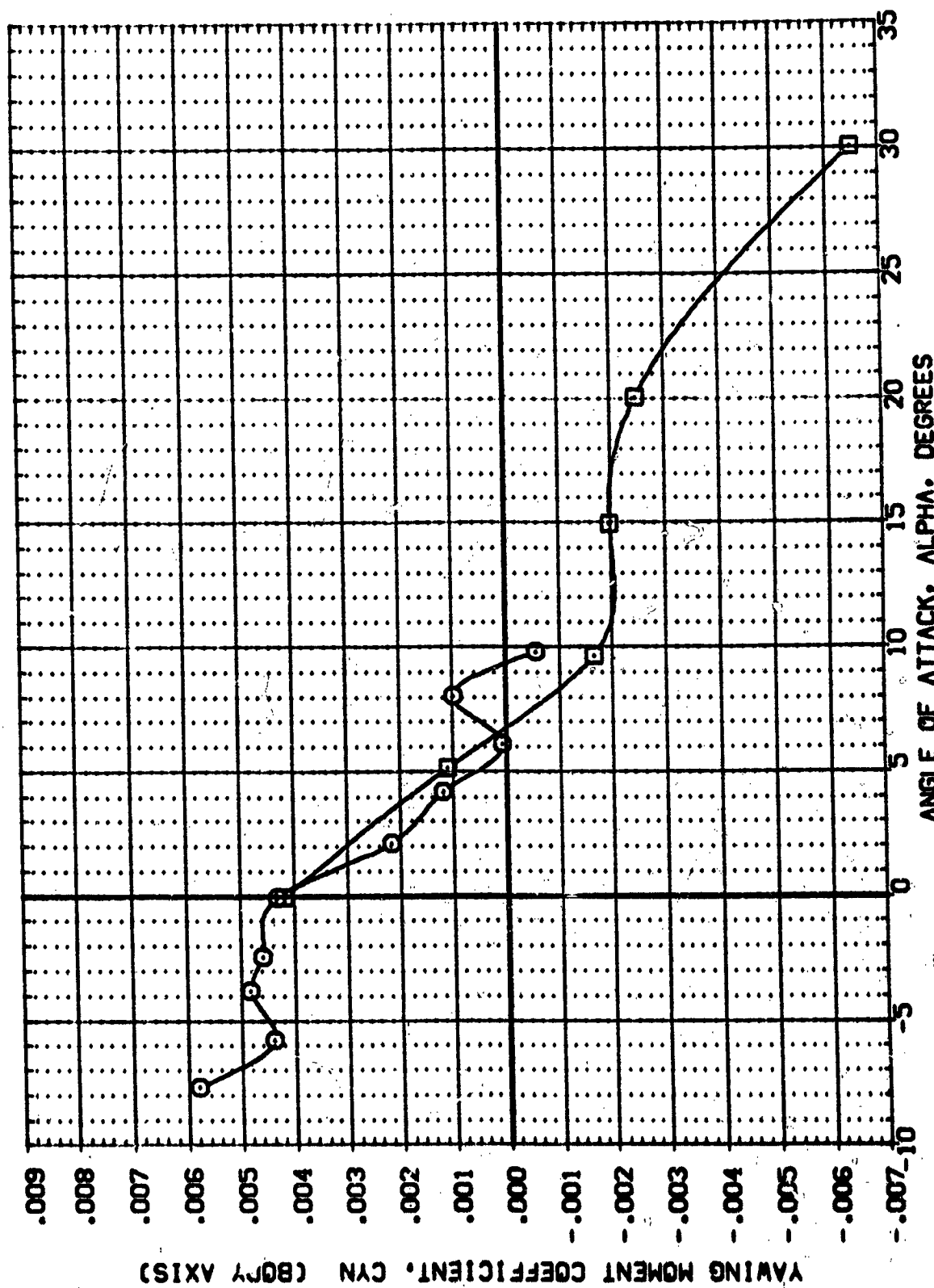


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(A)MACH = 5.26

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(087011) DATA NOT AVAILABLE  
 (587005) DATA NOT AVAILABLE  
 (587005) AVES 3.5-189 1A10 09 T10 AT2 PLUVE OFF  
 (187005) AVES 3.5-189 1A10 09 T10 AT2 PLUVE OFF

BETA AILRON ELEVON RUDDER REFERENCE INFORMATION  
 .000 .000 .000 SREF 2690.0000 SQ.FT.  
 .000 .000 .000 LREF 1290.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 .000 .000 YPRP 1076.4800 IN.  
 .000 .000 .000 ZPRP 400.0000 IN.  
 SCALE .0100

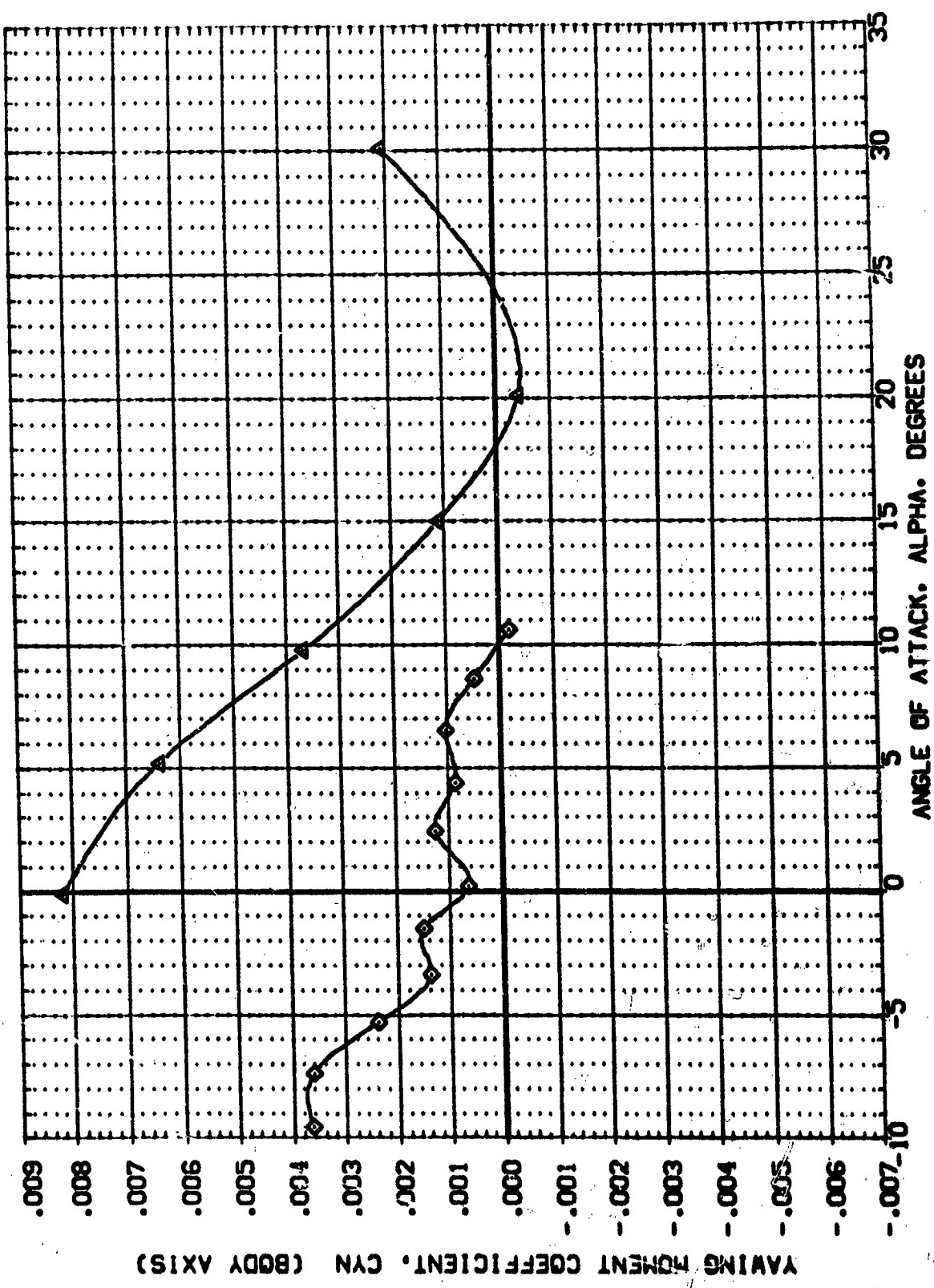


FIG. 7 EFFECT OF DIFFERENTIAL ELEVON DEFLECTION.

(B)MACH = 7.32

DATA SET SYMBOL: AVES 3.5-169 1A10 09 T10 AT2 PLUME OFF  
 (REF7006) ☐ AVES 3.5-169 1A10 09 T10 AT2 PLUME OFF  
 (REF7007) ☐

ALPHA: .000 .000 .000  
 AIRLON: .000 .000 .000  
 ELEVON: .000 .000 .000  
 RUDDER: .000 10.000 .000

REFERENCE INFORMATION:  
 SREF: 2690.0000 SQ.FT.  
 LREF: 1250.0000 IN.  
 BREF: 956.6800 IN.  
 XMRP: 1076.4800 IN.  
 YMRP: .0000 IN.  
 ZMRP: 400.0000 IN.  
 SCALE: .0100

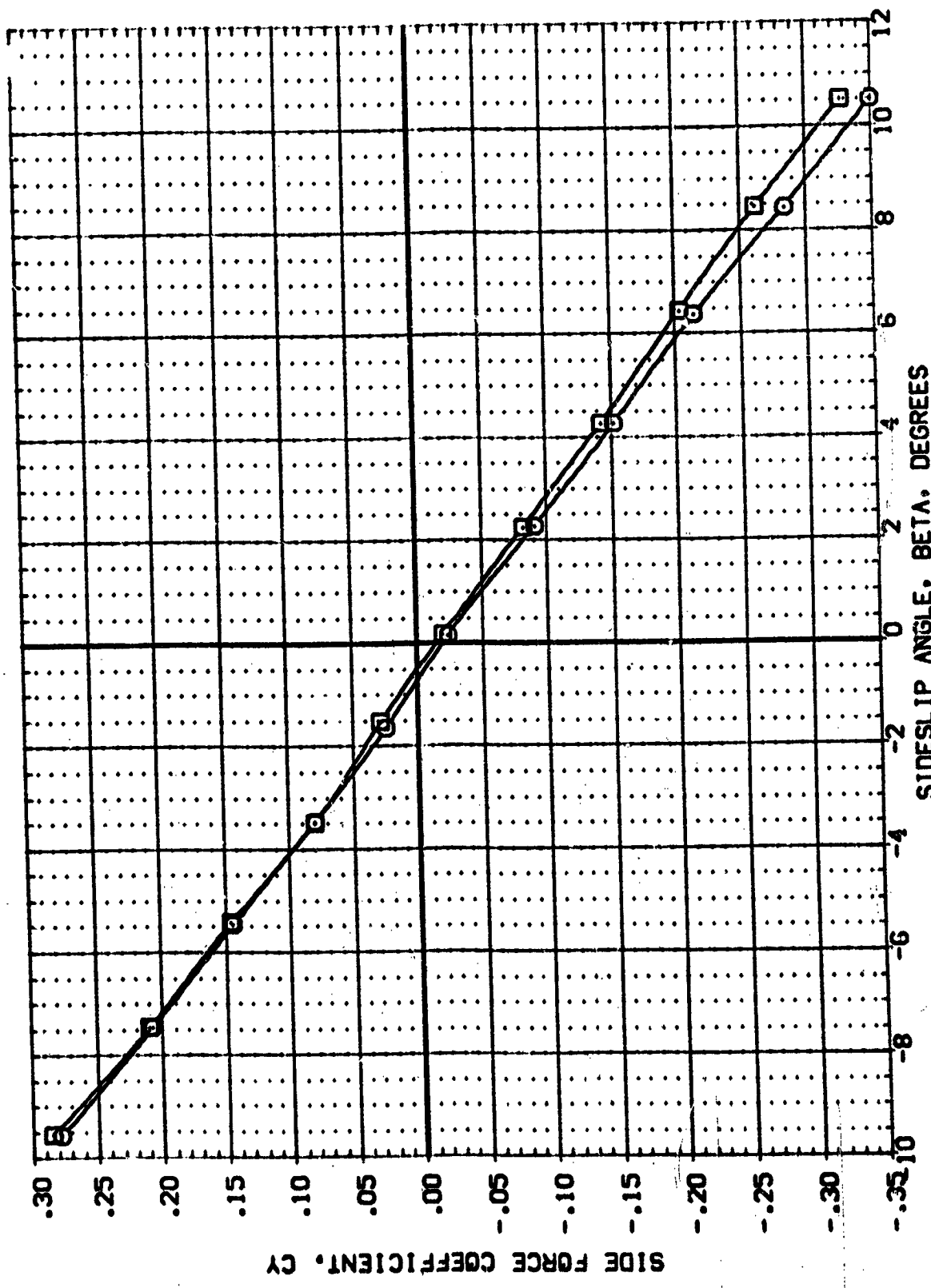


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(R87005)	AVES 3.5-169 IA10 09 T10 AT2 PLIVE OFF	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(R87007)	AVES 3.5-169 IA10 09 T10 AT2 PLIVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						YPRP 1076.4800 IN.
						ZPRP 400.0000 IN.
						SCALE .0100

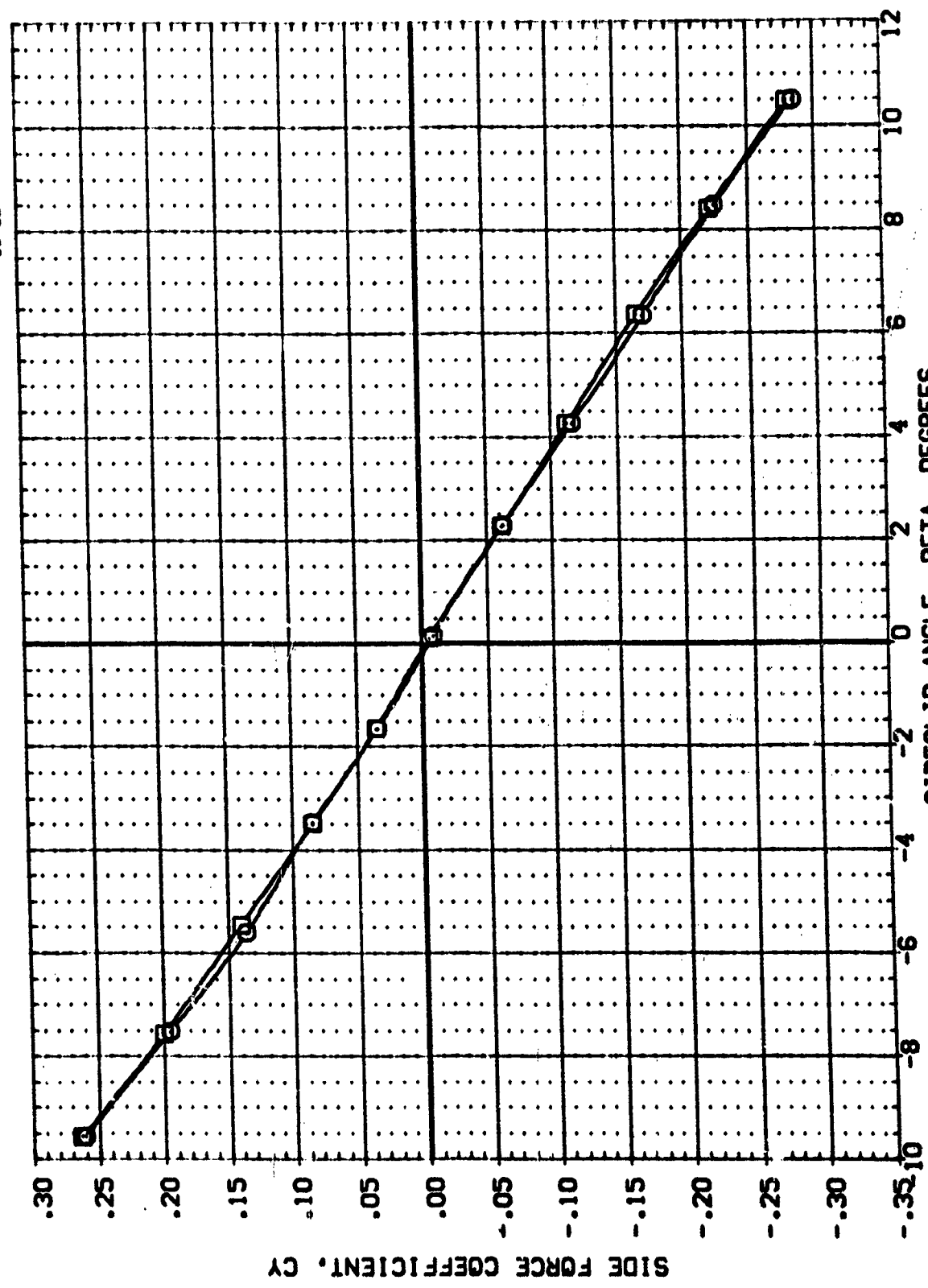


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION
(RB7006)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	SREF 1290.0000 SO.FT.
(RB7007)	AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1290.0000 IN.
						BREF 936.6800 IN.
						YPRP 1376.4800 IN.
						ZPRP 400.0000 IN.
						SCALE 400.0000
						0100

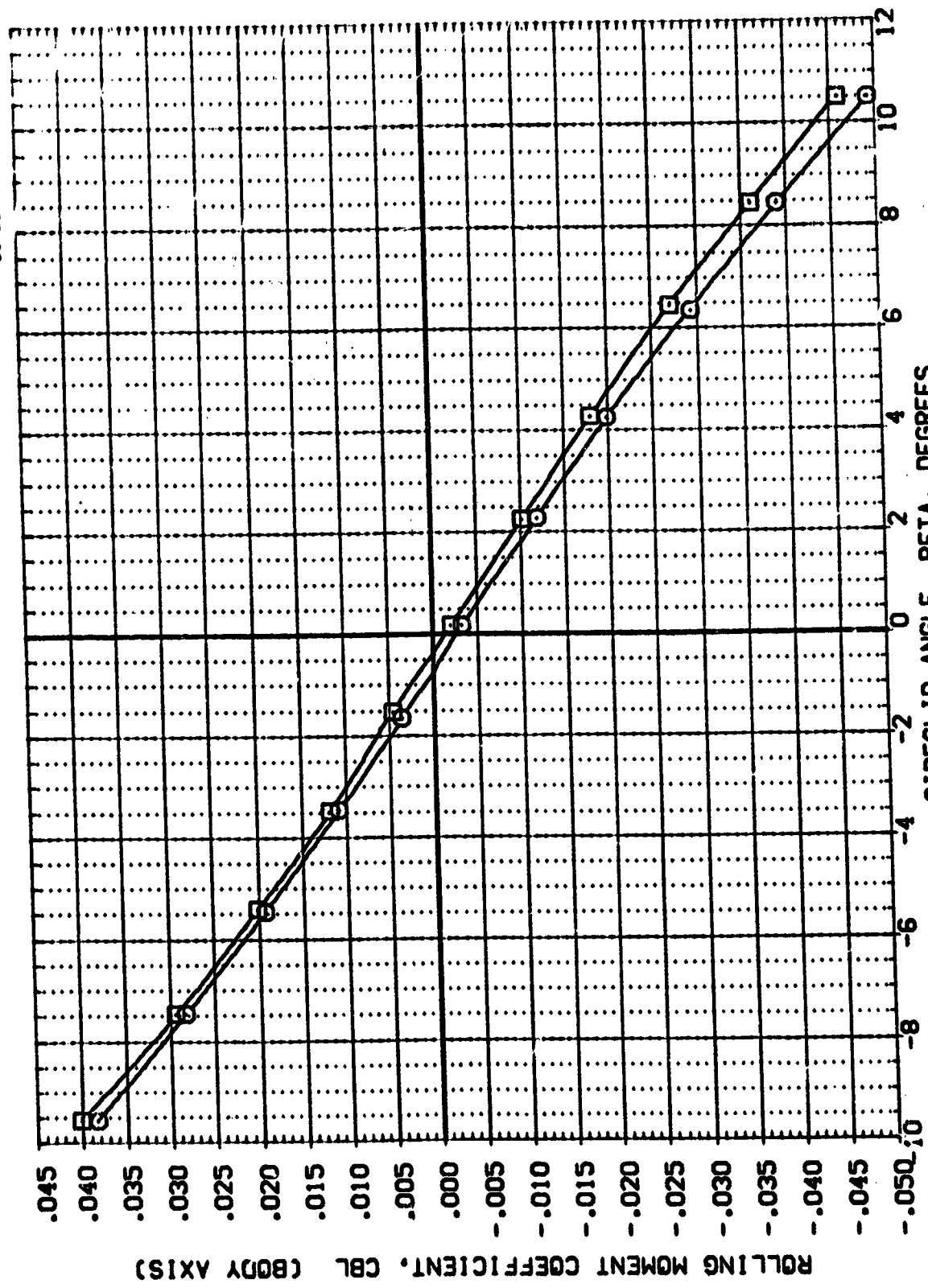


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION
(RB7006)	AMES 3.5-169 1A10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(RB7007)	AMES 3.5-169 1A10 09 T10 AT2 PLUVE OFF	.000	.000	.000	10.000	LREF 1790.0000 IN.
						BREF 936.6800 IN.
						VMRP 1076.4800 IN.
						ZMRP .0000 IN.
						SCALE 400.0000 IN.

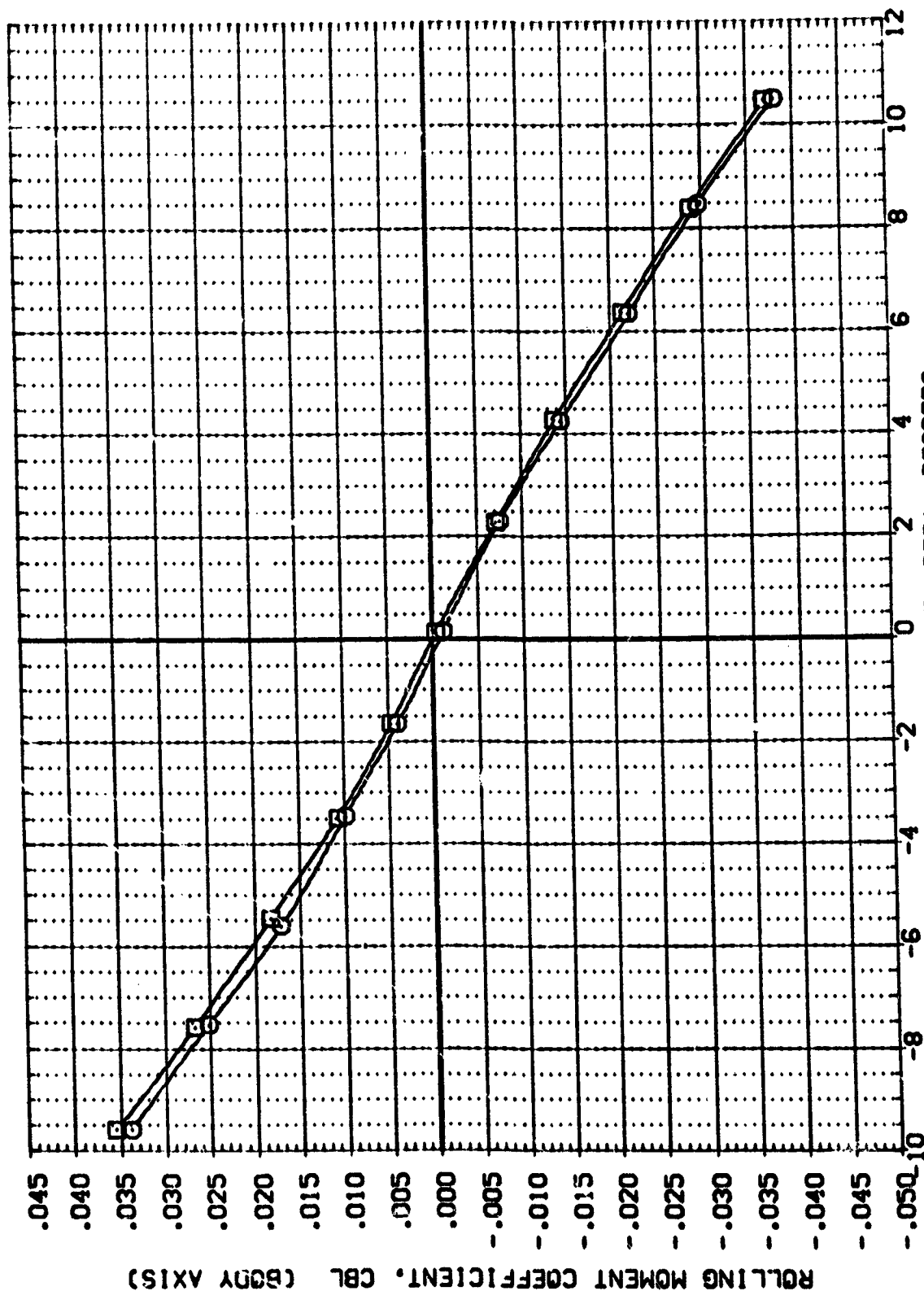


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
(R87006)	□	AVES 3.5-129	IA10 09 T10 AT2 PLUVE OFF	SREF	2690.0000 SD.FT.
(R87007)	□	AVES 3.5-169	IA10 09 T10 AT2 PLUVE OFF	LREF	1290.0000 IN.
				BREF	936.6800 IN.
				XMRP	1076.4800 IN.
				YMRP	.0000 IN.
				ZMRP	400.0000 IN.
				SCALE	.0100

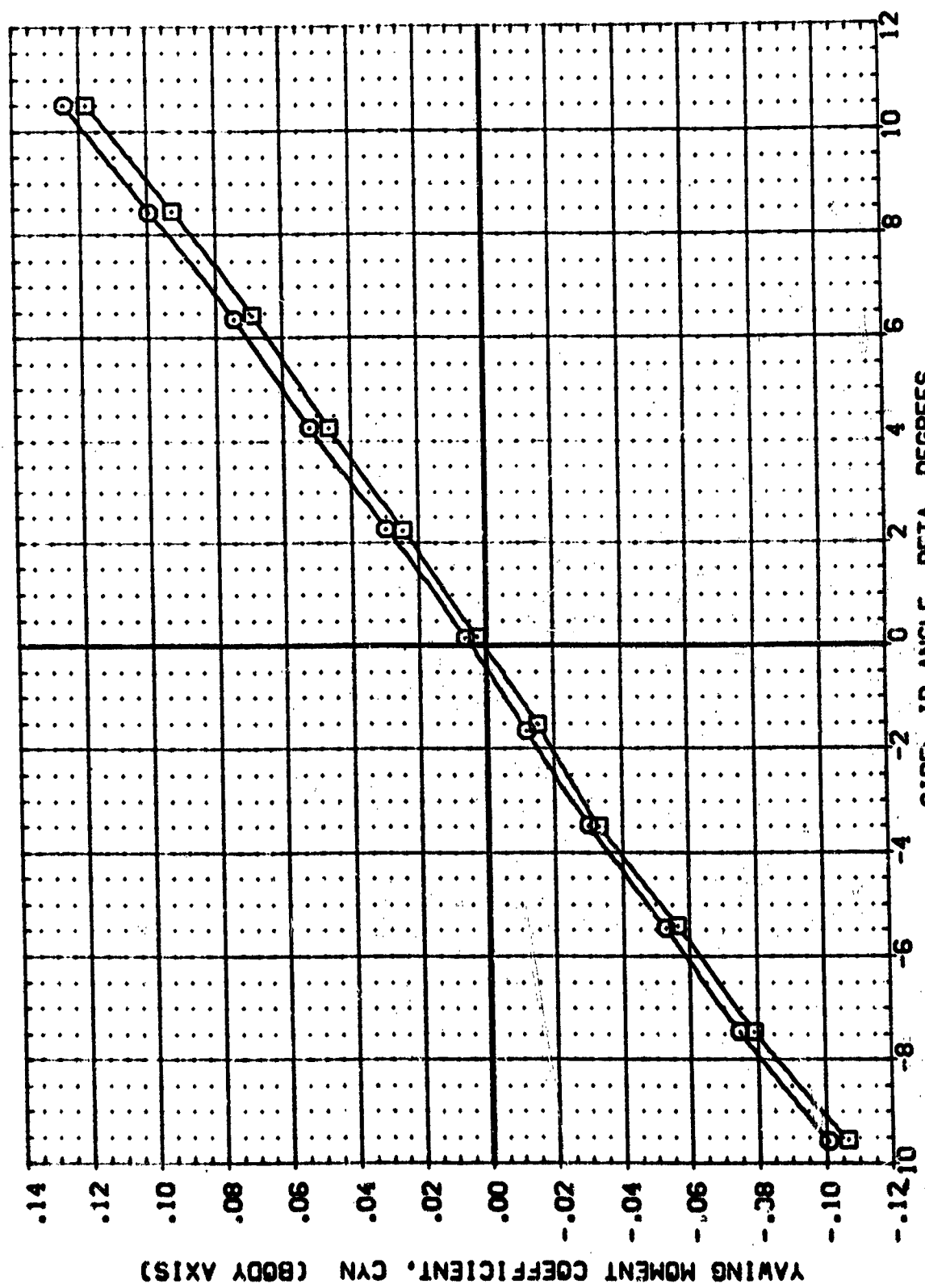


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL: (R87006) (R87007)

CONFIGURATION DESCRIPTION: AYES 3.5-169 IAI0 09 T10 AT2 PLUVE OFF

REFERENCE INFORMATION: SIZE 2890.000 IN. LIFT 1200.000 IN. DRAG 930.000 IN. YMRP 1076.480 IN. ZMRP 400.000 IN. SCALE 400.0000 0.0100

ALPHA: .000 .000

ELEVON: .000 .000

RUDER: .000 10.000

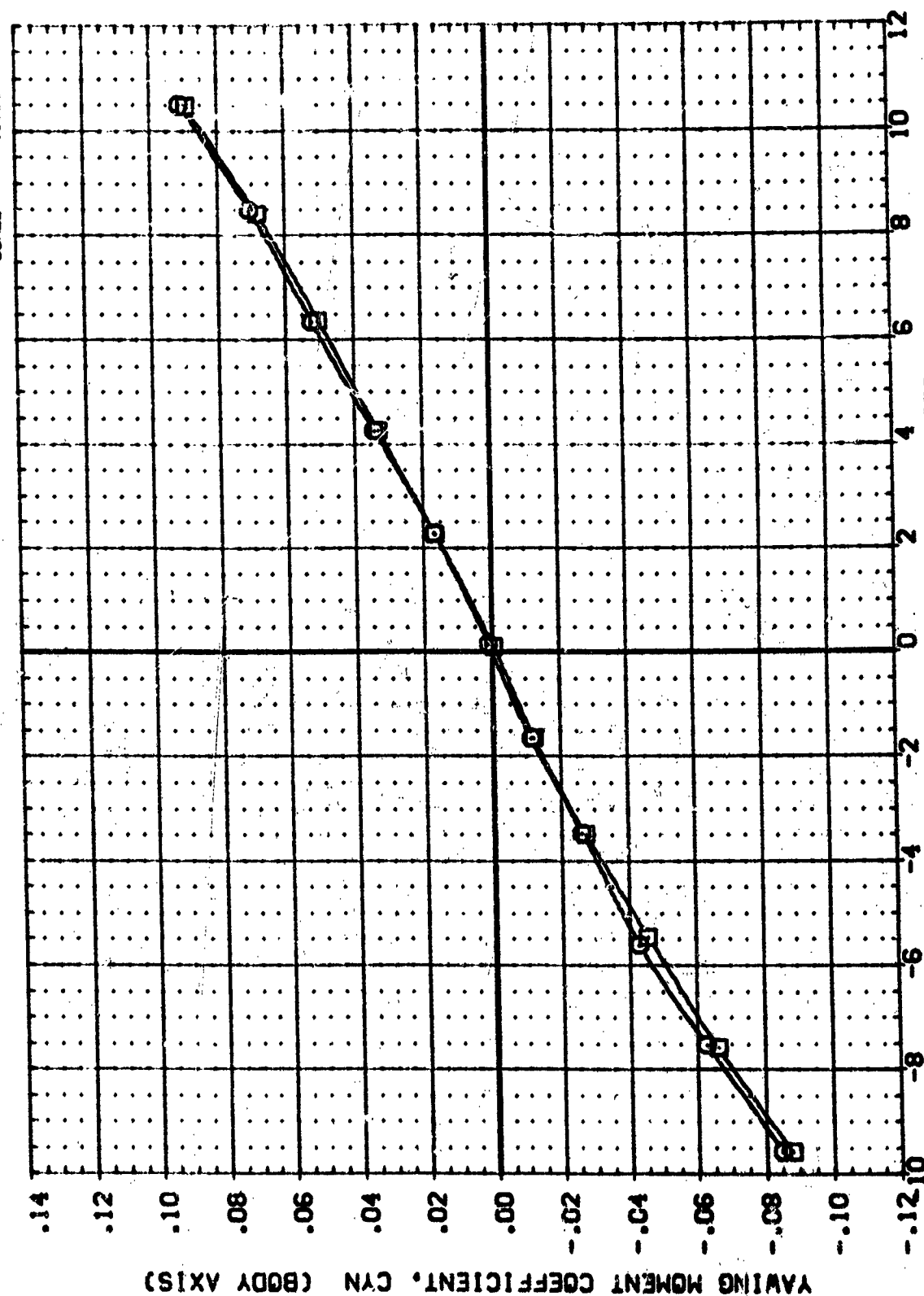


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		ALPHA		AILERON		ELEVON		RUDDER		REFERENCE INFORMATION			
(187005)	□	AVES 3-5-16S	1A10 09 T10 AT2 PLVE OFF	.000	.000	.000	.000	.000	.000	.000	.000	SREF	2690.0000	SQ.FT.	
(187007)	□	AVES 3-5-16S	1A10 09 T10 AT2 PLVE OFF	.000	.000	.000	.000	.000	.000	.000	.000	LREF	1290.0000	IN.	
												BREF	936.6800	IN.	
												XMRP	1076.4800	IN.	
												YMRP	.0000	IN.	
												ZMRP	400.0000	IN.	
												SCALE	.0100		

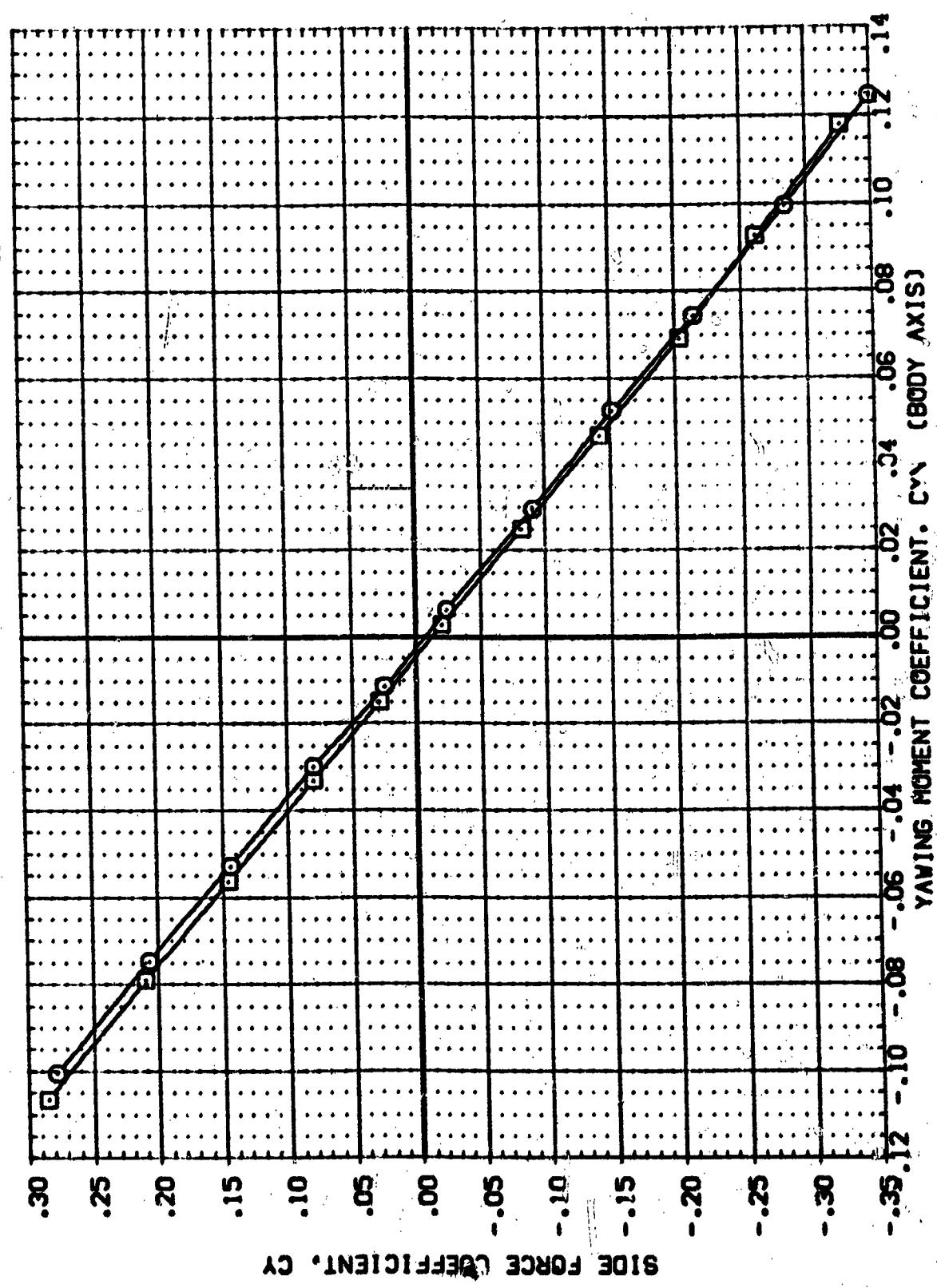


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(MACH = 5.26)

**CONFIGURATION DESCRIPTION**

ALPHA	AIRLON	ELEVON	RUDDER
.000	.000	.000	.000
.000	.000	.000	10.000

REFERENCE INFORMATION	
SREF	2690.0000 SQ.FT.
LREF	1290.0000 IN.
BREF	936.6800 IN.
XMRP	1076.4800 IN.
YMRP	.0000 IN.
ZMRP	400.0000 IN.
SCALE	.0100

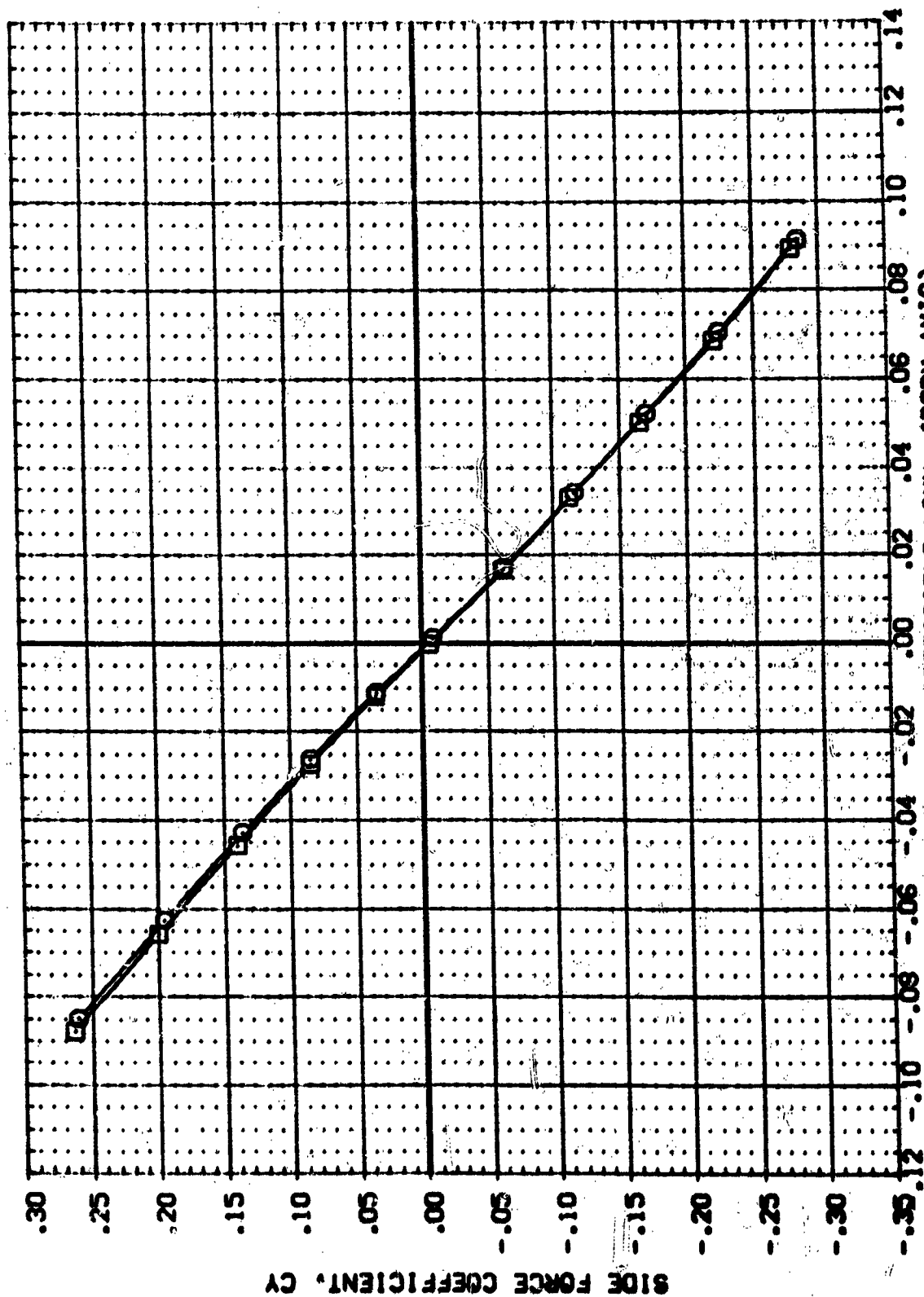


FIG. 8 EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS.

**{B}HACH = 7.32**

DATA SET SYMBOL: 10870111  
 CONFIGURATION DESCRIPTION: AVES 3-5-169 IA10 CS T10 AT2 PLUVE OFF  
 10870112 AVES 3-5-169 IA10 CS T10 AT2 PLUVE OFF  
 10870113 DATA NOT AVAILABLE  
 10870114 DATA NOT AVAILABLE

BETA: .000  
 .000  
 5.000  
 5.000

AIRLON: .000  
 .000  
 .000  
 .000

ELEVON: .000  
 .000  
 .000  
 .000

RUDDER: .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION:  
 SREF: 2650.0000 SO.FT.  
 LREF: 1250.0000 IN.  
 BREF: 935.5800 IN.  
 XREF: 1076.4800 IN.  
 YREF: .0000 IN.  
 ZREF: 400.0000 IN.  
 SCALE: .0100

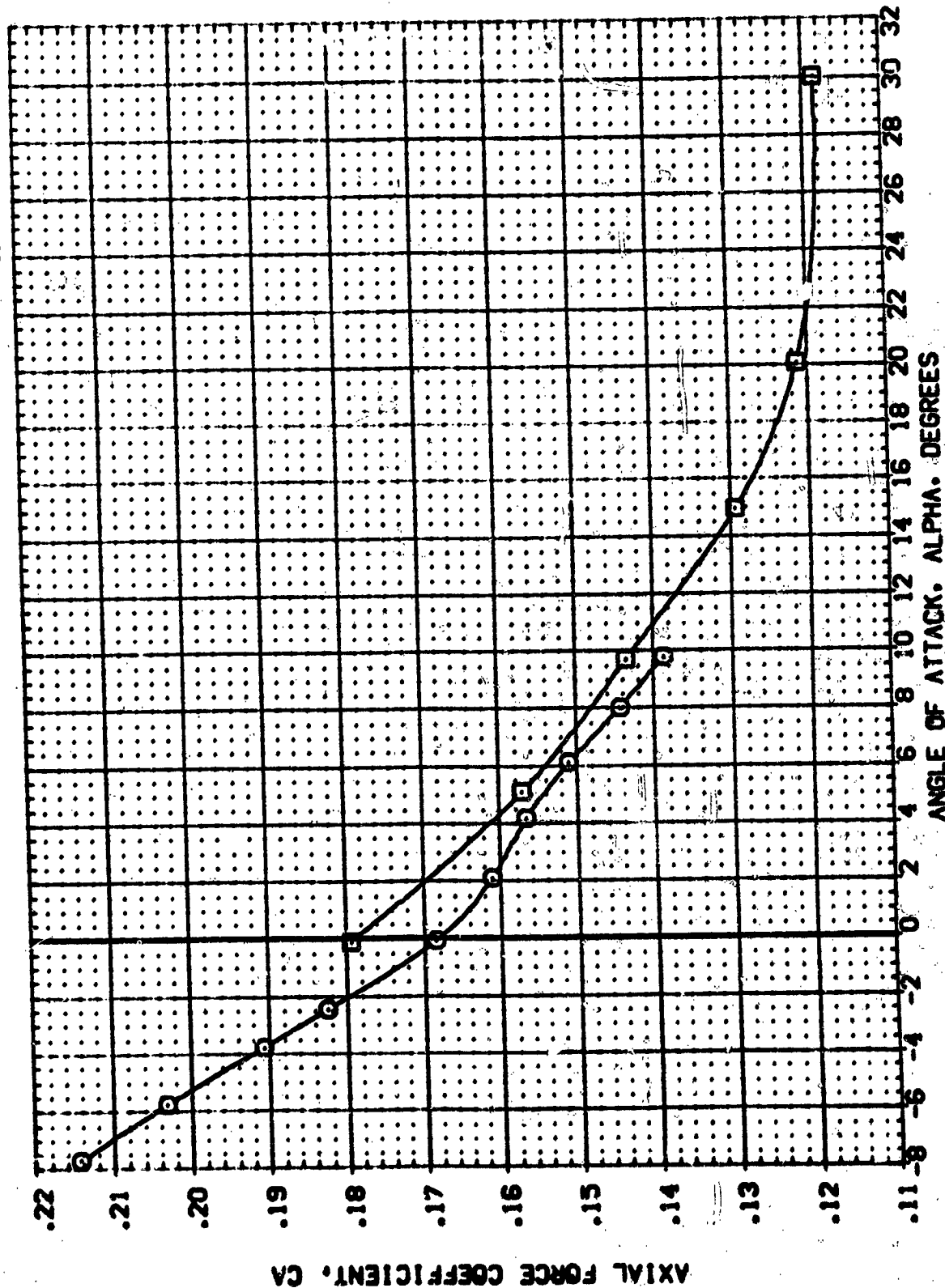


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



	DATA NOT AVAILABLE	09 110 AT2 PLUME OFF	09 110 AT2 PLUME OFF
(R87Q02)	AVES 3.5-169 IA10		
(R87Q04)	AVES 3.5-169 IA10		
(R87Q03)	AVES 3.5-169 IA10		

BETA	AIRLON	ELEVON	RUDDER	REFERENCE INFORMATION	SO. FT.
.000	.000	.000	.000	SREF	2690.0000
.000	.000	.000	.000	LREF	1290.0000
5.000	.000	.000	.000	EREF	936.6800
.000	.000	.000	.000	XTRP	1076.4800
.000	.000	.000	.000	YTRP	.0000
5.000	.000	.000	.000	ZTRP	400.0000
				SCALE	.0100

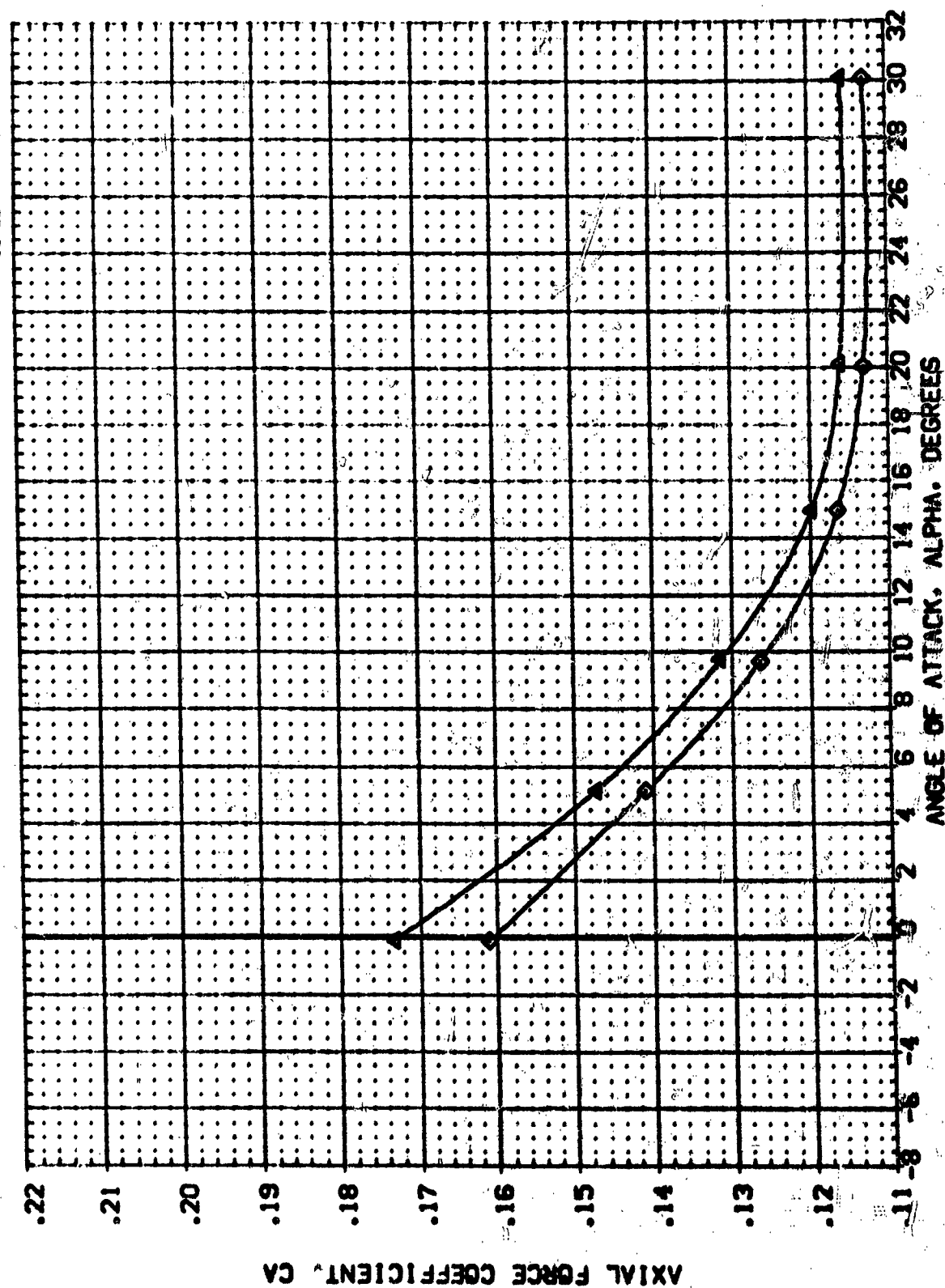


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AIRLON	ELEV	SLIDER	REFERENCE INFORMATION
(087011)	WES 3.5-169 IAD	.000	.000	.000	.000	SREF 2690.0000
(087012)	WES 3.5-169 IAD	.000	.000	.000	.000	LREF 1290.0000
(087013)	DATA NOT AVAILABLE	.000	.000	.000	.000	BREF 936.6800
		.000	.000	.000	.000	XREF 1076.4800
		.000	.000	.000	.000	YREF 400.0000
		.000	.000	.000	.000	ZREF 400.0000
		.000	.000	.000	.000	SCALE .0100

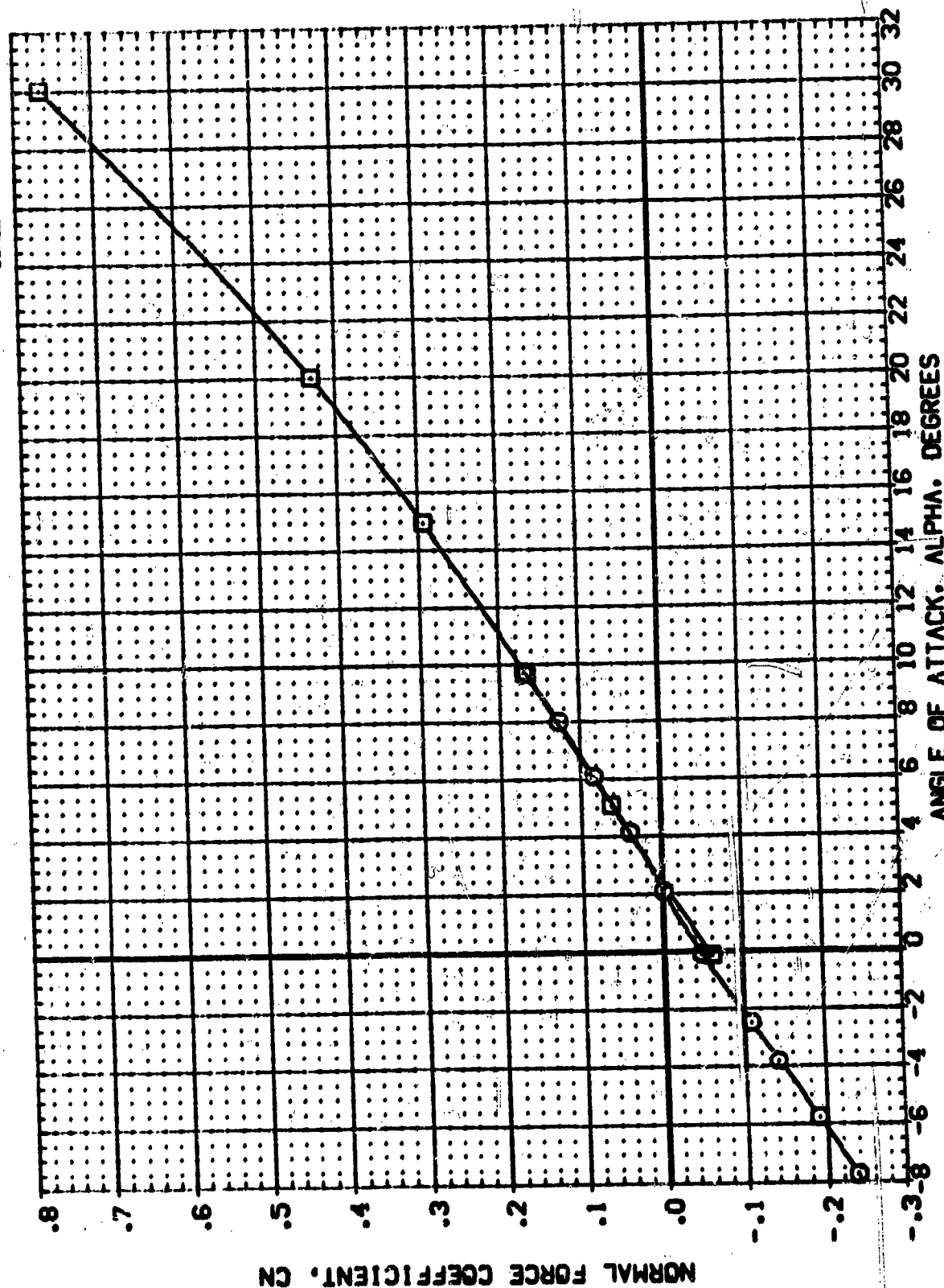


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = .26

DATA SET SYMB. CONFIGURATION DESCRIPTION

(037011) DATA NOT AVAILABLE

(037022) DATA NOT AVAILABLE

(037034) APES 3.5-169 1A10 09 T10 AT2 PLUVE OFF

(037033) APES 3.5-169 1A10 09 T10 AT2 PLUVE OFF

BETA .000

.000

5.000

5.000

AILRON .000

.000

.000

.000

ELEVON .000

.000

.000

.000

RUDER .000

.000

.000

.000

REFERENCE INFORMATION

SREF 2690.0000 SO.FT.

LREF 1290.0000 IN.

BREF 936.6800 IN.

XPBP 1076.4800 IN.

YBP 400.0000 IN.

ZBP 400.0000 IN.

SCALE .0100

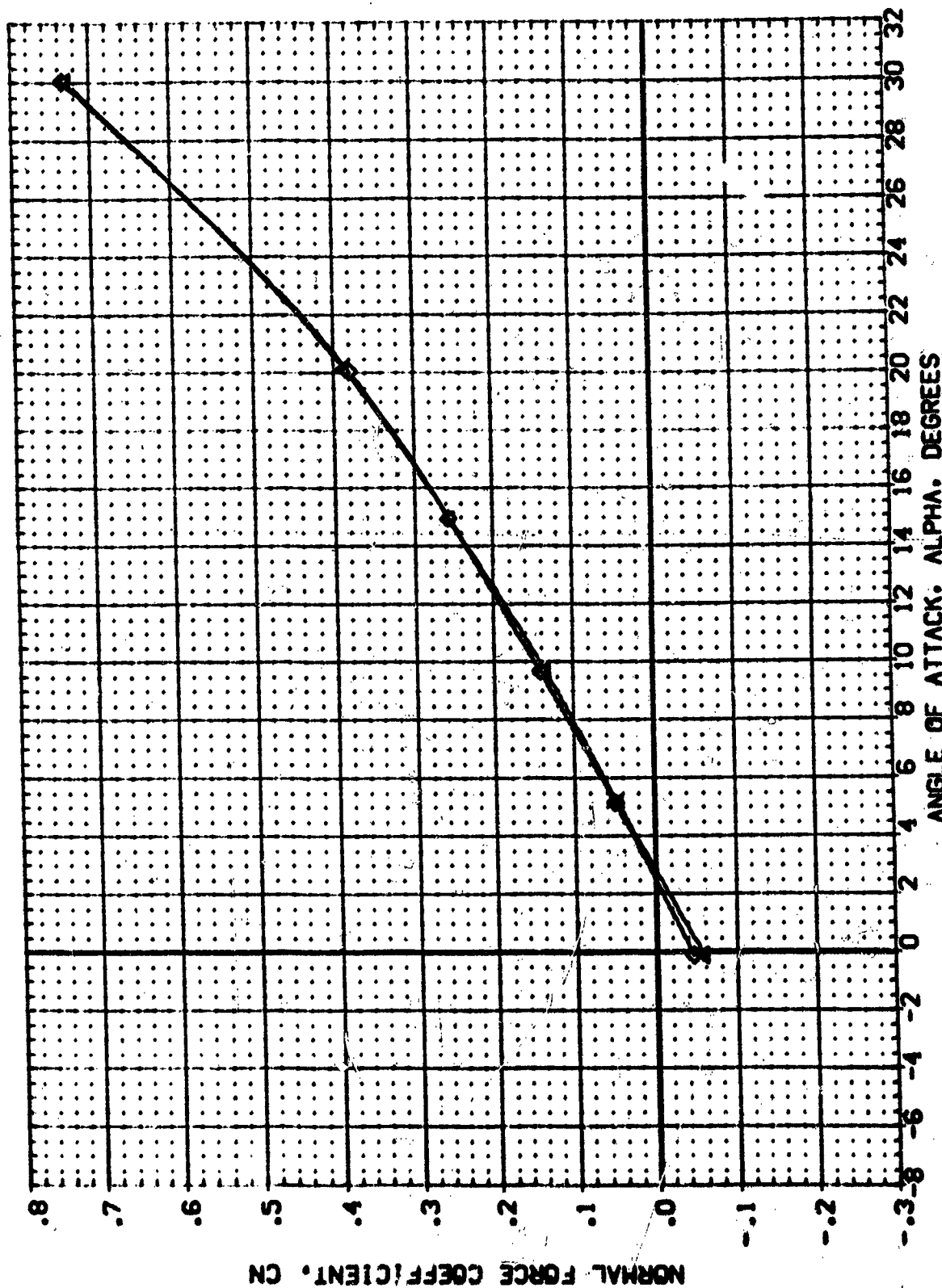


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AIRLON	ELEVON	FLUDER	REFERENCE INFORMATION
(087011)	AVES 3-5-169 1A10 09 T10 AT2 PLUVE OFF	.000	.000	.000	.000	SREF 2690.0000 SQ.FT.
(087012)	AVES 3-5-169 1A10 09 T10 AT2 PLUVE OFF	5.000	.000	.000	.000	LREF 1290.0000 IN.
(087003)	DATA NOT AVAILABLE	5.000	.000	.000	.000	BREF 936.6800 IN.
			.000	.000	.000	XPRP 1076.4800 IN.
			.000	.000	.000	YPRP .0000 IN.
			.000	.000	.000	ZPRP 400.0000 IN.
						SCALE .0100

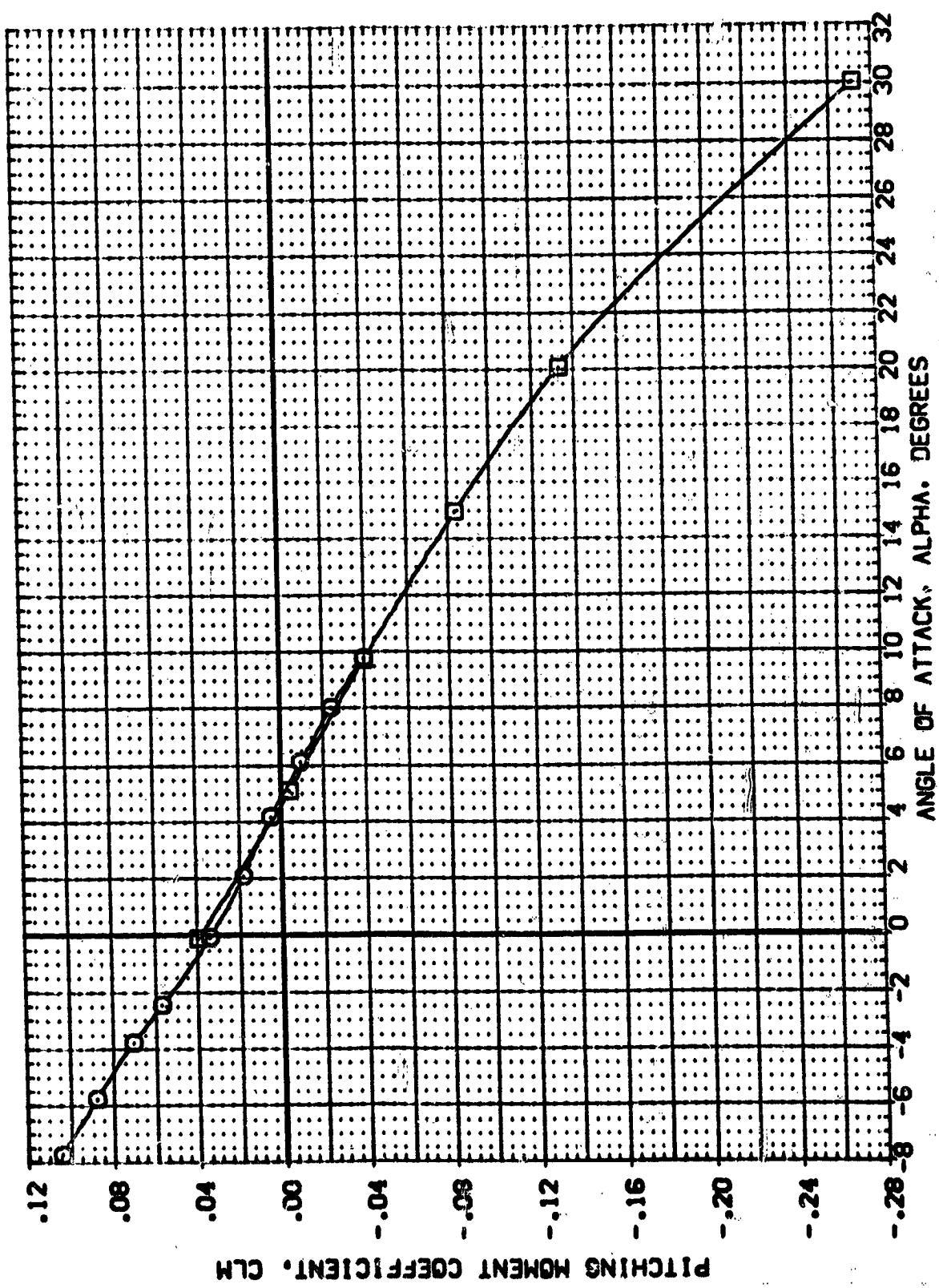


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

# DATA SET SYMBOL CONFIGURATION DESCRIPTION

(S87011) DATA NOT AVAILABLE  
 (S87002) DATA NOT AVAILABLE  
 (S87003) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF  
 (S87003) AVES 3.5-169 IA10 OS T10 AT2 PLUVE OFF

BETA .000  
 .000  
 5.000  
 .000  
 5.000

AIRLON .000  
 .000  
 .000  
 .000  
 .000

ELEVON .000  
 .000  
 .000  
 .000  
 .000

RUDER .000  
 .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

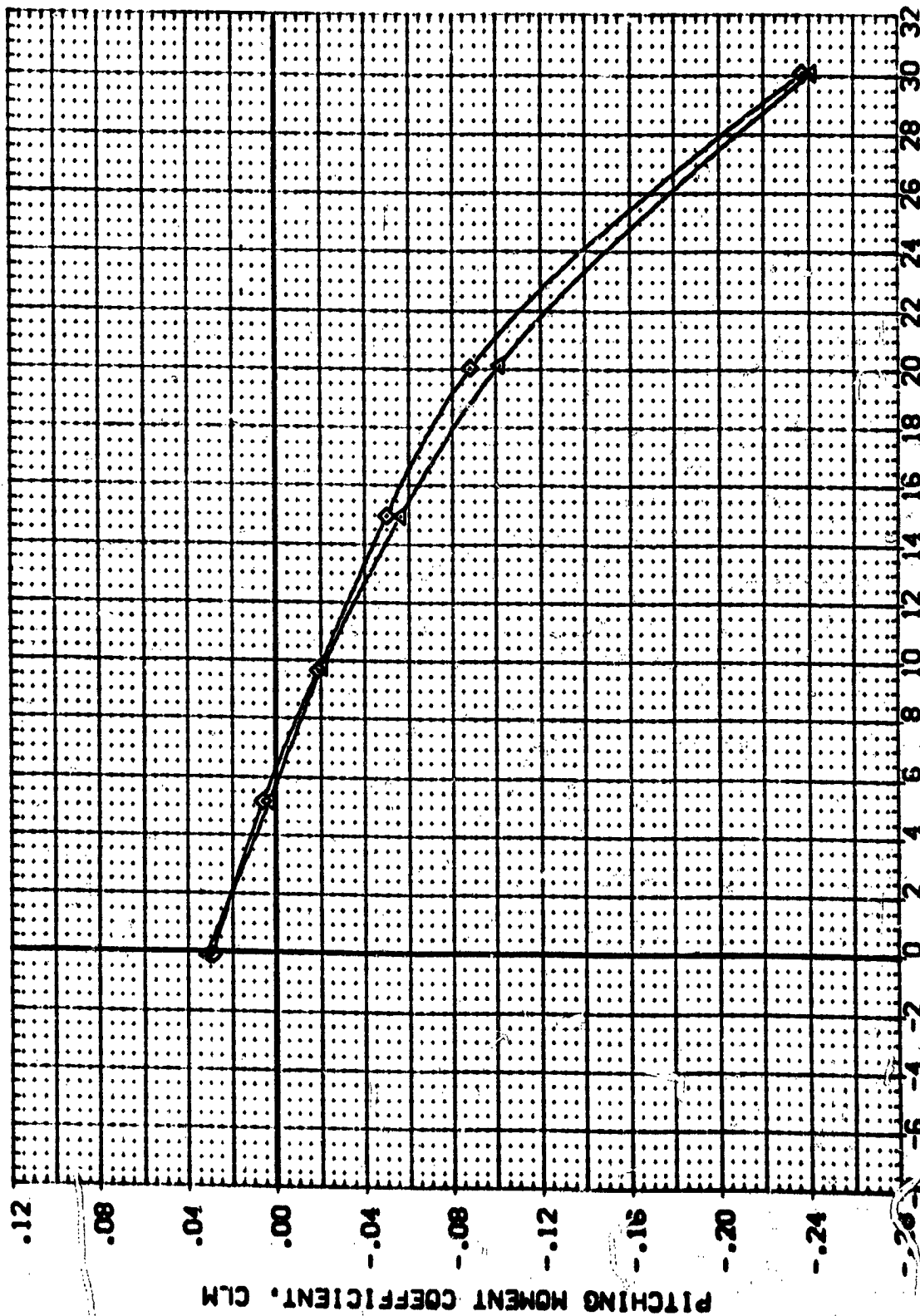


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32



DATA SET SYMOL		CONFIGURATION DESCRIPTION	BETA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION	
(DB7011)	□	DATA NOT AVAILABLE	.000	.000	.000	.000	SREF	2690.0000 SQ.FT.
(SB7002)	□	DATA NOT AVAILABLE	5.000	.000	.000	.000	LREF	1290.0000 IN.
(RB7004)	×	AVES 3.5-169 1A10	.000	.000	.000	.000	BREF	936.5900 IN.
(RB7003)	×	AVES 3.5-169 1A10	5.000	.000	.000	.000	YPRP	1076.4800 IN.
		OS T10 AT2 PLUE OFF					ZPRP	400.0000 IN.
		OS T10 AT2 PLUE OFF					SCALE	.0100

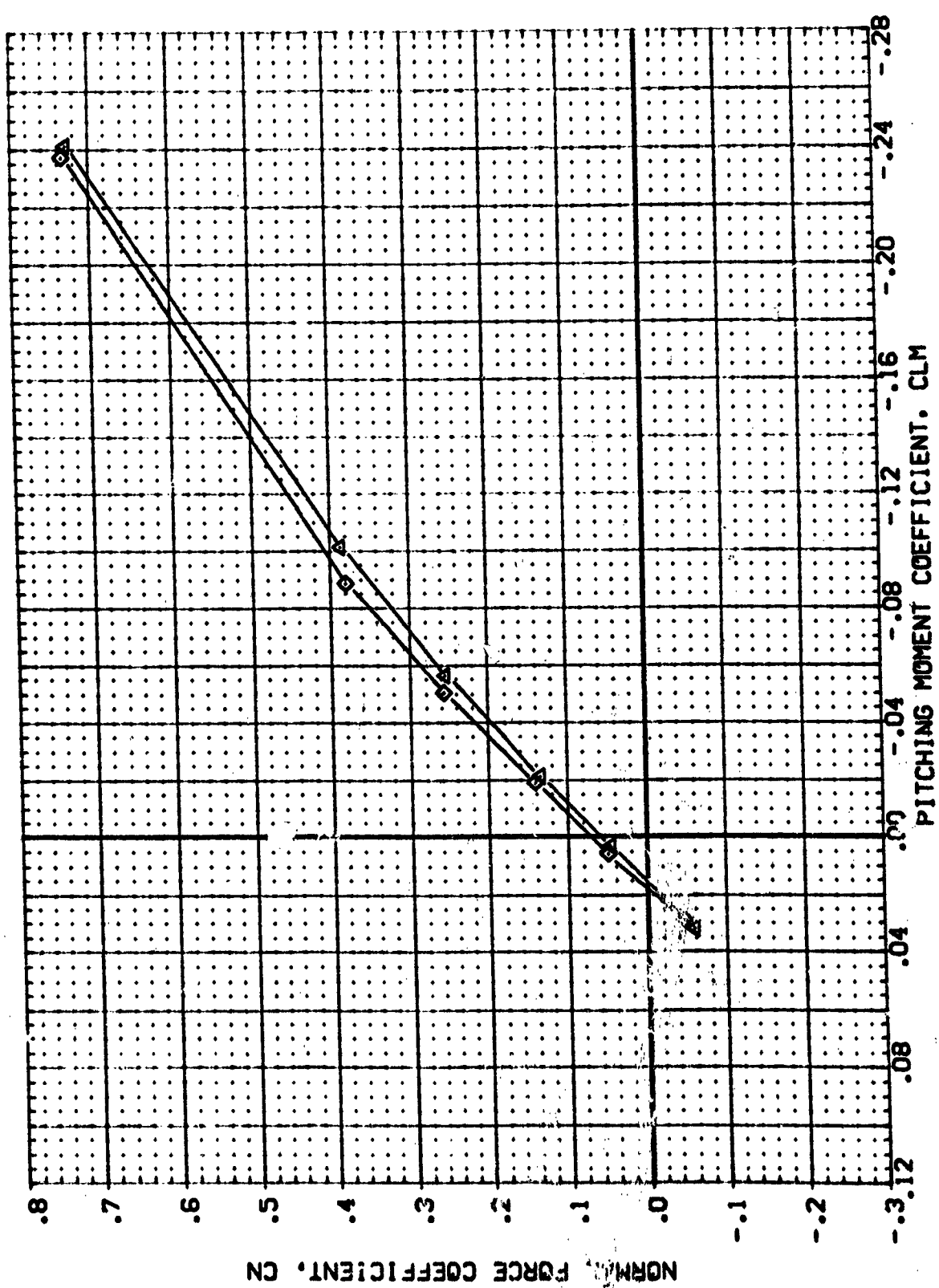


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AIRBORN	ELEVON	RUDDER	REFERENCE INFORMATION
(097011)	AVES 3.5-189 1A10 OS T10 AT2 PLUVE OFF	.000	.000	.000	.000	SREF 2650.0000 SQ.FT.
(097002)	AVES 3.5-189 1A10 OS T10 AT2 PLUVE OFF	5.000	.000	.000	.000	LREF 1250.0000 IN.
(097004)	DATA NOT AVAILABLE	.000	.000	.000	.000	BREF 936.6800 IN.
(097003)	DATA NOT AVAILABLE	5.000	.000	.000	.000	YMRP 1076.4800 IN.
						ZMRP .0000 IN.
						SCALE 400.0000 IN.

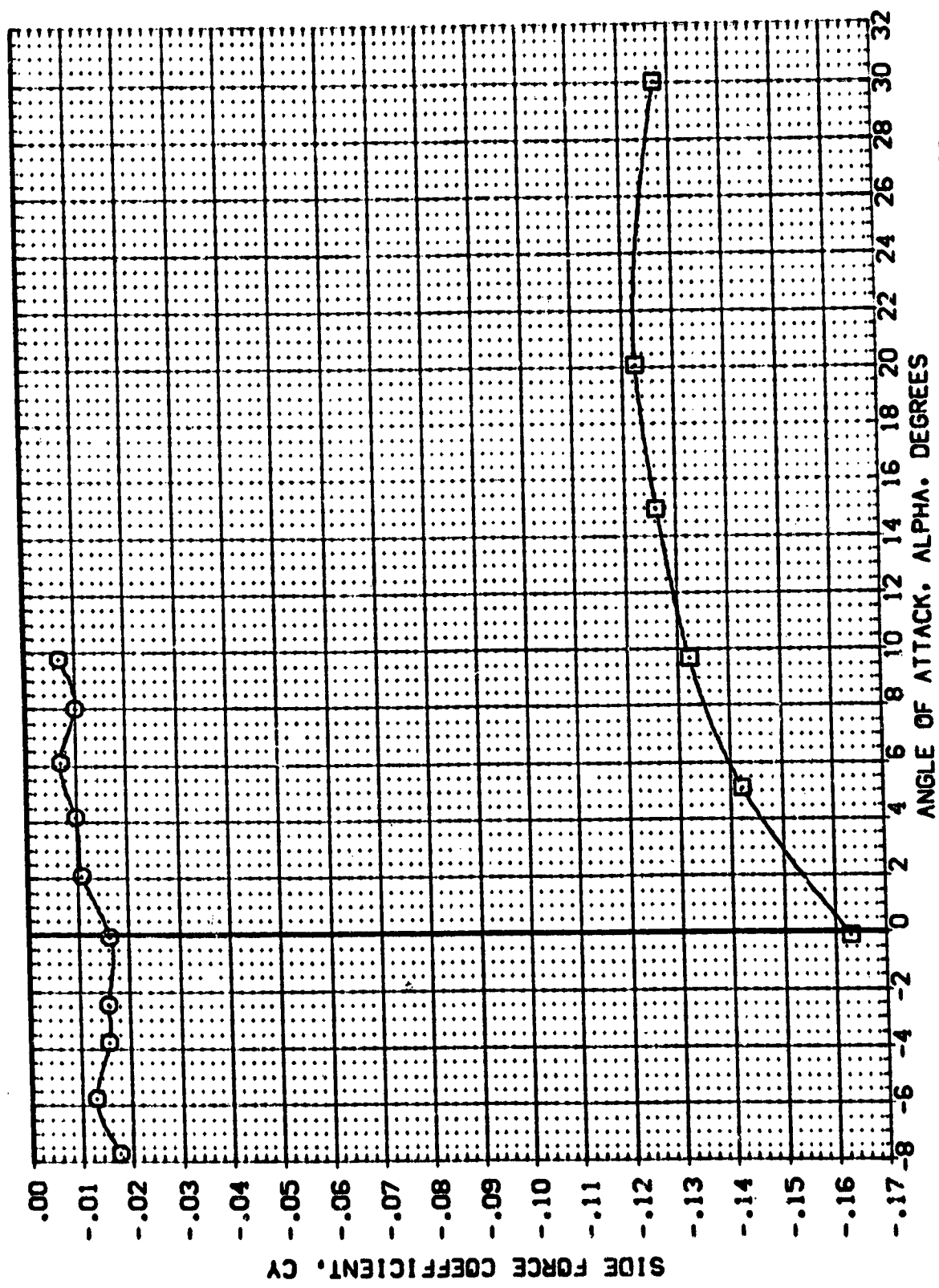


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.



DATA NOT AVAILABLE  
 ANES 3.5-169 1A10 09 T10 AT2 PLUME OFF  
 ANES 3.5-169 1A10 09 T10 AT2 PLUME OFF



FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

**[B]MACH = 7.32**

DATA SET SYMBOL  
 (DB701)  
 (SB7002)  
 (RB7004)  
 (RB7003)

CONFIGURATION DESCRIPTION  
 AXES 3-5-169 IAI0 0- T10 AT2 PLUVE OFF  
 AXES 3-5-169 IAI0 0- T10 AT2 PLUVE OFF  
 DATA NOT AVAILABLE  
 DATA NOT AVAILABLE

BETA  
 .000  
 5.000  
 5.000

AILRON  
 .000  
 .000  
 .000

ELEVON  
 .000  
 .000  
 .000

RUDDER  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 7590.0000  
 LREF 7290.0000  
 BREF 536.6800  
 XMRP 076.4800  
 YMRP 0.00  
 ZMRP 0.00  
 SCALE .0100

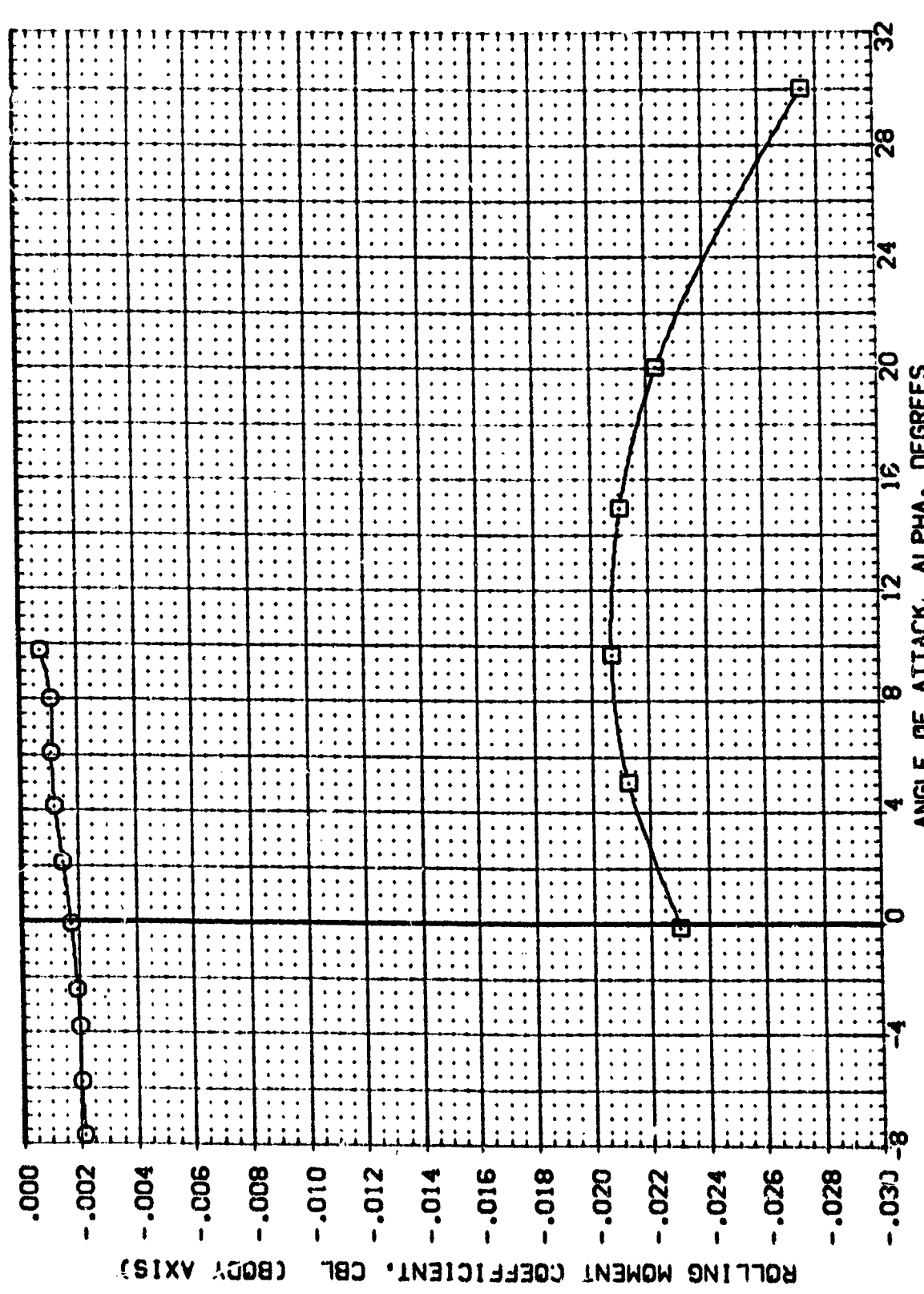


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

(DB7011)  
 (SB7002)  
 (RB7004)  
 (RS7003)

DATA NOT AVAILABLE  
DATA NOT AVAILABLE  
AMES 3-5-69 1A10  
AMES 3-5-69 1A10

09 T10 AT2 PLUME OFF  
09 T10 AT2 PLUME OFF

BETA  
5.000  
5.000  
5.000

AIL RON	ELEVON	RUDDER
.000	.000	.000
.000	.000	.000
.000	.000	.000
.000	.000	.000

REFERENCE INFORMATION	
SREF	2690.0000 SQ.FT.
LREF	1290.0000 IN.
BREF	936.6800 IN.
XMRP	1076.4800 IN.
YMRP	.0000 IN.
ZMRP	400.0000 IN.
SCALE	.0100

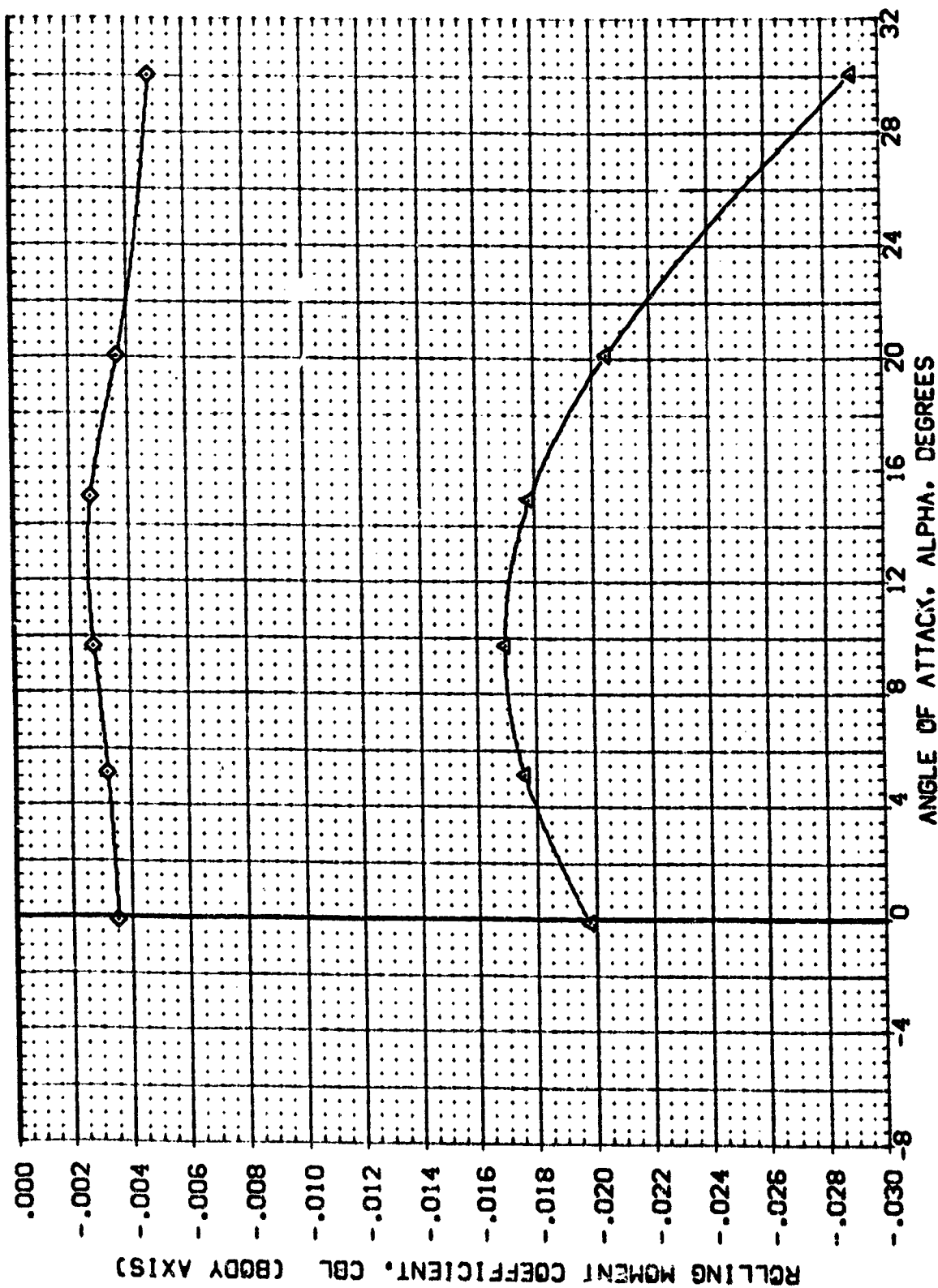


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

DATA SET SYMBOL: 12870111  
 CONFIGURATION DESCRIPTION: AVES 3.5-169 IA10 09 T10 AT2 PLUVE OFF  
 (587002) AVES 3.5-169 IA10 05 T10 AT2 PLUVE OFF  
 (587003) DATA NOT AVAILABLE  
 (587004) DATA NOT AVAILABLE  
 (587005) DATA NOT AVAILABLE

BETA: .000  
 .000  
 5.000  
 5.000

AILERON: .000  
 .000  
 .000  
 .000

ELEVON: .000  
 .000  
 .000  
 .000

RUDDER: .000  
 .000  
 .000  
 .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XPRP 1076.4800 IN.  
 YPRP 400.0000 IN.  
 ZPRP 400.0000 IN.  
 SCALE .0100

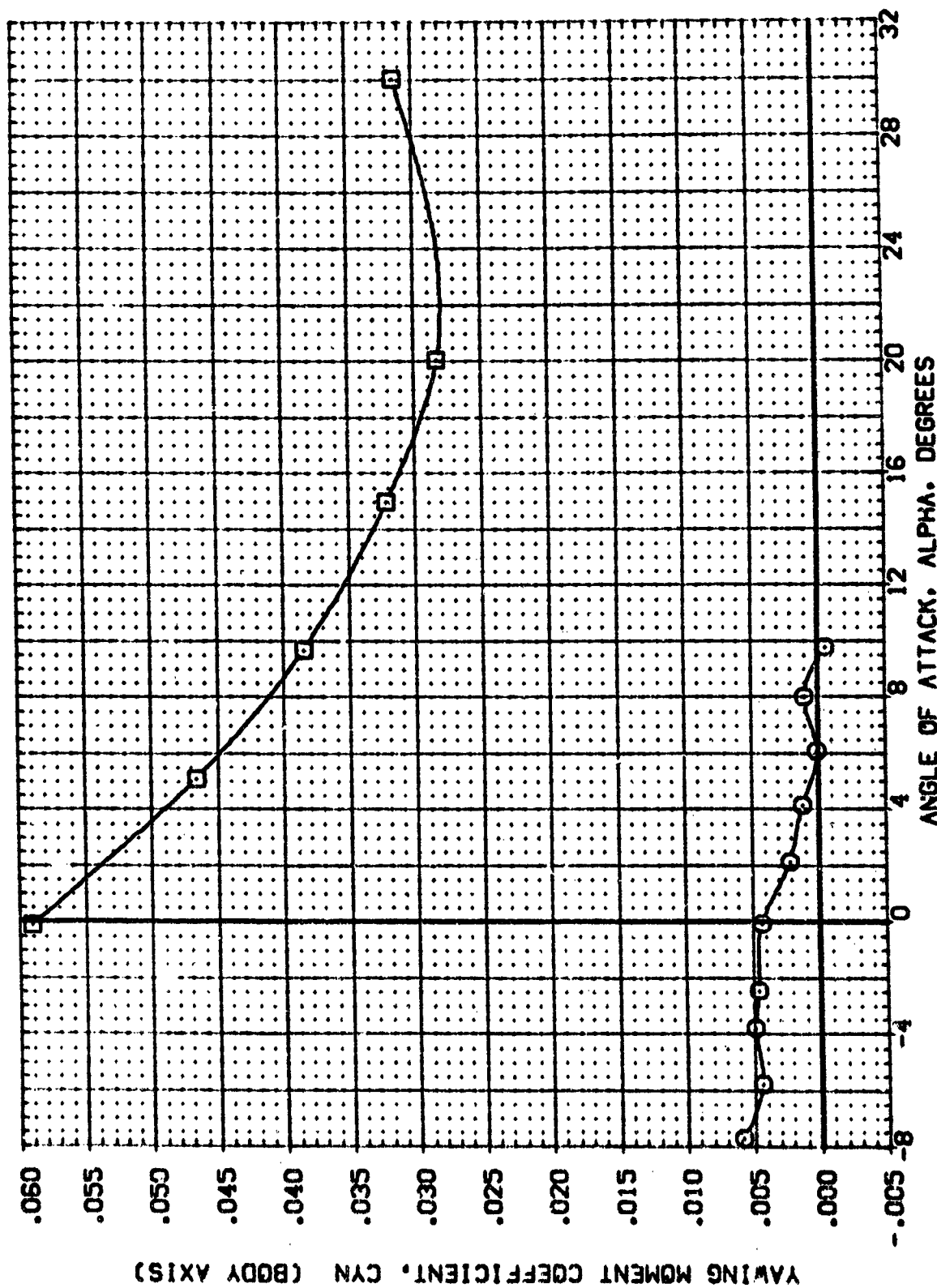


FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A) MACH = 5.26

# DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DB7011) DATA NOT AVAILABLE  
 (SB7032) DATA NOT AVAILABLE  
 (RB7014) AYES 3.5-169 IAI0 09 T10 AT2 PLUVE OFF  
 (RB7033) AYES 3.5-169 IAI0 09 T10 AT2 PLUVE OFF

BETA  
 .000  
 5.000  
 5.000

AIRLON  
 .000  
 .000  
 .000

ELEVON  
 .000  
 .000  
 .000

FLAPER  
 .000  
 .000  
 .000

REFERENCE INFORMATION  
 SREF 2650.0000 SO.FT.  
 LREF 1250.0000 IN.  
 BREF 936.6800 IN.  
 XPROP 1076.4800 IN.  
 YPROP .0000 IN.  
 ZPROP 400.0000 IN.  
 SCALE .0100

YAWING MOMENT COEFFICIENT, CYN (BODY AXIS)

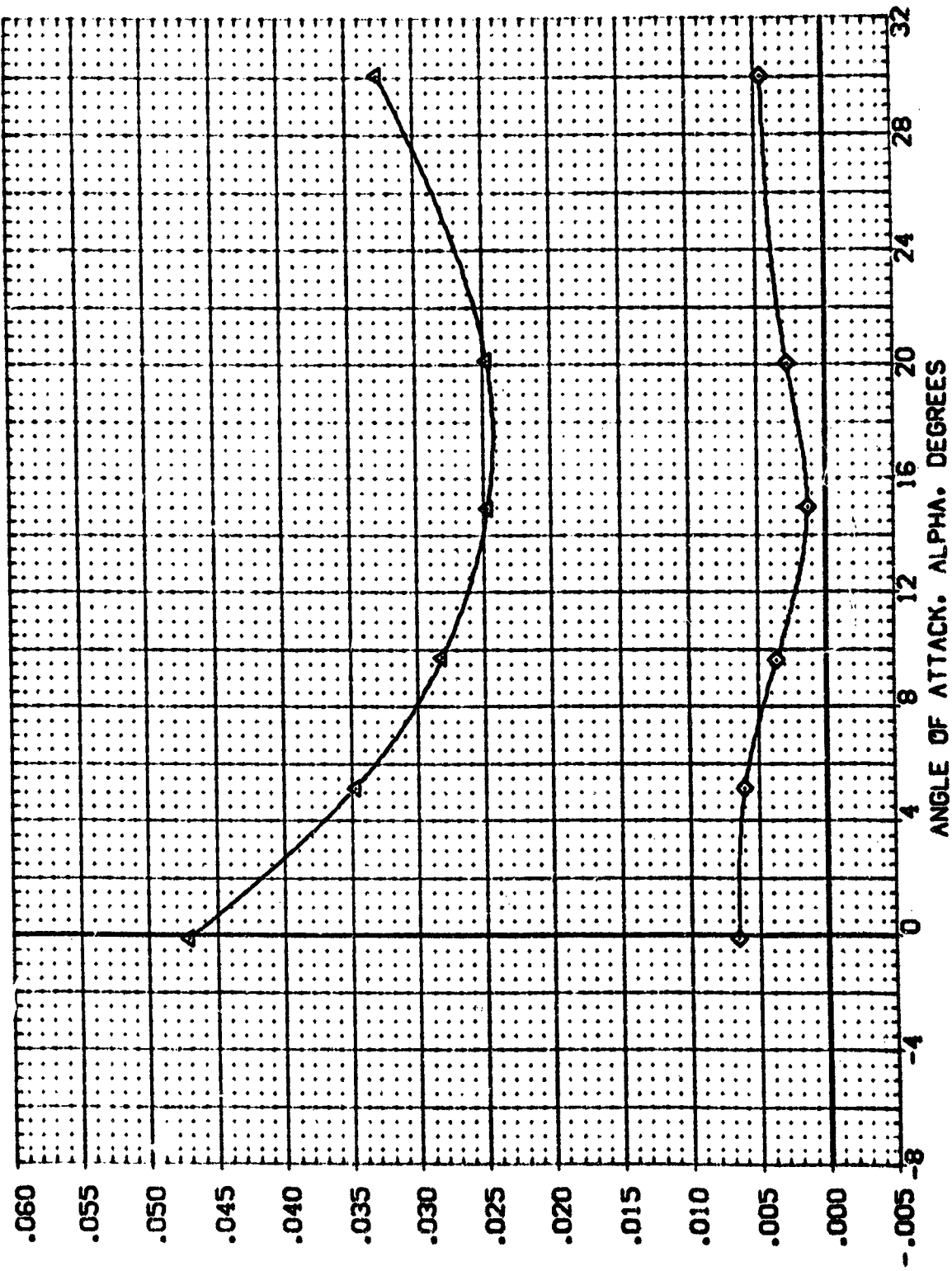



FIG. 9 EFFECT OF ANGLE OF ATTACK ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL: (487008) (1887008)  CONFIGURATION DESCRIPTION: AYES 3.5-169 1A10 OS T10 AT2 PLUVE ON AYES 3.5-169 1A10 OS T10 AT2 PLUVE ON

ALPHA: .000 .000  
AILRON: .000 .000  
ELEVON: .000 .000  
RUDDER: .000 10.000

REFERENCE INFORMATION:  
SREF: 2690.0000 SQ.FT.  
LREF: 1290.0000 IN.  
BREF: 936.6800 IN.  
XPRP: 1076.4800 IN.  
YPRP: .0000 IN.  
ZPRP: 400.0000 IN.  
SCALE: .0100

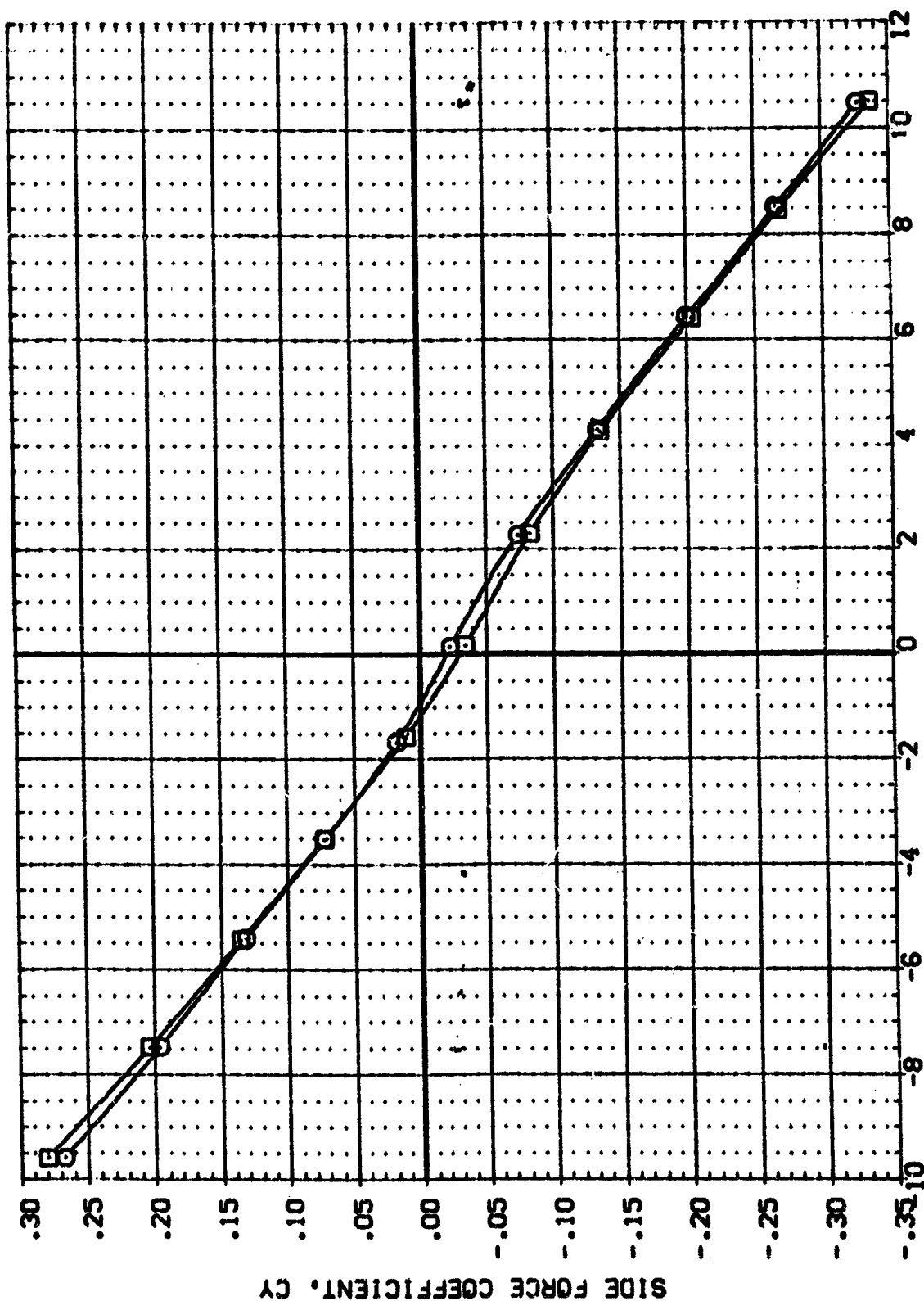


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL: (R87029) (R87028)

CONFIGURATION DESCRIPTION: AVES 3.5-169 1A10 09 T10 AT2 PLANE ON AVES 3.5-169 1A10 09 T10 AT2 PLANE ON

ALPHA: .000 .000 .000 .000

ELEVON: .000 .000 .000 .000

RUDDER: .000 .000 .000 .000

REFERENCE INFORMATION:

	SO.FT.
SREF	2690.0000
LREF	1290.0000
BREF	936.6900
XMRP	1076.4800
YMRP	.0000
ZMRP	400.0000
SCALE	.0100

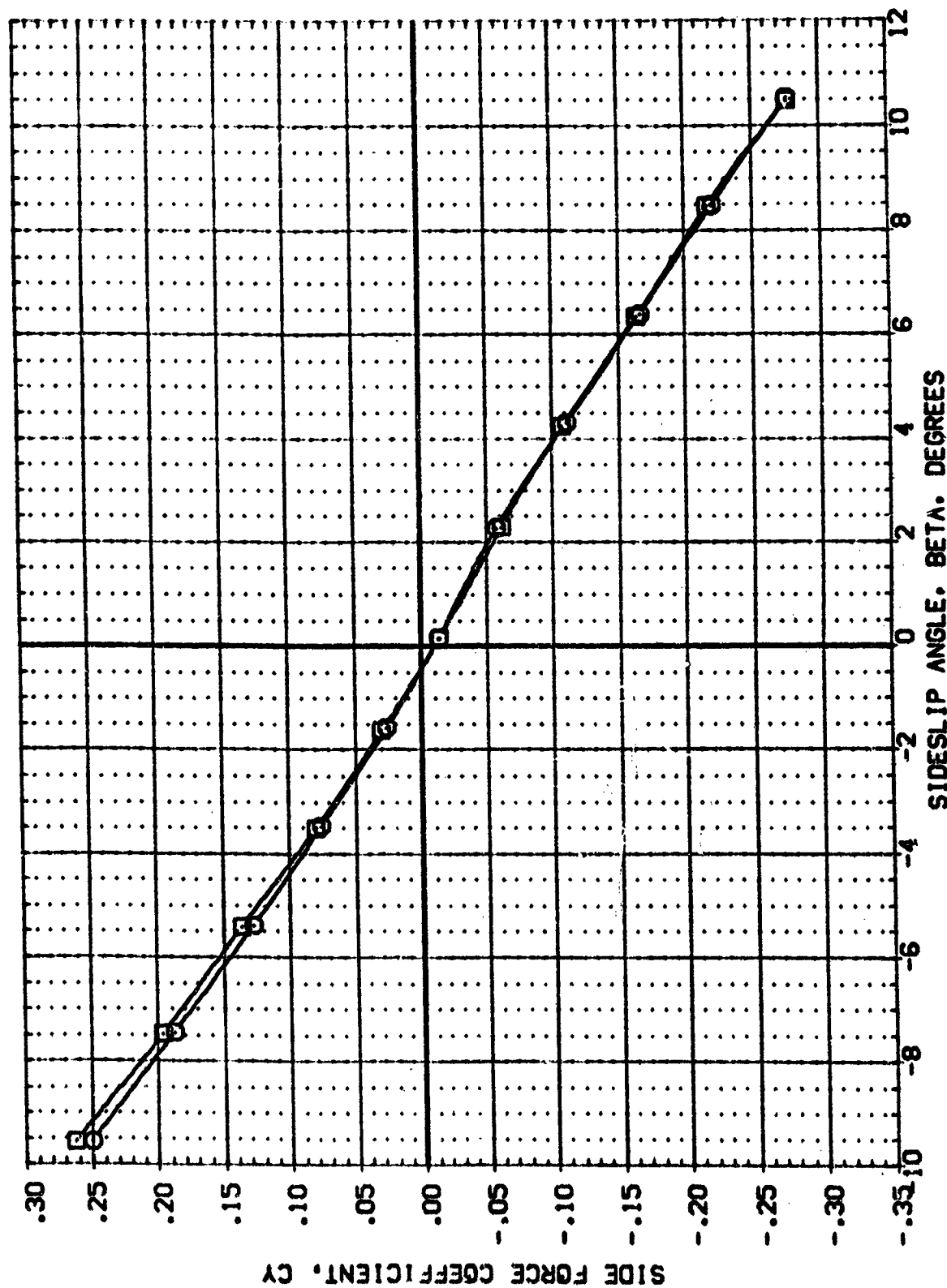


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL: (REF7008) CONFIGURATION DESCRIPTION: AYES 3.5-169 1A10 09 T10 AT2 PLANE ON REFERENCE INFORMATION: 2690.0000 50.FT.

ALPHA: .000 AILRON: .000 ELEVON: .000 RUDDER: .000

ROLLING MOMENT COEFFICIENT, CBL (BODY AXIS): .000 LREF: 1250.0000 BREF: 936.6800 XMRP: 1076.4800 YMRP: 0.0000 ZMRP: 400.0000 SCALE: 400.0000

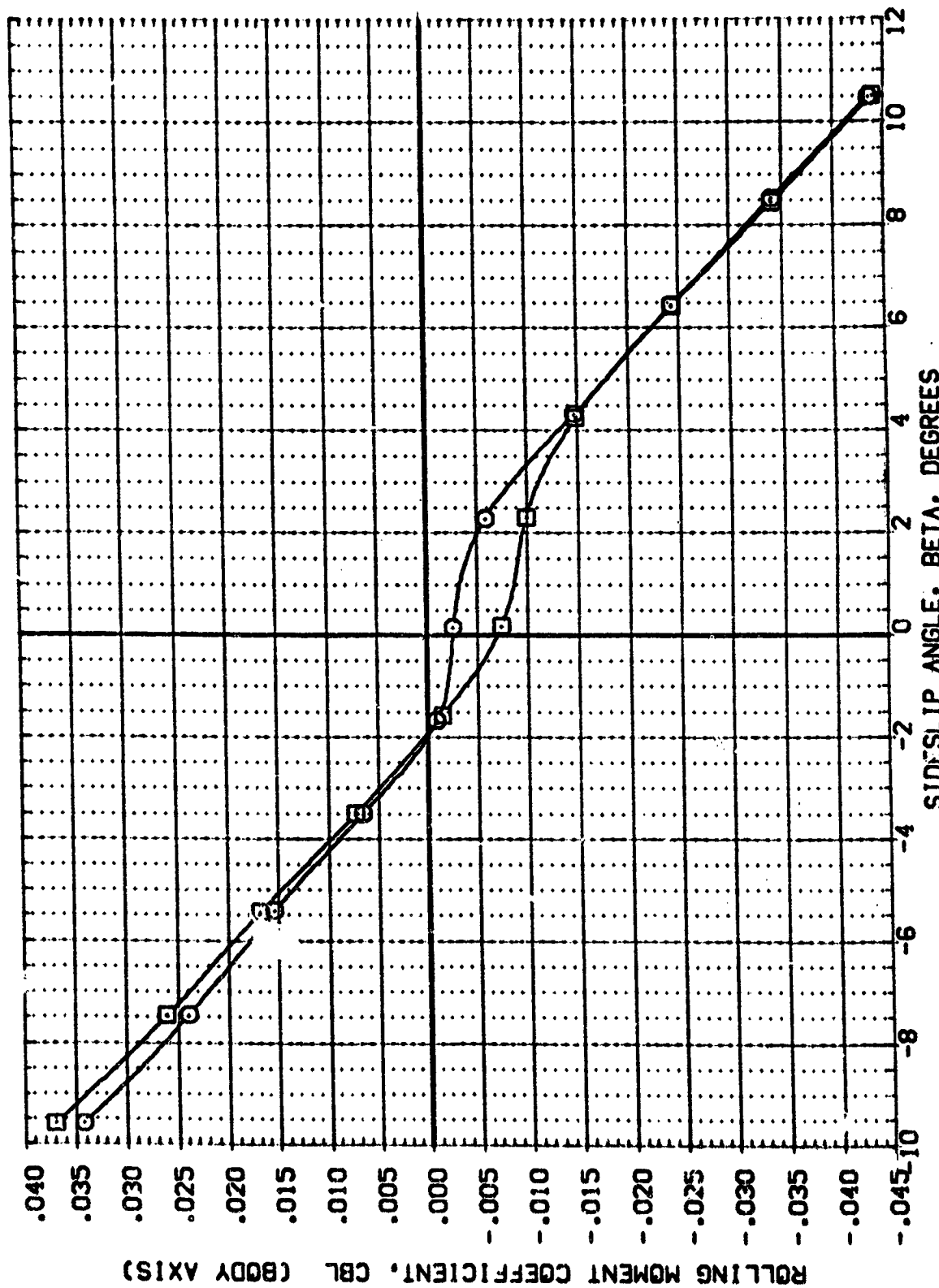


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26



REFERENCE INFORMATION		SG.FT.
SPEC	7600	0000
DATE	1200	0000
TIME	0306	5800
TYPE	1076	4800
WIND	400	0000
WAVE	400	0000
SCALE		0:00

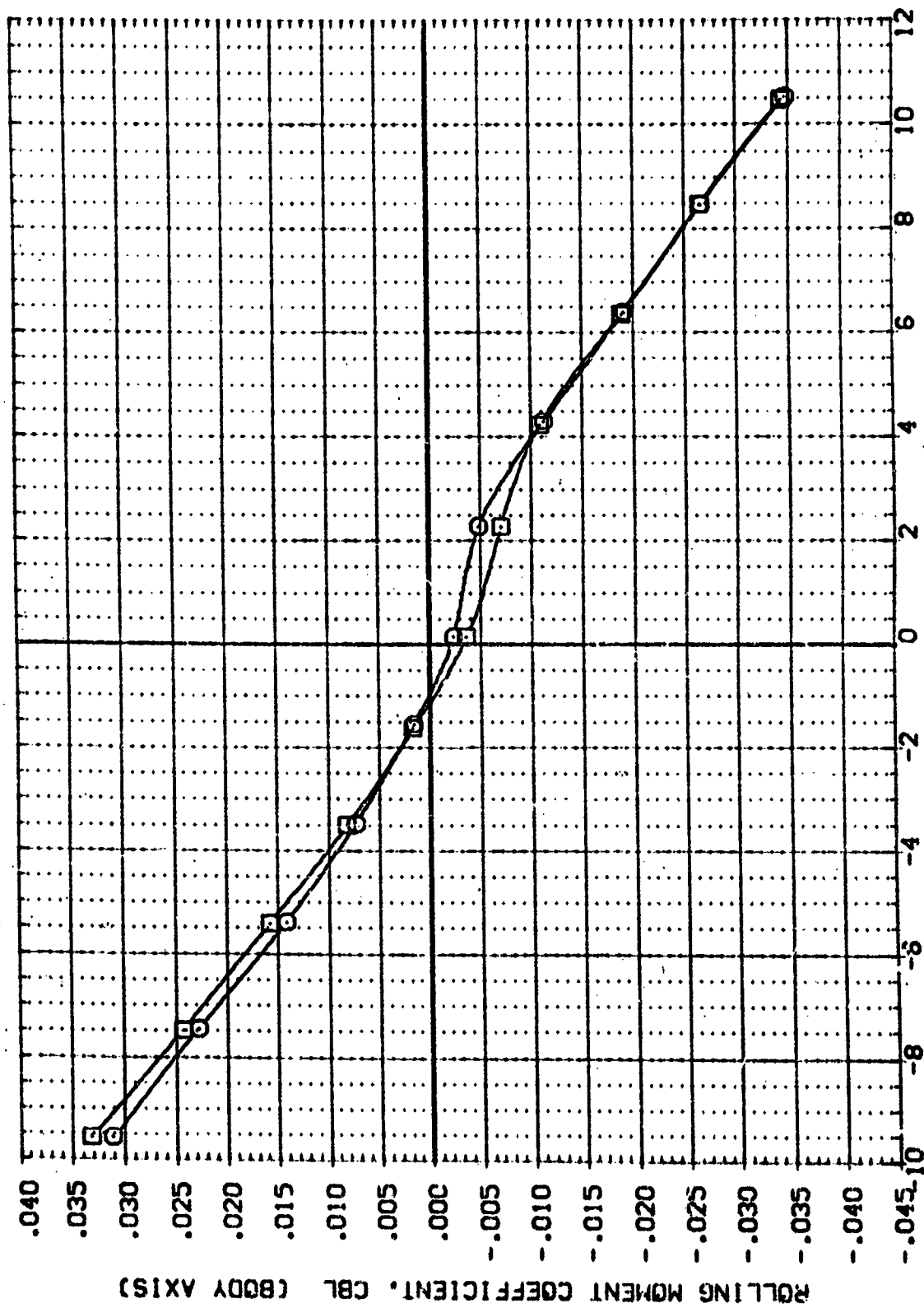


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

**[B]MACH = 7.32**

DATA SET SYMBL. CONFIGURATION DESCRIPTION

(R87009) [ ] AYES 3.5-169 1A10 09 T10 AT2 PLUVE ON

(R87008) [ ] AYES 3.5-169 1A10 09 T10 AT2 PLUVE ON

ALPHA AILRON ELEVON RUDDER REFERENCE INFORMATION

.000 .000 .000 SREF 2690.0000 SQ.FT.

.000 .000 .000 LREF 1790.0000 IN.

.000 .000 .000 BRFP 536.6800 IN.

.000 .000 .000 YMRP 1076.4800 IN.

.000 .000 .000 ZMRP 400.0000 IN.

SCALE 400.0000

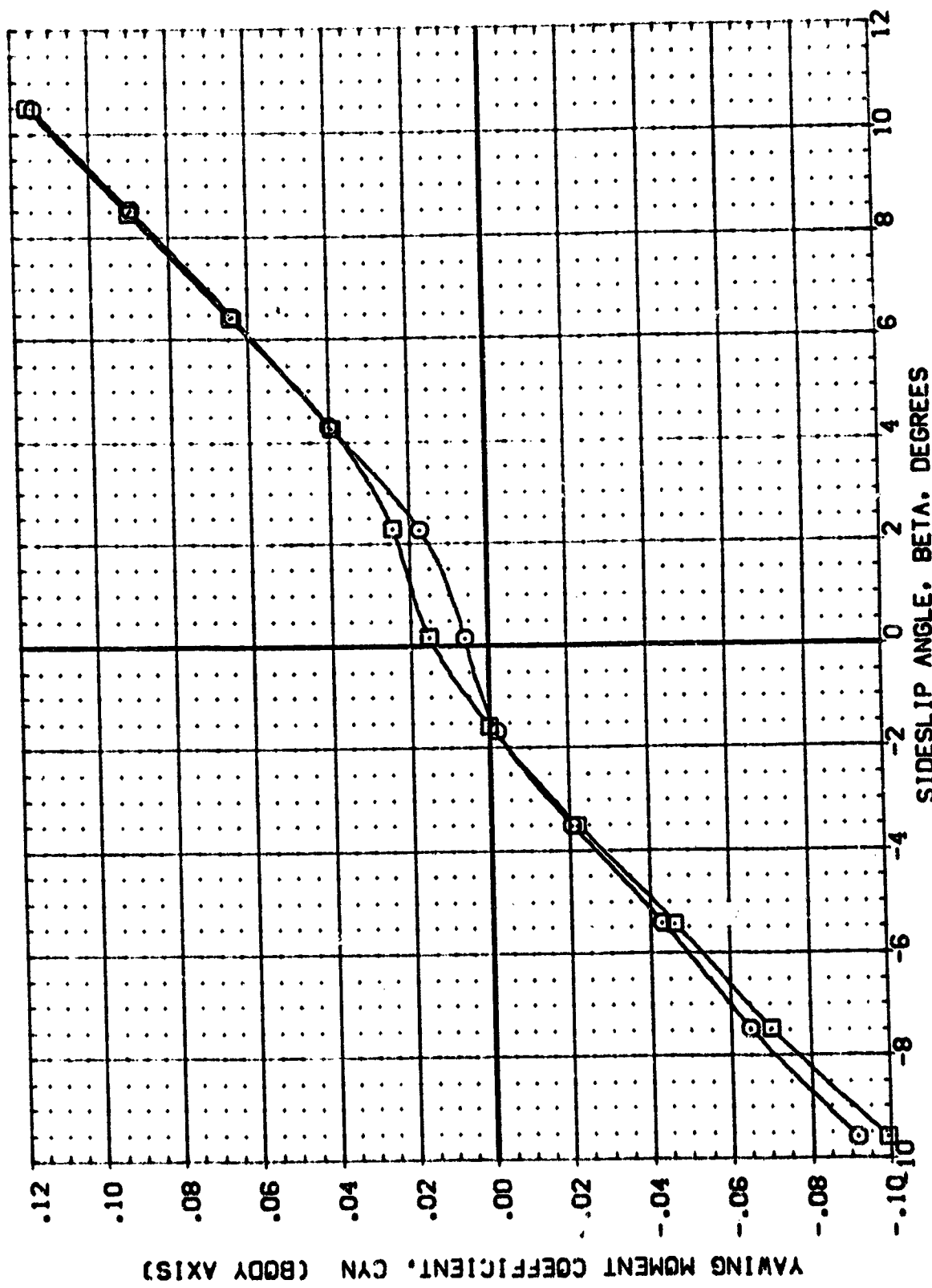


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
(987008)	8	AVES 3.5-169	1A10 09 T10 AT2 PLANE ON	SREF	2690.0000
(987008)		AVES 3.5-169	1A10 09 T10 AT2 PLANE ON	LREF	1250.0000
				SREF	936.6900
				VMRP	1076.4800
				VMRP	0000
				VMRP	0000
				SCALE	0.000

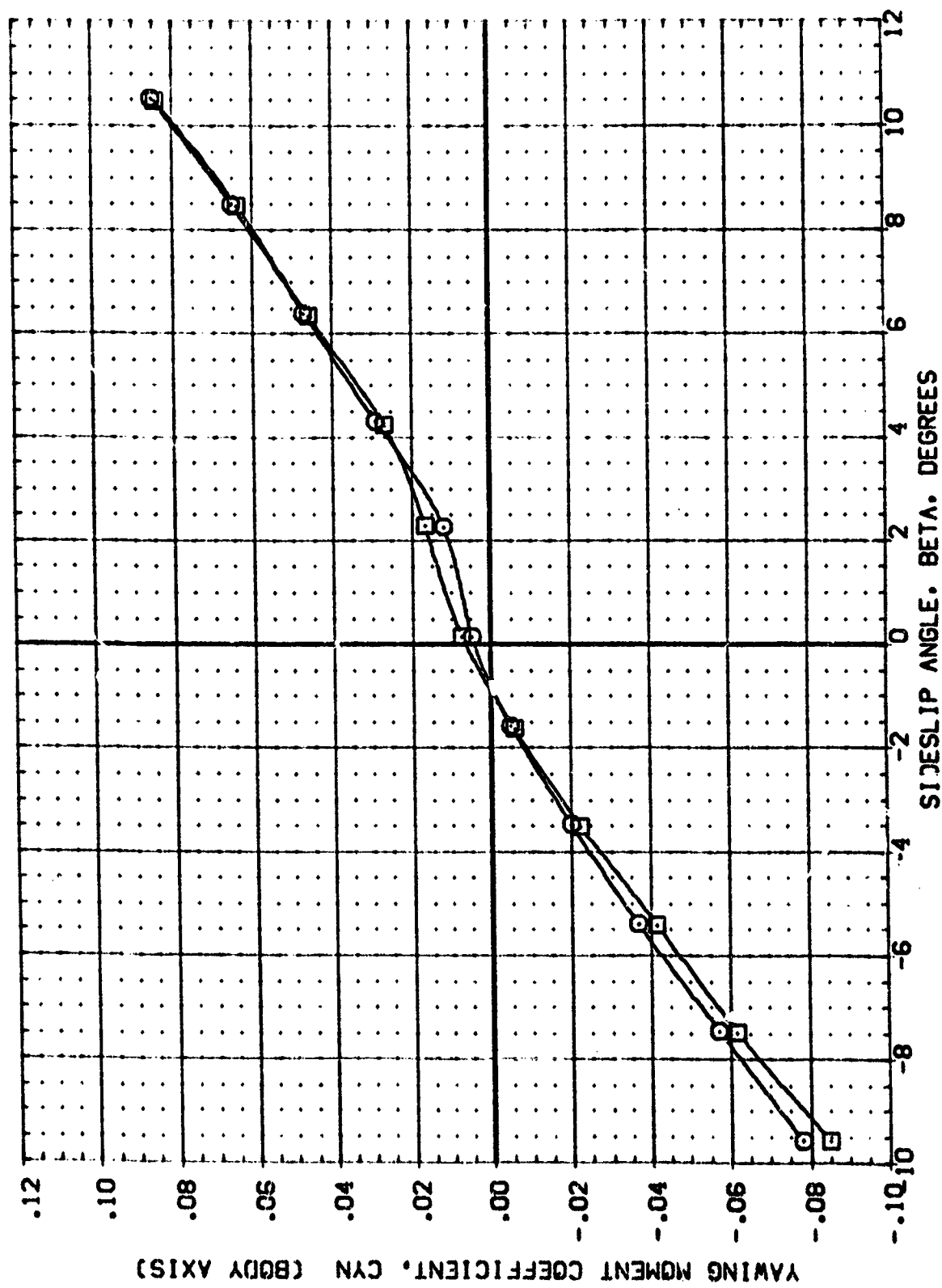


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
(R87009)	AVES 3-5-169	IA10	09 T10 AT2 PLUVE ON	SREF	2690.0000 SQ.FT.
(R87008)	AVES 3-5-169	IA10	09 T10 AT2 PLUVE ON	LREF	1290.0000 IN.
				BREF	536.8600 IN.
				XPRP	1076.4800 IN.
				YPRP	0.0000 IN.
				ZPRP	400.0000 IN.
				SCALE	.0100

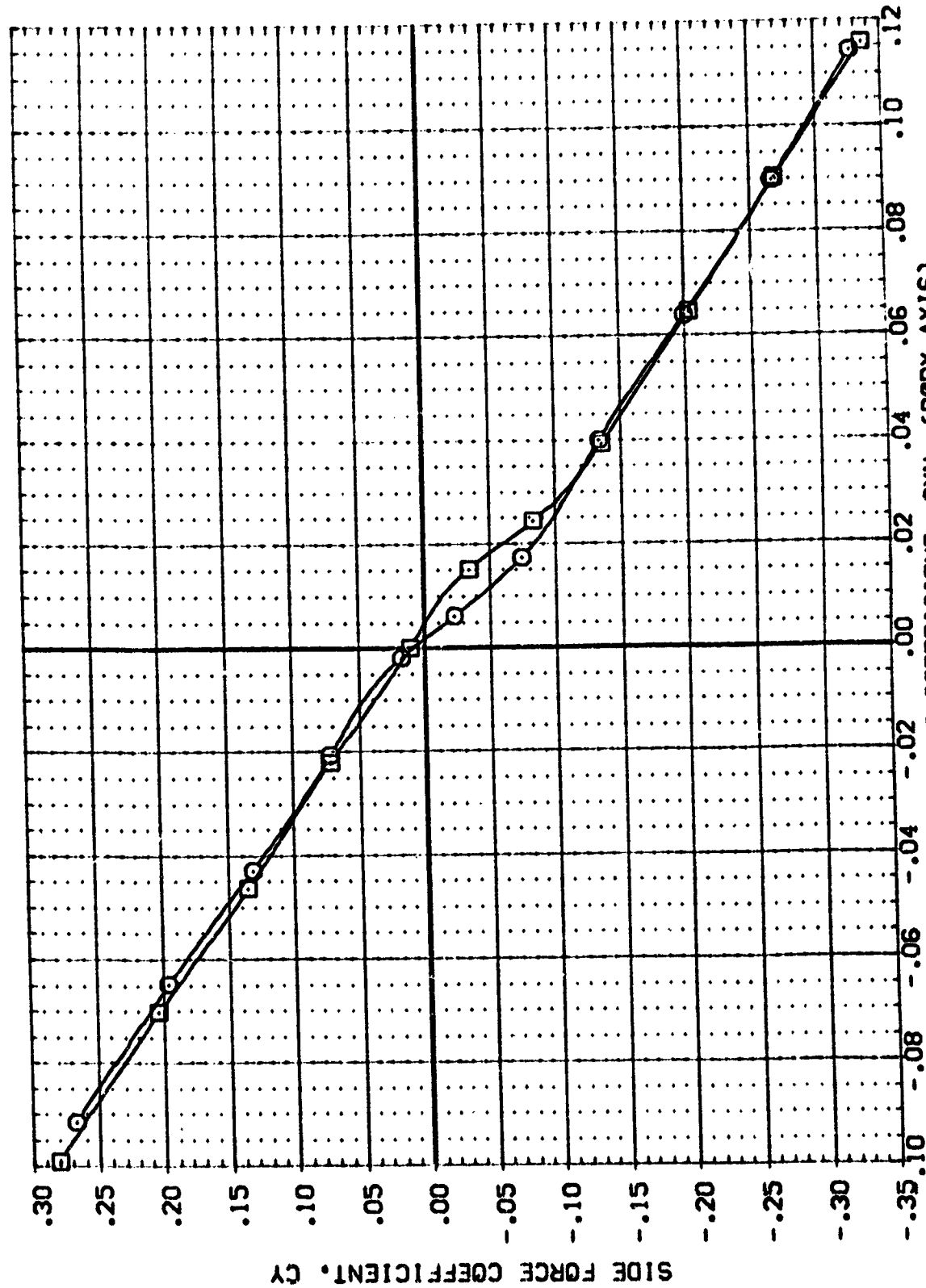


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(A)MACH = 5.26

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		ALPHA		AILRON		ELEVON		RUDDER		REFERENCE INFORMATION	
(R87009)	(C)	AVES 3.5-169	IA10 OS T10 AT2 PLUPE ON	.000	.000	.000	.000	.000	.000	.000	.000	SREF 2690.0000	53. FT.
(R87338)	(C)	AVES 3.5-169	IA10 OS T10 AT2 PLUPE ON	.000	.000	.000	.000	.000	.000	.000	.000	LREF 1290.0000	IN.
												BREF 536.6800	IN.
												YMRP 1076.4800	IN.
												ZMRP .0000	IN.
												SCALE 400.0000	IN.

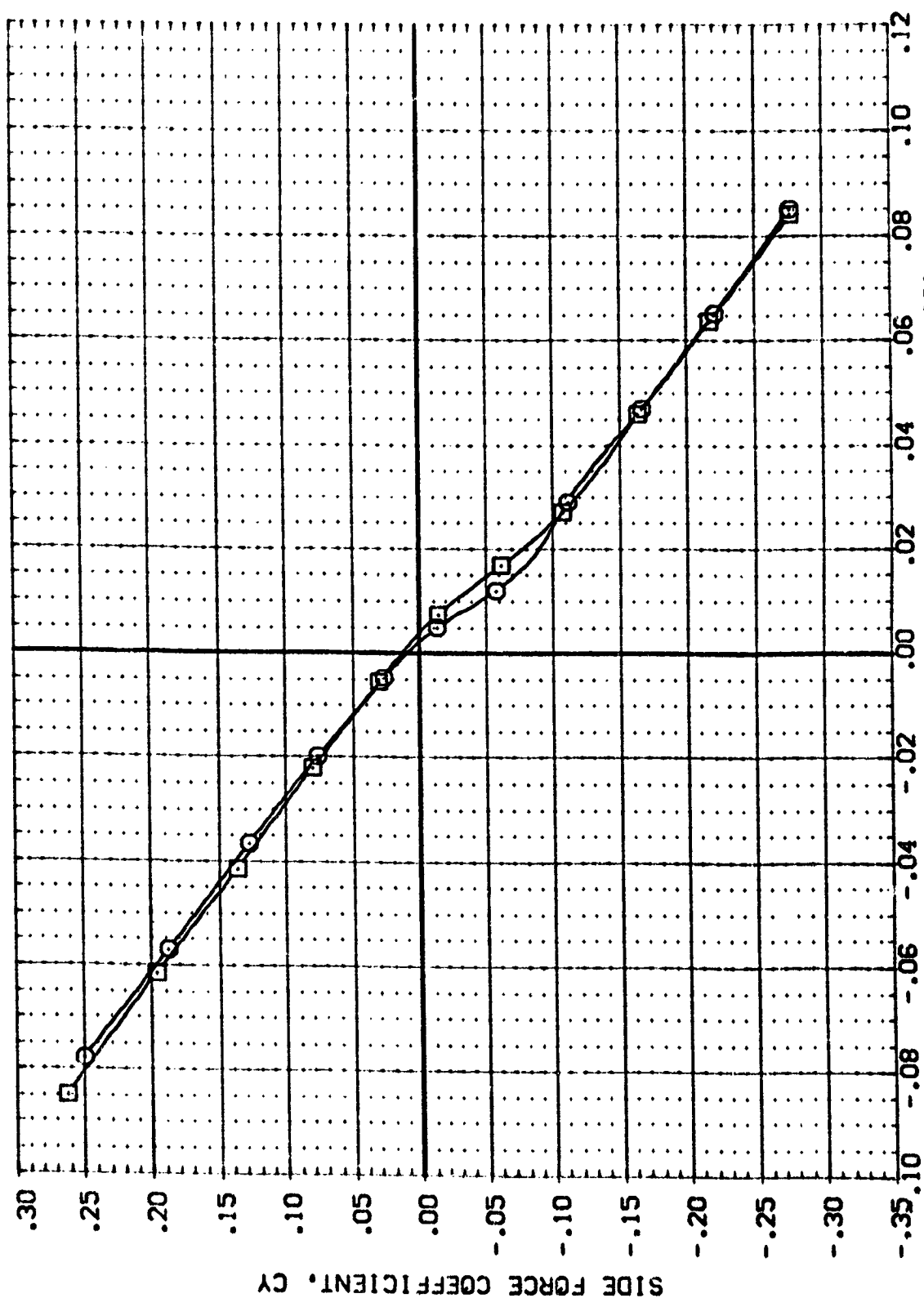


FIG. 10 EFFECT OF RUDDER ON LATERAL-DIRECTIONAL CHARACTERISTICS.

(B)MACH = 7.32

DATA SET SYMBOL:   
 (M87011)   
 (M87005)   
 (M87004)   
 (M87005)   
 CONFIGURATION DESCRIPTION:   
 AVES 3.5-169 IAI0 09 T10 AT2 PLUVE OFF   
 AVES 3.5-169 IAI0 09 T10 AT2 PLUVE OFF   
 DATA NOT AVAILABLE   
 DATA NOT AVAILABLE

BETA: .000 .000 .000 .000   
 AIRLON: .000 .000 .000 .000   
 ELEVON: .000 .000 .000 .000   
 RUDDER: .000 .000 .000 .000   
 REFERENCE INFORMATION:   
 SREF: 2690.0000 SO.FT.   
 LREF: 1290.0000 IN.   
 BREF: 936.6800 IN.   
 XMRP: 1076.4800 IN.   
 YMRP: 400.0000 IN.   
 ZMRP: 400.0000 IN.   
 SCALE: .0100

LOCAL NORMAL FORCE COEFFICIENT DERIVATIVE WITH ALPHA, CN/A, PER DEGREE

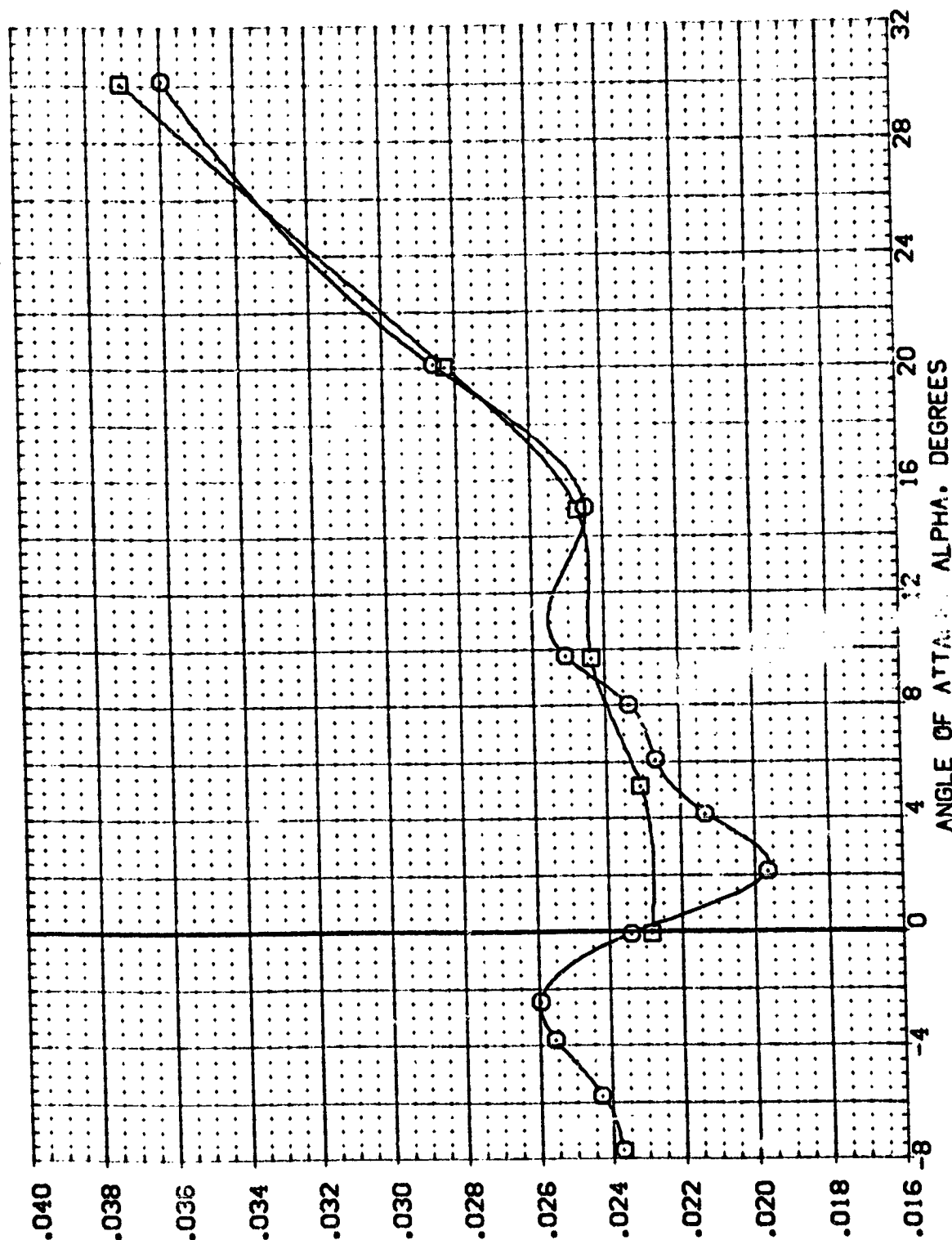


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(M970111) DATA NOT AVAILABLE  
 (V97005) DATA NOT AVAILABLE  
 (M97004) ARES 3.5-169 1A10 CS T10 AT2 PLANE OFF  
 (V97005) ARES 3.5-169 1A10 CS T10 AT2 PLANE OFF

BETA AILRON ELEVON RUDDER REFERENCE INFORMATION

.000 .000 .000 SREF 2650.0000 SG.FT.  
 .000 .000 .000 LREF 1250.0000 IN.  
 .000 .000 .000 BREF 936.6800 IN.  
 .000 .000 .000 XMRP 1073.4800 IN.  
 .000 .000 .000 YMRP 400.0000 IN.  
 .000 .000 .000 ZMRP 400.0000 IN.  
 SCALE .0100

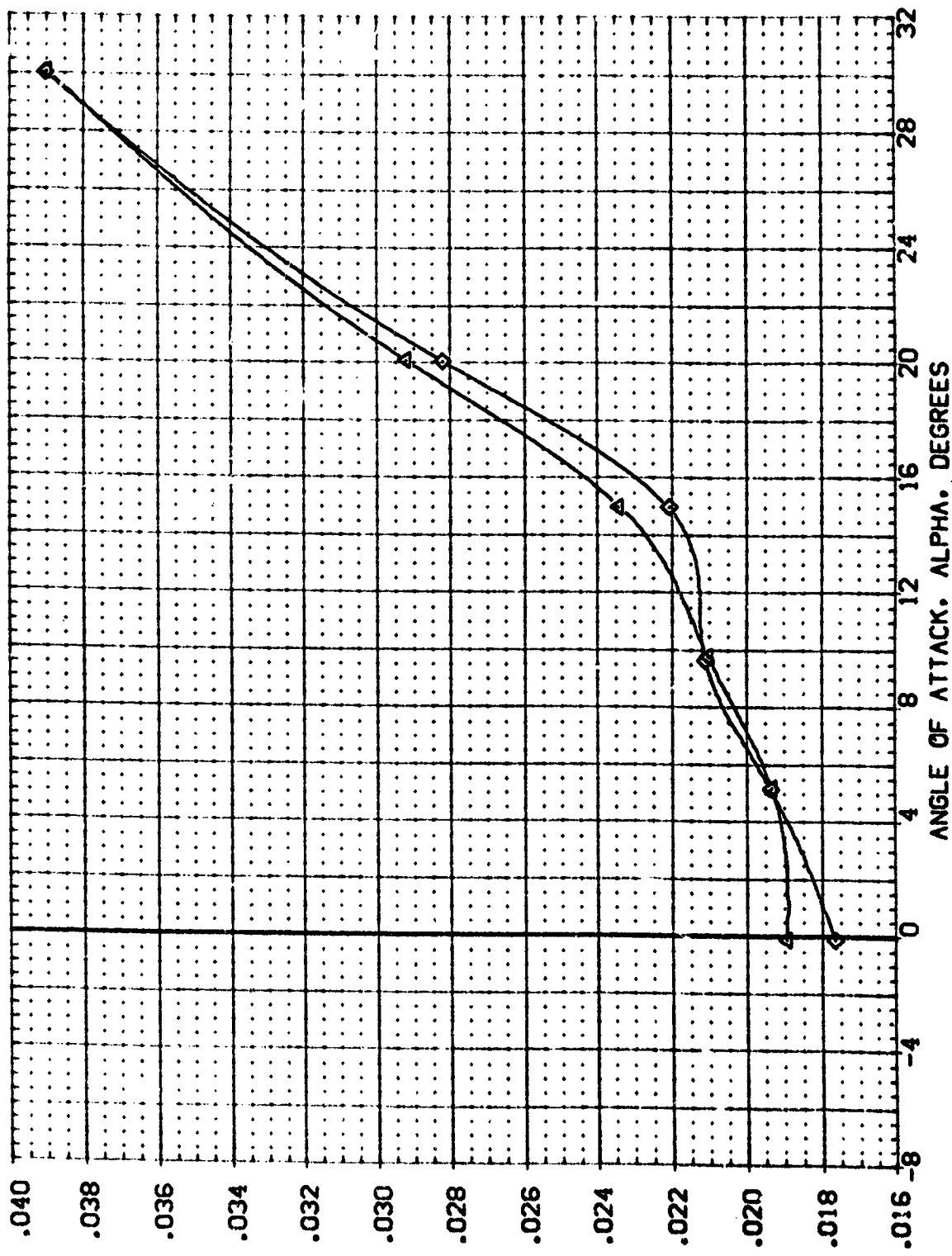


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(B)MACH = 7.32

DATA SET SYMBOL: (M87011) (M87005) (M87004) (M87003)

CONFIGURATION DESCRIPTION: ANES 3.5-169 IAD 09 TIC AT2 PLUVE OFF  
 ANES 3.5-169 IAD 09 TIC AT2 PLUVE OFF  
 DATA NOT AVAILABLE  
 DATA NOT AVAILABLE

BETA: .000 .000 .000 .000

AILRON: .000 .000 .000 .000

ELEVON: .000 .000 .000 .000

FLUDER: .000 .000 .000 .000

REFERENCE INFORMATION: SREF 2690.0000 SQ.FT. IN.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

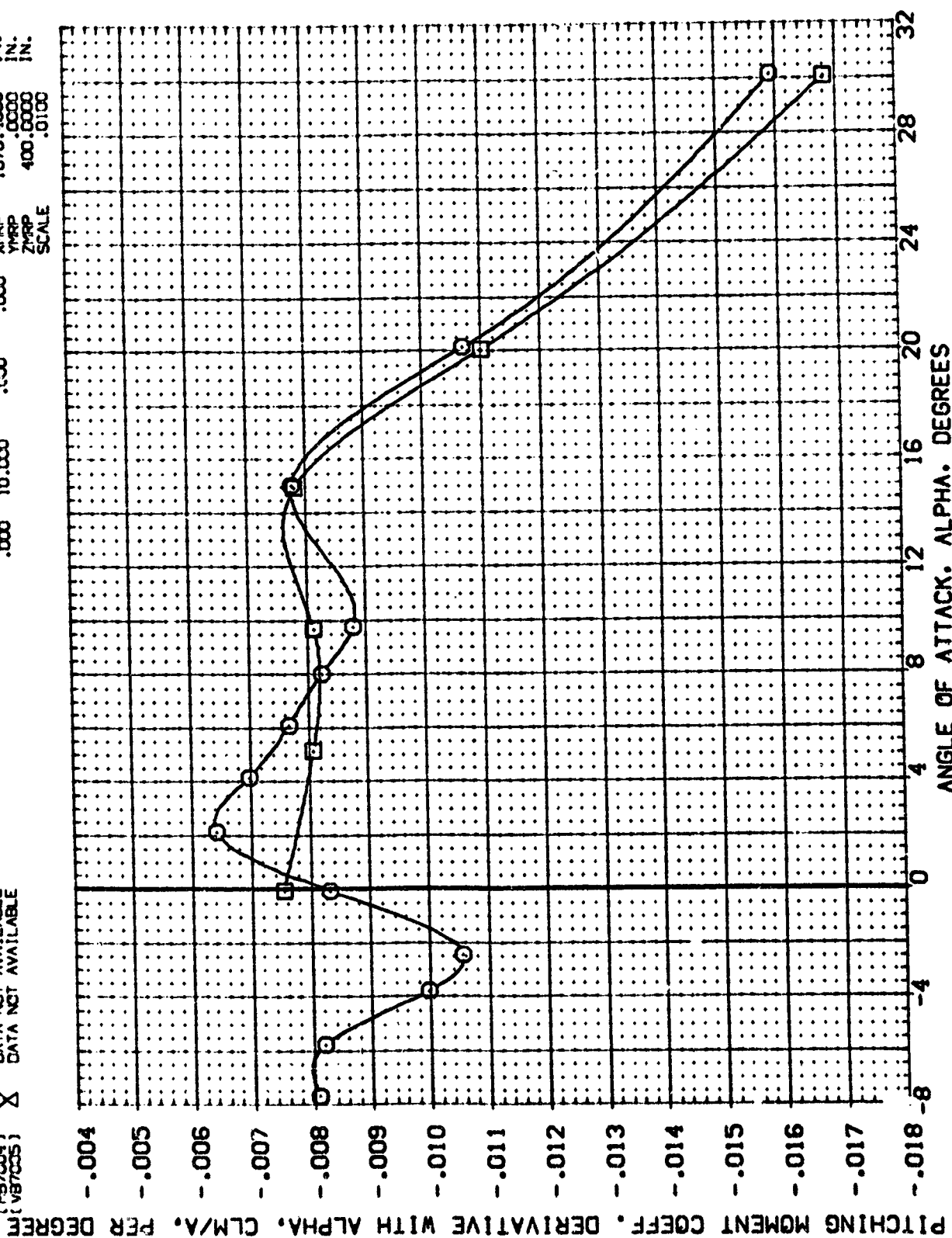


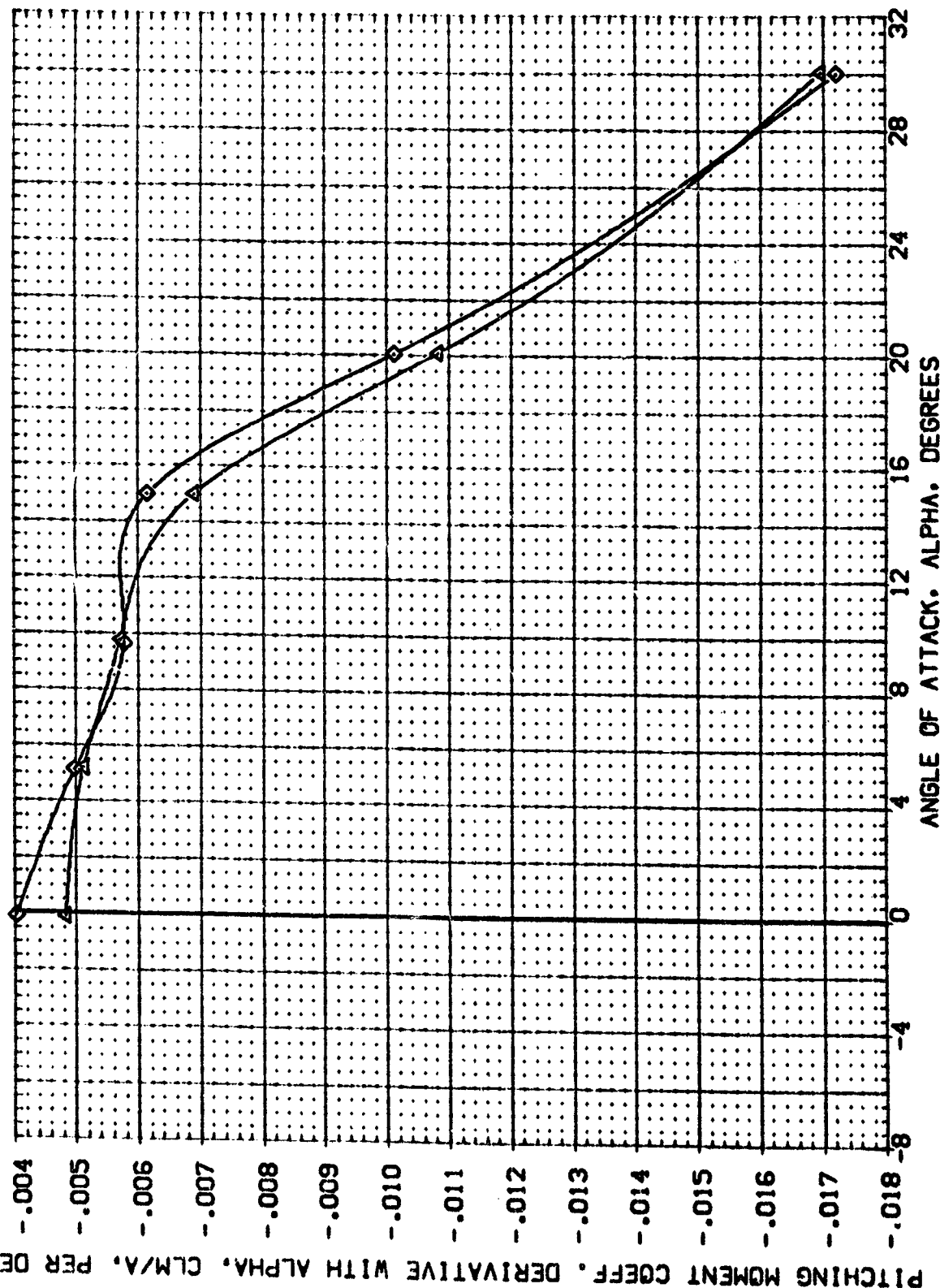
FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26



DATA NOT AVAILABLE	09	T10	AT2	PLUME	OFF
AMES 3.5-169	IA10				
AMES 3.5-162	IA10				

BETA	AILRON	ELEVON	RUDDER	REFERENCE INFORMATION	SQ. FT.
.000	.000	.000	.000	SREF	2690.0000
.000	.000	.000	.000	LRUF	1290.0000
.000	10.000	.000	.000	LRBF	936.6800
.000	.000	.000	.000	XRFP	1076.4800
.000	10.000	.000	.000	YRFP	.0000
				ZRFP	.0000
				SCALE	.0100



**FIG. 11 ELEVON EFFECTIVENESS SUMMARY.**

**[B]MACH = 7.32**

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	AILERON	ELEVON	RUDDER	REFERENCE INFORMATION
(M7011)	WES 3.5-169 1A10 09 T10 AT2 PLANE OFF	.000	.000	.000	.000	SREF 2680.0000 SQ.FT.
(M7005)	WES 3.5-169 1A10 09 T10 AT2 PLANE OFF	.000	10.000	.000	.000	LREF 1290.0000 IN.
(M7004)	DATA NOT AVAILABLE	.000	.000	.000	.000	BREF 936.6800 IN.
(M7005)	DATA NOT AVAILABLE	.000	10.000	.000	.000	VMRP 1076.4800 IN.
						ZMRP 400.0000 IN.
						SCALE .0100

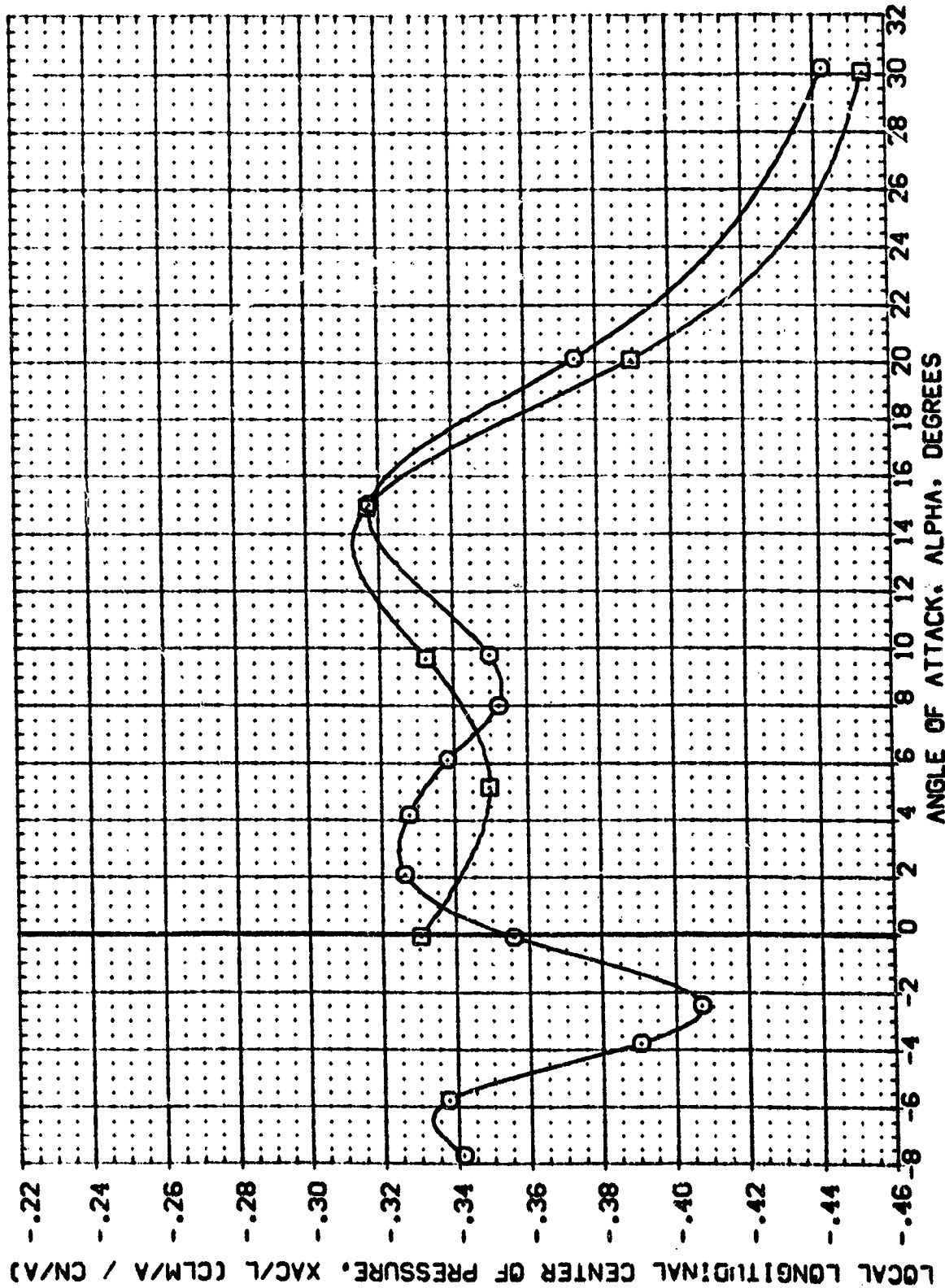


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(A)MACH = 5.26

DATA SET SYMBOL    CONFIGURATION DESCRIPTION  
 (M87011)    DATA NOT AVAILABLE  
 (UB7005)    DATA NOT AVAILABLE  
 (M87004)    AVES 3.5-169 1A10    09 T10 AT2 PLANE OFF  
 (UB7005)    AVES 3.5-169 1A10    09 T10 AT2 PLANE OFF

BETA    ALL/ROV    ELEVON    RUDDER    REFERENCE INFORMATION  
 .000    .000    .000    .000    SREF 2690.0000    SQ.FT.  
 .000    .000    .000    .000    LREF 1250.0000    IN.  
 .000    .000    .000    .000    BREF 936.6800    IN.  
 .000    .000    .000    .000    XPRP 1076.4800    IN.  
 .000    .000    .000    .000    YPRP .0000    IN.  
 .000    .000    .000    .000    ZPRP 400.0000    IN.  
 .000    .000    .000    .000    SCALE .0100

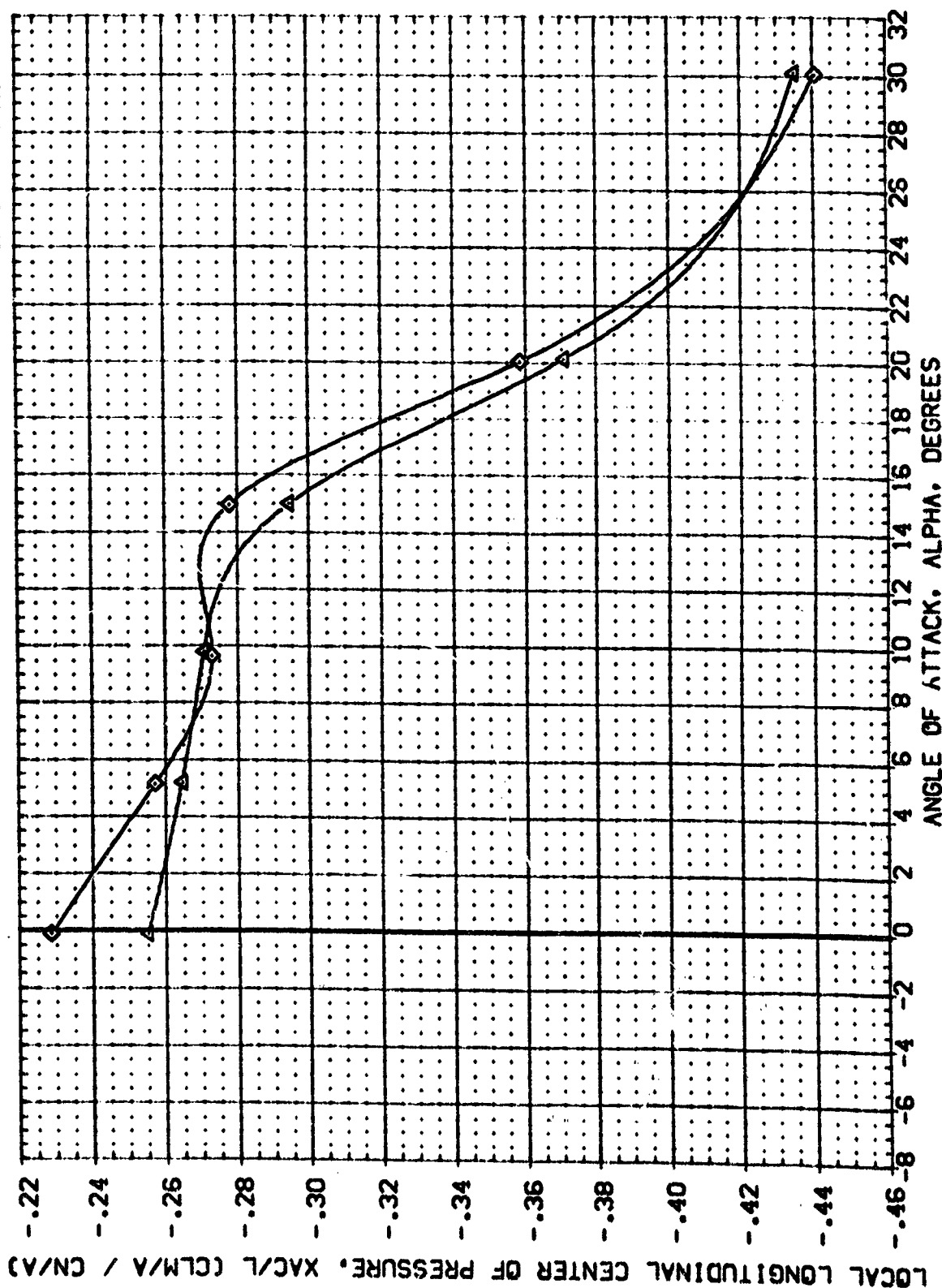


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(B)MACH = 7.32



AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF (087005)

SYMBOL	MACH	BETA	PARAMETRIC VALUES	REFERENCE INFORMATION
○	5.300		.000	SREF 2680.0000 SO.FT.
□	7.300			LREF 1250.0000 IN.
				BRF 936.6800 IN.
				XMRP 1076.4800 IN.
				YMRP .0000 IN.
				ZMRP 400.0000 IN.
				SCALE .0100

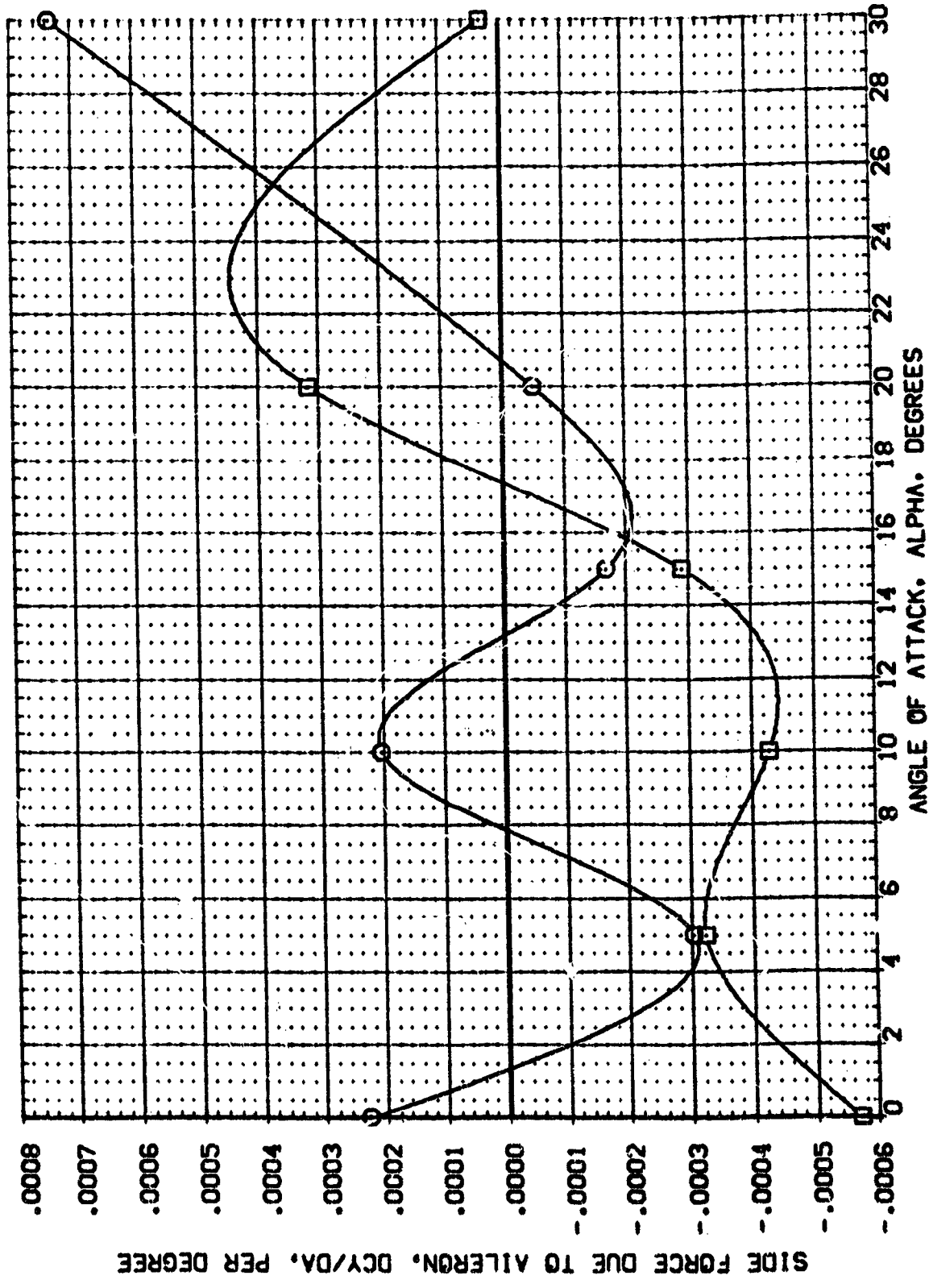


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF (QB7005)

REFERENCE INFORMATION  
 SREF 2630.0000 SO.FT.  
 LREF 1250.0000 IN.  
 BREF 936.6800 IN.  
 YMRP 1076.4800 IN.  
 ZMRP .0000 IN.  
 SCALE 400.0000 IN.  
 .0100

SYMBOL MACH BETA  
 ○ 5.300  
 □ 7.300

PARAMETRIC VALUES  
 .000

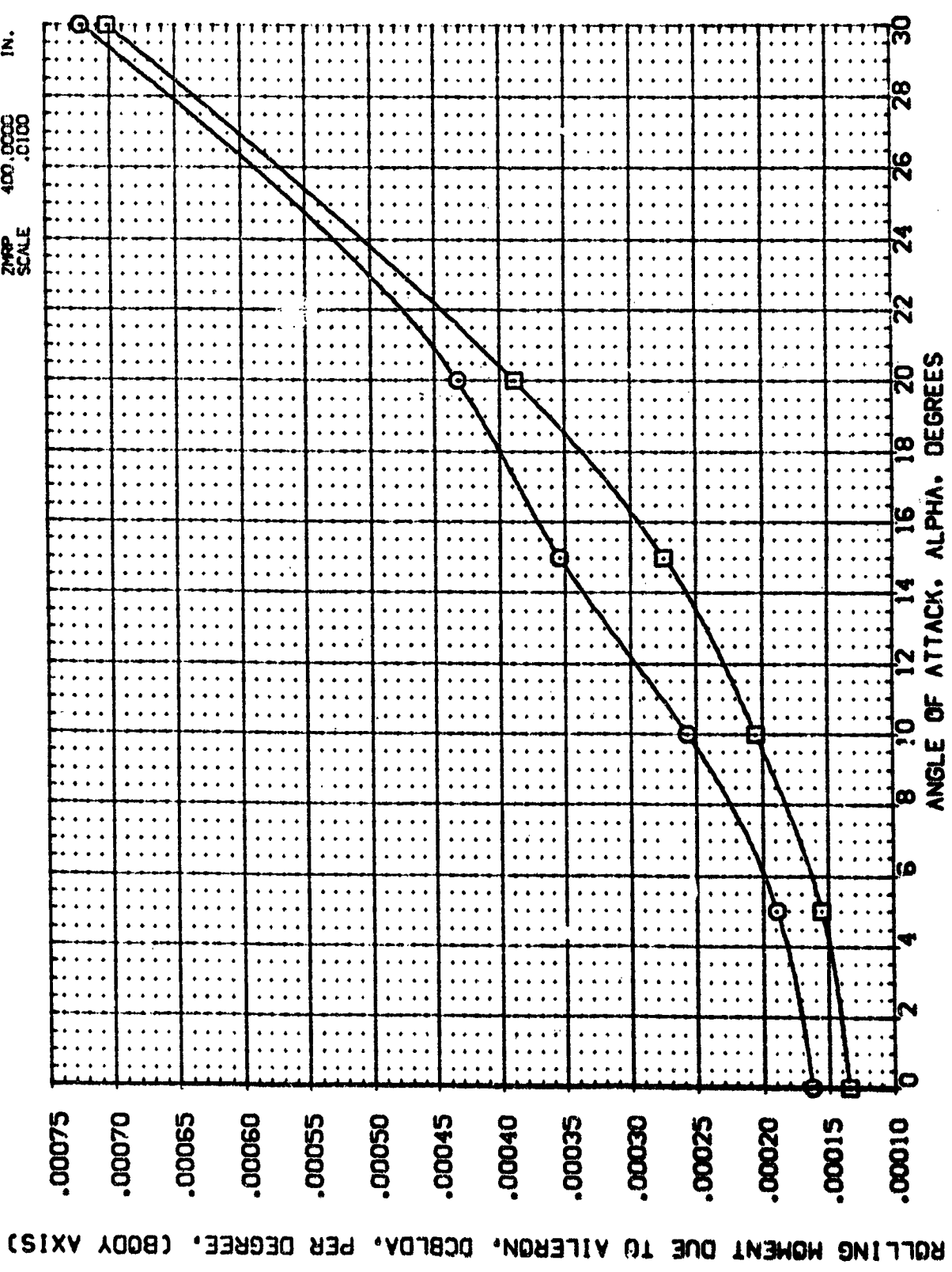


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

(087005)

SYMBOL  $\square$   $\circ$  NACH 5.300 7.300 BETA .000

PARAMETRIC VALUES

REFERENCE INFORMATION

	SO. FT.
SREF	2690.0000
LREF	1290.0000
BREF	936.8800
XMRP	1076.4800
YMRP	.0000
ZMRP	400.0000
SCALE	.0100

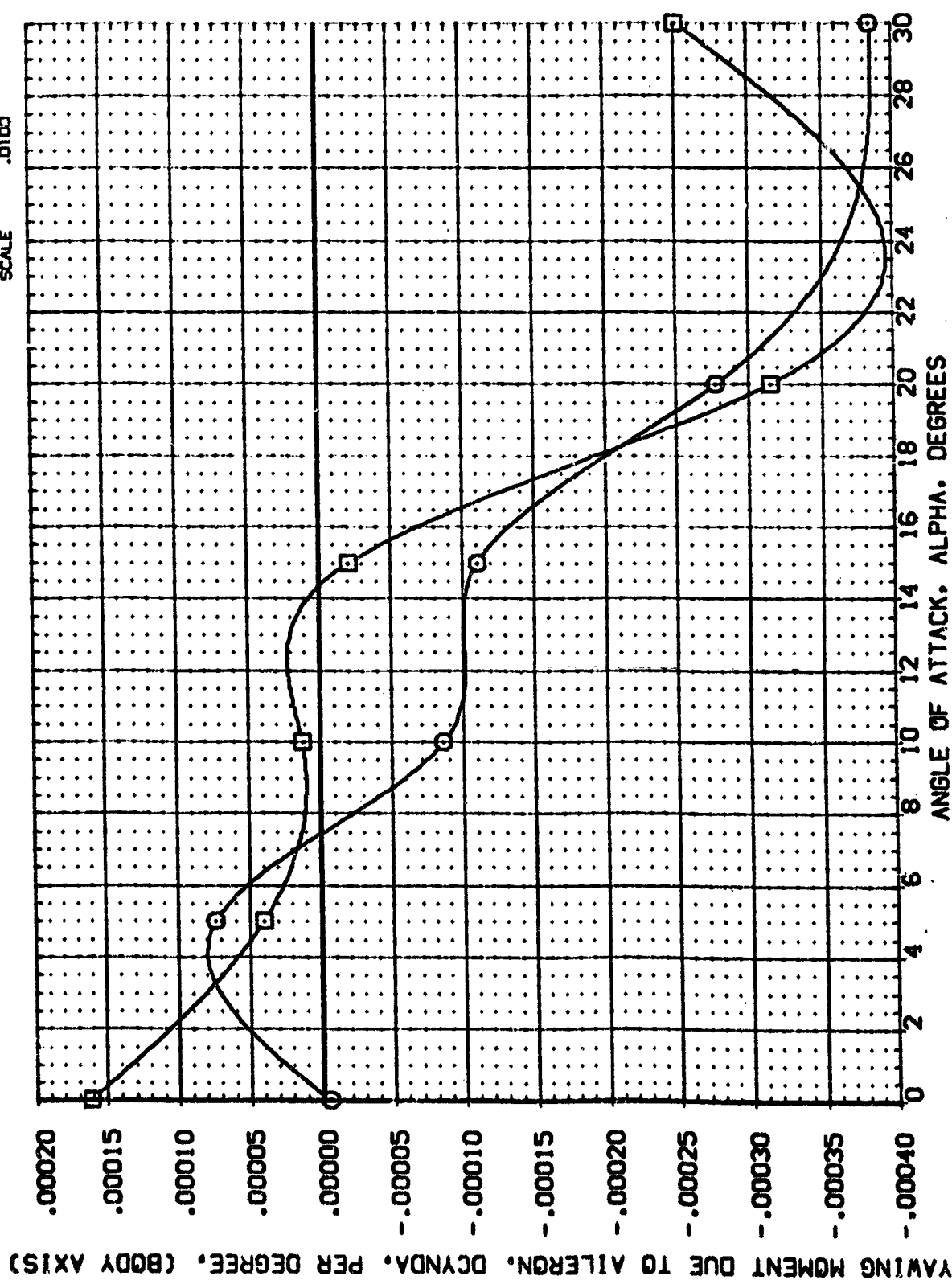


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(087005)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

PARAMETRIC VALUES

SYMBOL MACH BETA  
□ 5.300  
○ 7.300

REFERENCE INFORMATION  
SREF 2690.0000 SQ.FT.  
LREF 1290.0000 IN.  
BREF 936.6900 IN.  
YREF 1076.4800 IN.  
ZREF 400.0000 IN.  
SCALE .0100

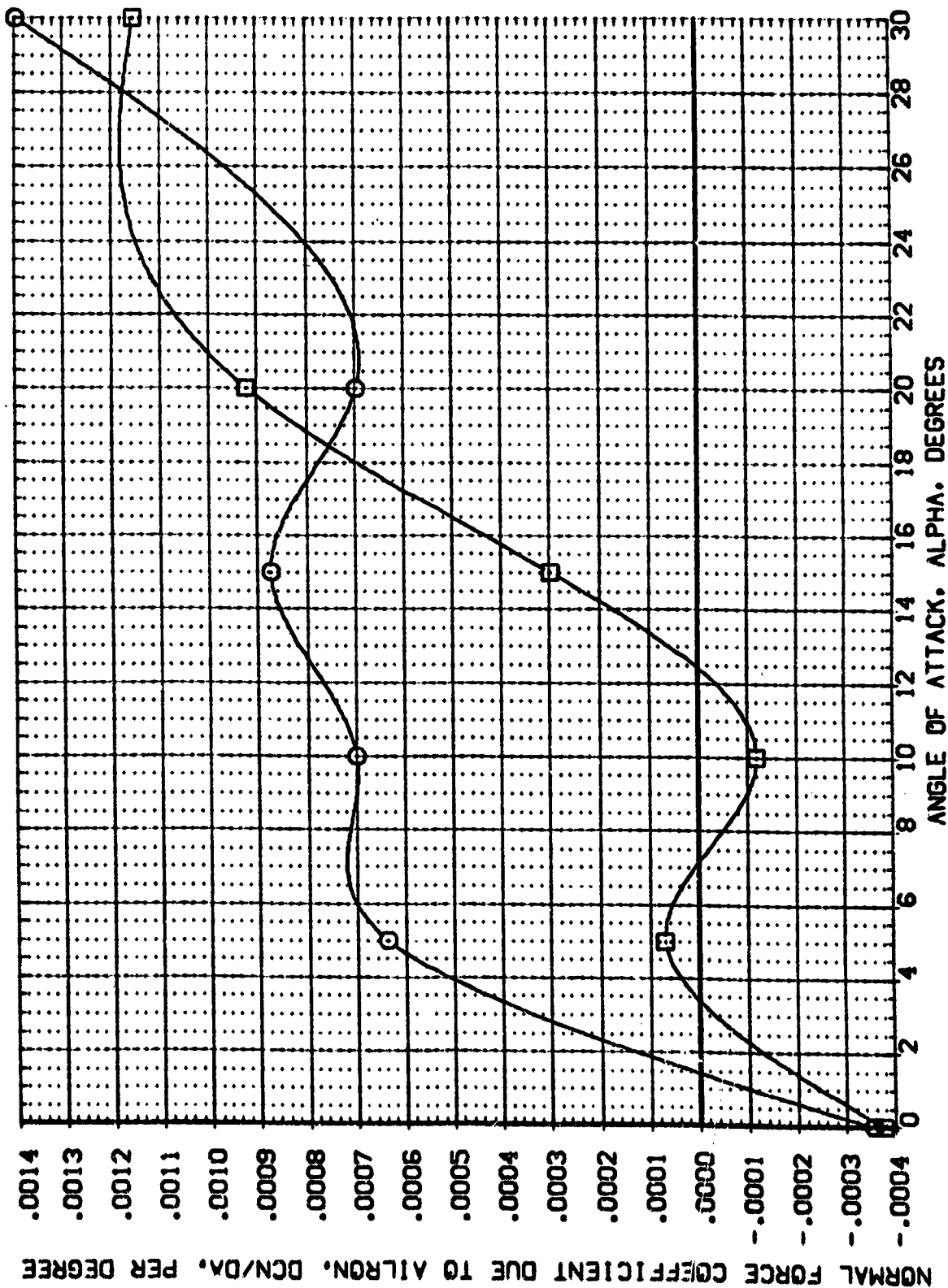


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.

(QB7005)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XPRP 1076.4800 IN.  
 YPRP .0000 IN.  
 ZPRP 400.0000 IN.  
 SCALE .0100

PARAMETRIC VALUES

BETA

MACH 5.300  
7.300

SYMBOL  
 ○  
 □

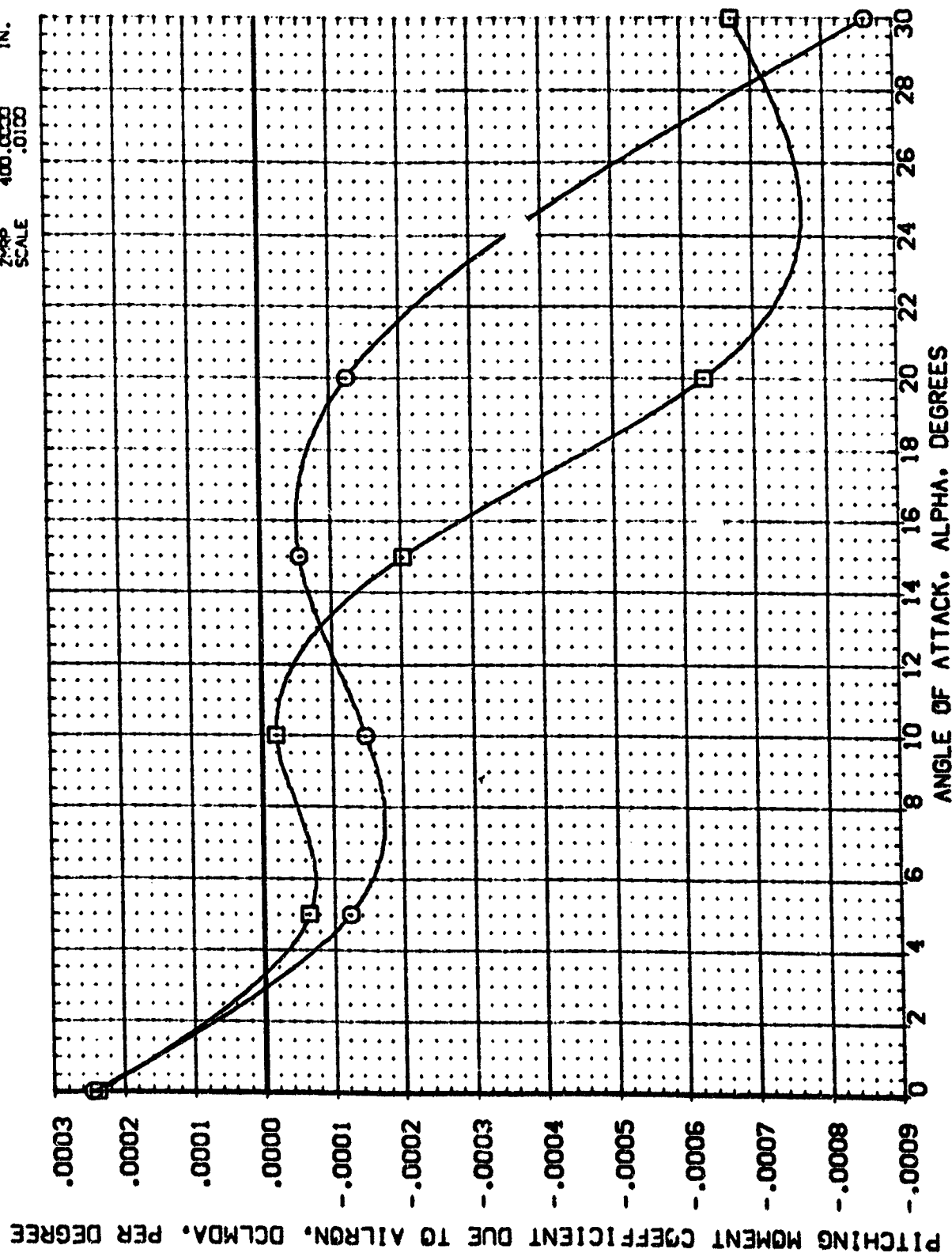


FIG 11 ELEVON EFFECTIVENESS SUMMARY.



(QB7005)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

SYMBOL MACH BETA PARAMETRIC VALUES  
□ 5.300 .000  
□ 7.300

REFERENCE INFORMATION  
SREF 2650.0000 SQ.Ft.  
REF 1250.0000 IN.  
REF 936.6800 IN.  
REF 1076.4800 IN.  
YPRP .0000 IN.  
ZPRP 400.0000 IN.  
SCALE .0100

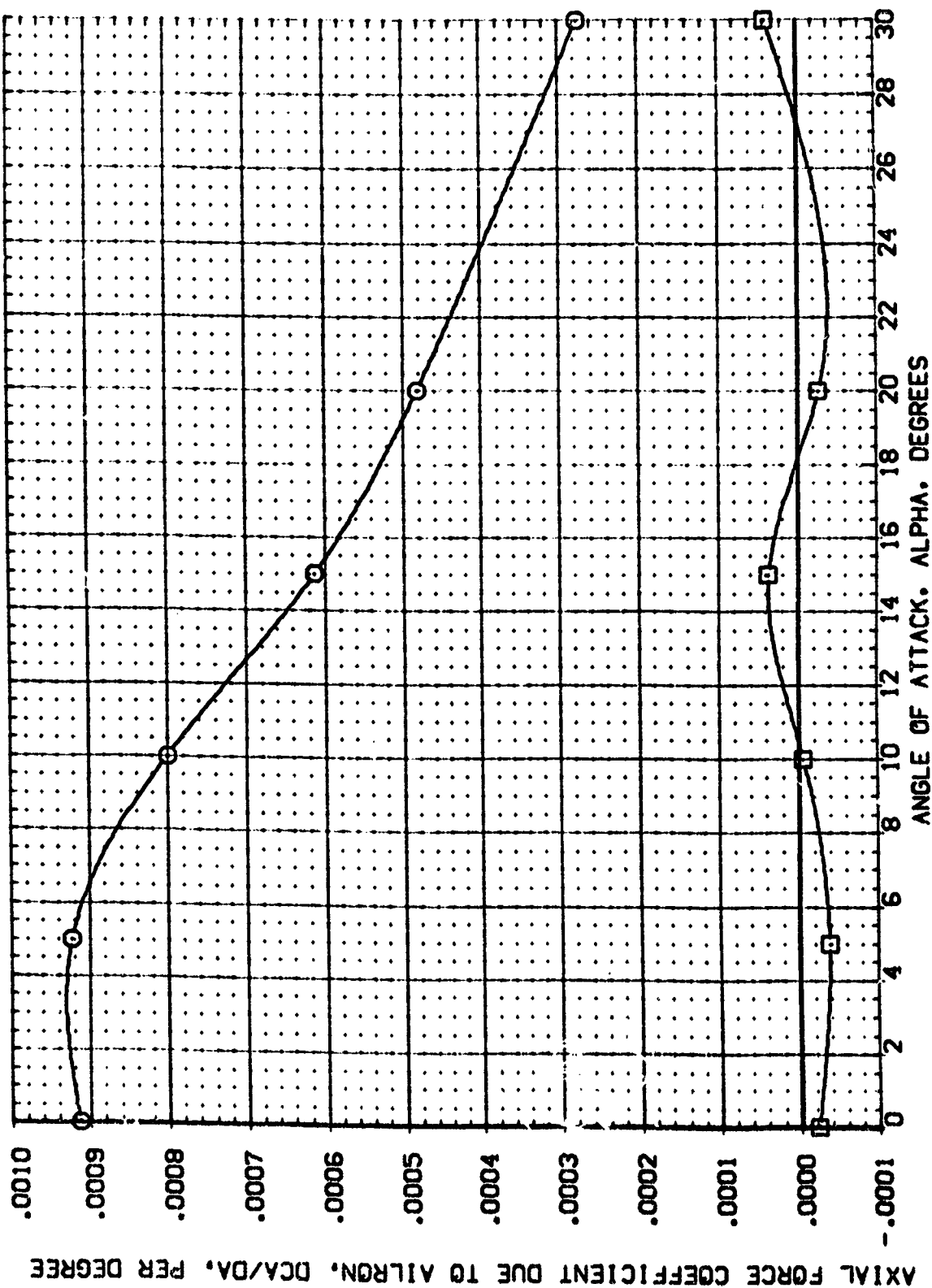


FIG. 11 ELEVON EFFECTIVENESS SUMMARY.



(087002)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

SYMBOL MACH BETA  
□ 5.300 5.000  
□ 7.300

PARAMETRIC VALUES

REFERENCE INFORMATION  
SREF 2690.0000 SQ.FT.  
LREF 1290.0000 IN.  
BREF 936.6800 IN.  
XPRP 1076.4800 IN.  
YPRP 400.0000 IN.  
ZPRP 400.0000 IN.  
SCALE .0100

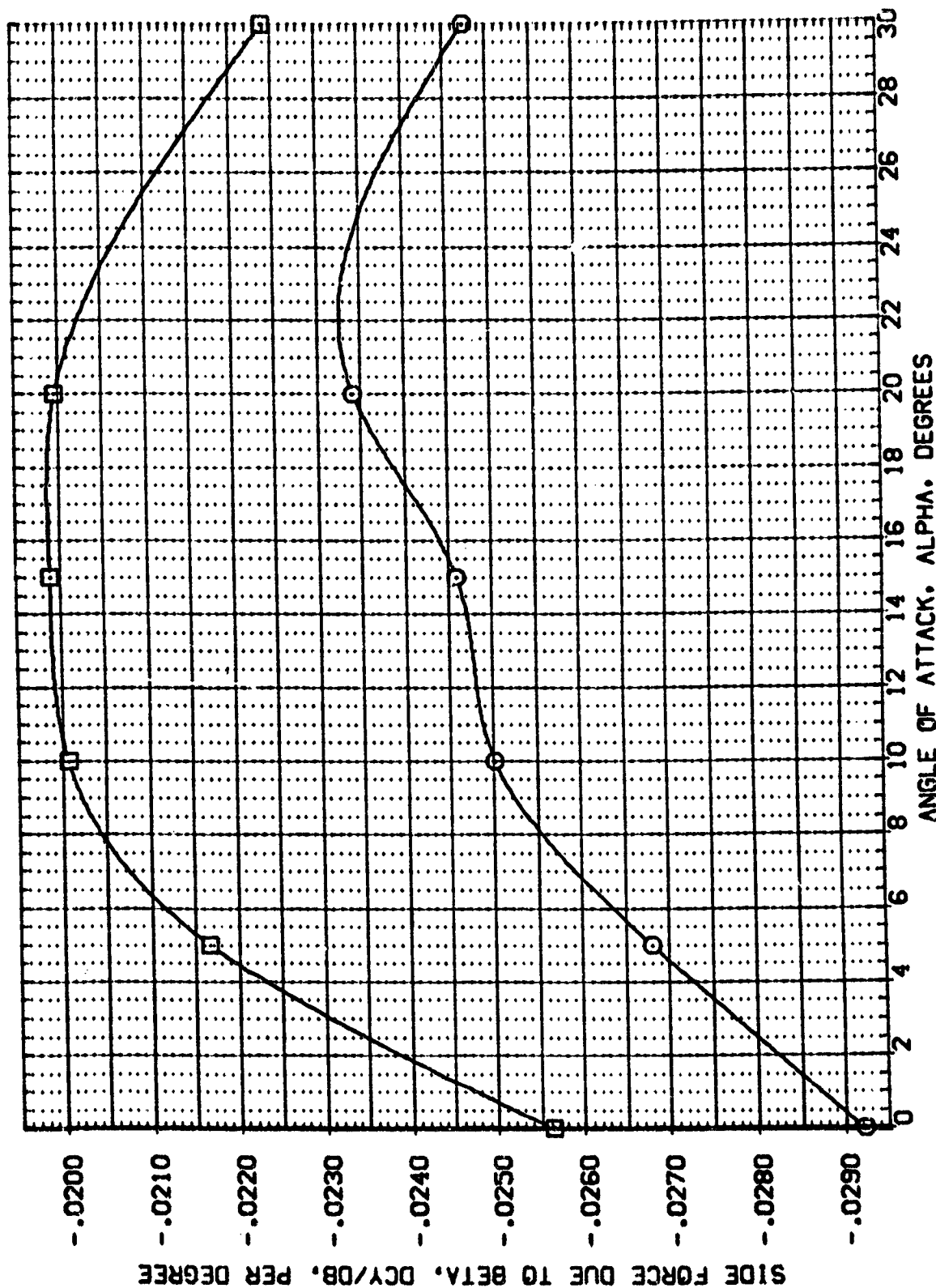


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACTER.

(QB7002)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

SYMBOL MACH PARAMETRIC VALUES  
 □ 5.300 BETA 5.000  
 □ 7.300

REFERENCE INFORMATION  
 SREF 2690.0000 SQ.FT.  
 LREF 1290.0000 IN.  
 BREF 936.6800 IN.  
 XMRP 1076.4800 IN.  
 YMRP .0000 IN.  
 ZMRP 400.0000 IN.  
 SCALE .0100

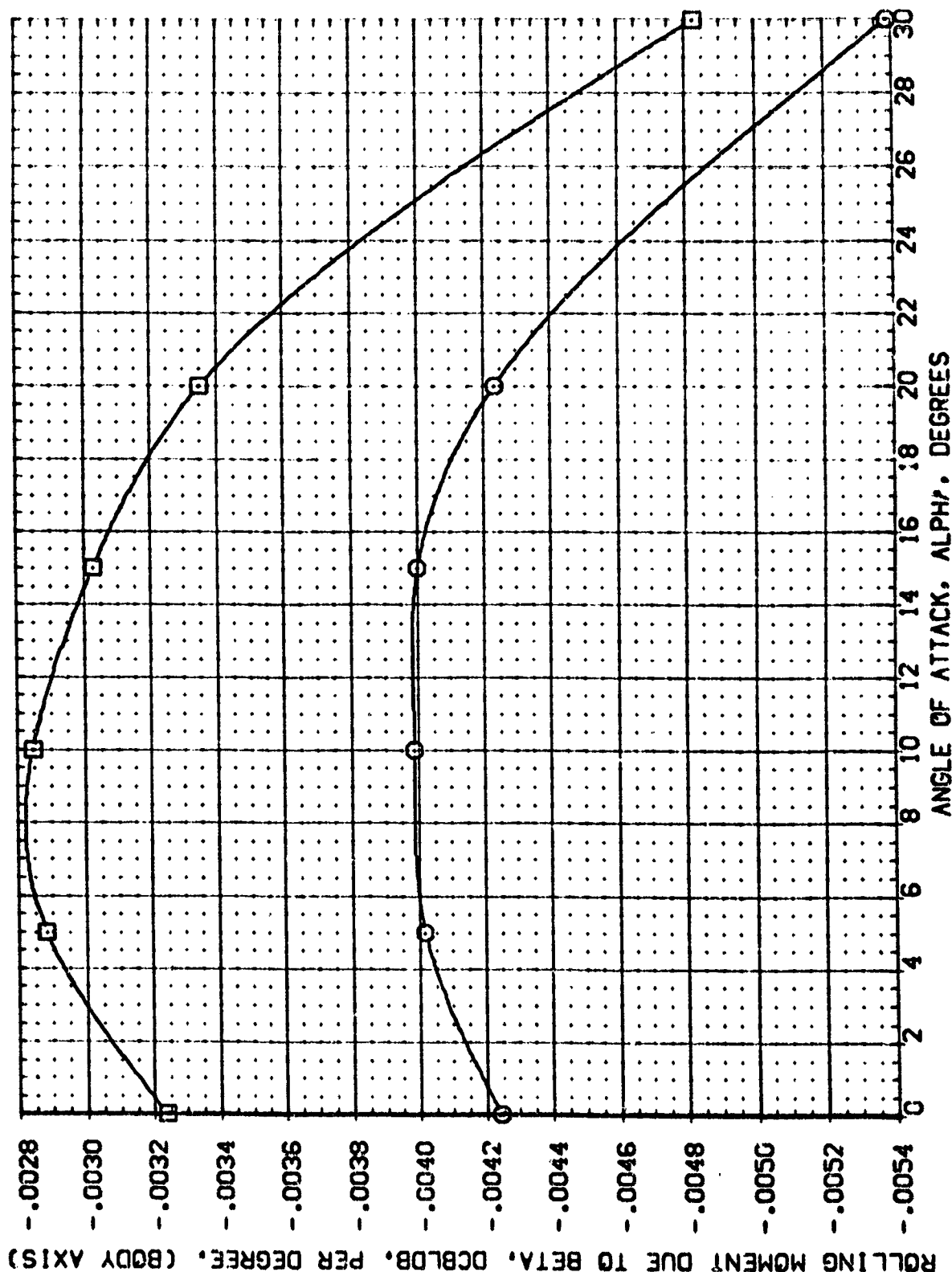


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACTER.



(QB7002)

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

PARAMETRIC VALUES

SYMBOL MACH BETA  
○ 5.300  
□ 7.300

REFERENCE INFORMATION  
SREF 2690.0000 SQ.FT.  
LREF 1290.0000 IN.  
BREF 936.6800 IN.  
XPRP 1576.4800 IN.  
YPRP .0000 IN.  
ZPRP 400.0000 IN.  
SCALE .0100

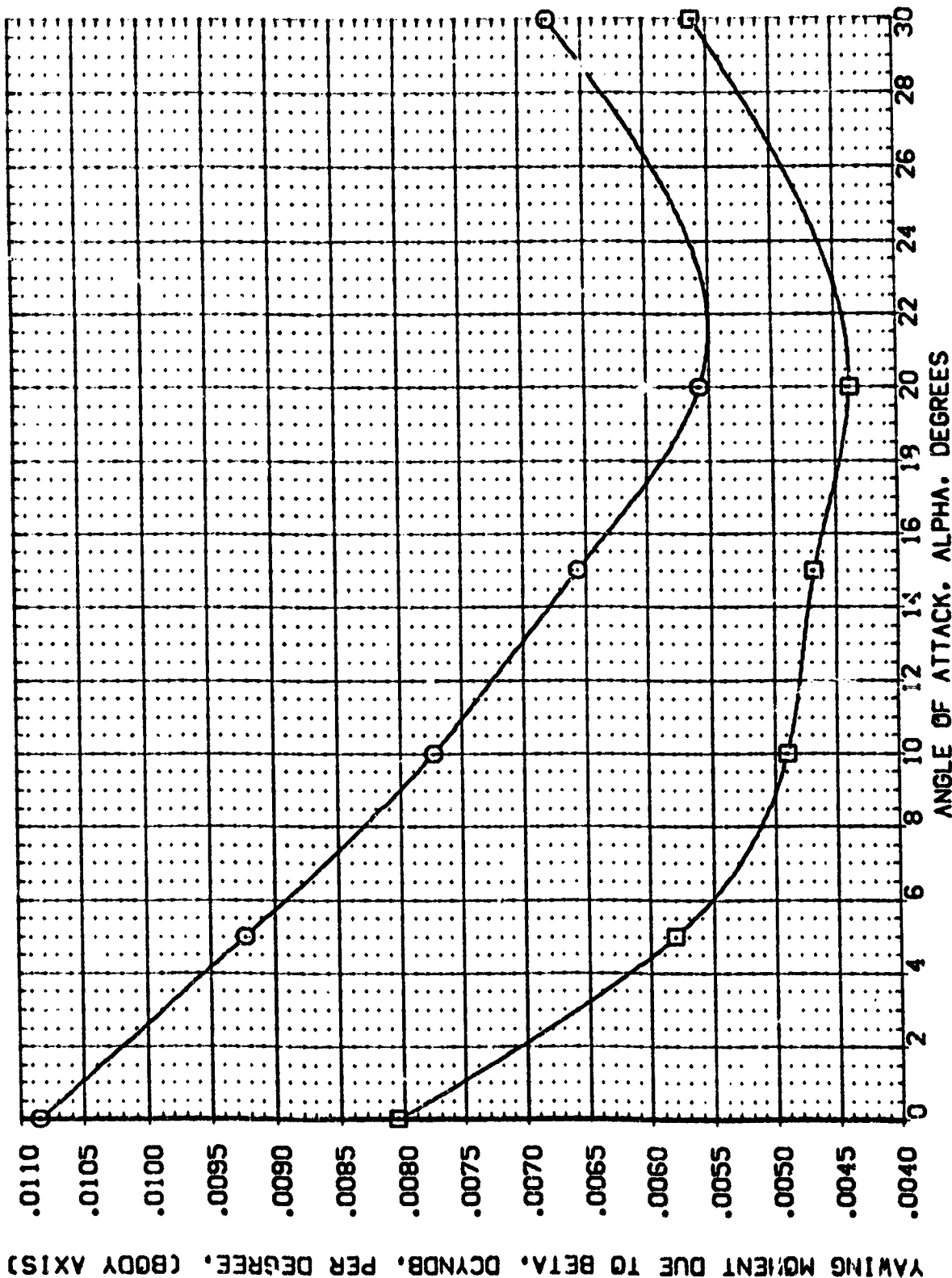


FIG. 12 SUMMARY OF ANGLE OF ATTACK EFFECTS ON LATERAL-DIRECTIONAL CHARACTER.

## APPENDIX

### TABULATED SOURCE DATA

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Tabulations of plotted data are available on request from  
Data Management Systems

TRANSLATED SOURCE ARC 3.5-100, 1A10

SOURCE ARC 3.5-169, 1A10  
 ANES 3.5-169 1A10 09 T10 AT2 PLANE OFF

### PARAMETRIC DATA

BETA	=	.000	AILRON	=	.000
ELEVEN	=	.000	BOFLAP	=	.000
SPRDER	=	.000	RUDDER	=	.000

## REFERENCE DATA

967P = 2690.0000 50.17. 196PP = 1076.4000 IN.  
 196P = 1290.0000 IN. 196PP = .0000 IN.  
 266P = 936.6000 IN. 266PP = 400.0000 IN.  
 SCALE = .0100

$\text{SE}_{\text{B}} = 4.13$  GRADIENT INTERVAL = -5.00/ 5.00

PARAMETER	7.320	ALPHA	ON	CT	CA	CLM	CYN	CEL	CL	CD	L/D
PARAM	7.320										
ALPHA	-9.580										
ON	-25482			-01010	21747	09320	07365	-00193	-21486	25661	-83671
CT	-21672			-00964	20424	08651	07361	-00201	-19068	23067	-82661
CA	-17336			-03614	19355	07158	00239	-00162	-15463	20463	-74046
CLM	-3.539			-00610	18164	05909	00139	-00149	-12260	16922	-64699
CYN	-3.384			-00727	17063	04430	00151	-00136	-08879	17306	-51303
CEL	-1.516			-09334	16151	03328	00063	-00106	-05710	16134	-33393
CL	-0.9684			-03466	14979	02163	00127	-00107	-02245	14988	-19070
CD	2.413			-00714	14169	01446	00066	-00087	-00556	14254	03904
L/D	4.291			-00580	13785	00404	00103	-00075	04080	14339	28432
PARAM	6.507			-00594	13785	00628	00048	-00037	07417	14295	51680
ALPHA	8.591			-00510	13727	00628	00048	-00037	07417	14295	51680
ON	09489			-00510	13727	00628	00048	-00037	07417	14295	51680
CT	13597			-00444	12494	-01785	-00020	-00243	11565	14763	74831
CA	01954			-00016	-00516	-00573	-00016	00007	01654	-07670	06932
CLM											
CYN											
CEL											
CL											
CD											
L/D											
PARAM											
ALPHA											
ON											
CT											
CA											
CLM											
CYN											
CEL											
CL											
CD											
L/D											
PARAM											
ALPHA											
ON											
CT											
CA											
CLM											
CYN											
CEL											
CL											
CD											
L/D											
PARAM											
ALPHA											
ON											
CT											
CA											
CLM											
CYN											
CEL											
CL											
CD											
L/D											
PARAM											
ALPHA											
ON				</							

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TABULATED JUNE ARC 3.5-169-711

ANES 3.5-169 1A10 09 T10 ATZ PLANE OFF

(NS7002) (28 NOV 73)

## REFERENCE DATA

REF = 2890.0000 88.17.  
 LREF = 1290.0000 IN.  
 BREF = 936.6800 IN.  
 SCALE = .0100

WARP = 1076.4000 IN.  
 WARP = .0000 IN.  
 ZARP = 400.0000 IN.

## PARAMETRIC DATA

BETA = 5.000 AILRON = .000  
 ELEVON = .000 SOFLAP = .000  
 SPDRK = .000 RUDDER = .000

RUN NO. 0/ 0 RVAL = 2.37 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.280	-1.10	-0.06013	-1.16291	.17932	.03937	.05904	-.02304	-.05979	.17543	-.30986
5.280	5.108	.06267	-1.4218	.15707	-.00997	.04669	-.02126	.04843	.16203	.27875
5.280	9.893	.16683	-1.3218	.14344	-.00969	.03851	-.02072	.14030	.16948	.77852
5.280	14.980	.29331	-1.2822	.12927	-.00980	.03216	-.02105	.24993	.20069	1.18672
5.280	20.084	.42996	-1.2280	.12131	-.13296	.02829	-.02236	.36216	.26159	1.33781
5.280	30.044	.76590	-1.1716	.11844	-.27161	.03140	-.02753	.63370	.48999	1.22156
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 0/ 0 RVAL = 4.99 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.320	-0.066	-0.05910	-1.3144	.17141	.03106	.04634	-.01978	-.05990	.1748	-.32070
7.320	5.188	.04699	-1.3109	.14567	.00198	.03472	-.01785	.03362	.14932	.20996
7.320	9.821	.13969	-1.2085	.12977	-.02374	.02831	-.01754	.11551	.15170	.71515
7.320	15.124	.25675	-1.1612	.11904	-.00904	.02442	-.01820	.21712	.18153	1.13876
7.320	20.180	.39229	-1.1009	.11482	-.10433	.02444	-.02084	.32870	.24282	1.30565
7.320	30.025	.74373	-1.3207	.11411	-.24850	.03333	-.02880	.56654	.47092	1.22372
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

MACH CLAMPD CMAPD CMAPD CLMAPD  
 5.280 .176807 -.057511 .038434  
 7.320 .171065 -.057756 .030689

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TABULATED SOURCE ARC 3.5-100, 1A10

ARC 3.5-100 1A10 OF T10 ATE PLANE OFF

090002 ( 28 NOV 73 )

# REFERENCE DATA

9007 = 2000.0000 IN. PT. 1000 = 1076.4000 IN.  
 1007 = 2000.0000 IN. 1000 = 1000.0000 IN.  
 2007 = 2000.0000 IN. 2000 = 400.0000 IN.  
 SCALE = .0100

# PARAMETRIC DATA

BETA = 5.000 AILRON = .000  
 ELEVON = .000 BOFLAP = .000  
 SPDRK = .000 RUDDER = .000

RUN NO. 0/ 0 RVAL = 2.37 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLM/A	CMA	XAC/L
5.200	-1.110	-0.0040	.02302	-3.9006
5.200	5.108	-0.0796	.02297	-3.4631
5.200	9.093	-0.0792	.02298	-3.4450
5.200	14.900	-0.0872	.02495	-3.4039
5.200	20.064	-0.1115	.02306	-3.7905
5.200	30.044	-0.1531	.03591	-4.2817
	GRADIENT	.0000	.0000	.0000

RUN NO. 0/ 0 RVAL = 4.99 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLM/A	CMA	XAC/L
7.320	-0.086	-0.0564	.02036	-2.7699
7.320	5.169	-0.0532	.01901	-2.6038
7.320	9.021	-0.0586	.02090	-2.8015
7.320	15.024	-0.0754	.02392	-3.1540
7.320	20.180	-0.1109	.02979	-3.7233
7.320	30.023	-0.1642	.03866	-4.2463
	GRADIENT	.0000	.0000	.0000



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TABULATED SOURCE ARC 3.5-169. 1A10

(087003) ( 21 SEP 75 )

AVES 3.5-169 1A10 CB T10 AT2 PLUME OFF

PARAMETRIC DATA

REFERENCE DATA

REF = 2880.0000 90.FT. 100P = 1076.4800 IN.  
 LREF = 1290.0000 IN. 100P = .0000 IN.  
 SPREF = 936.6800 IN. 200P = 400.0000 IN.  
 SCALE = .0100

RM/L = 3.79 GRADIENT INTERVAL = -5.00/ 5.00

MACH = 7.320	ALPHA	CN	CV	CA	CLM	CYN	CEL	CL	CD	L/D
	-.123	-.05996	-.15166	.17326	.03197	.04709	-.01978	-.05949	.17339	-.31993
	5.161	.04615	-.12902	.14732	.00311	.03473	-.01759	.03271	.18087	.20253
	9.726	.13572	-.11750	.13136	-.02392	.02813	-.01696	.11157	.15241	.06902
	14.955	.23489	-.11460	.11981	-.05645	.02466	-.01785	.21534	.18153	1.13003
	20.136	.39626	-.11784	.11603	-.10127	.02464	-.02051	.32273	.24191	1.28701
	30.074	.72961	-.12694	.11567	-.24118	.03280	-.02904	.57343	.46571	1.20947
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

(087004) ( 26 NOV 75 )

AVES 3.5-169 1A10 CB T10 AT2 PLUME OFF

PARAMETRIC DATA

REFERENCE DATA

REF = 2880.0000 90.FT. 100P = 1076.4800 IN.  
 LREF = 1290.0000 IN. 100P = .0000 IN.  
 SPREF = 936.6800 IN. 200P = 400.0000 IN.  
 SCALE = .0100

RM/L = 4.76 GRADIENT INTERVAL = -5.00/ 5.00

MACH = 7.320	ALPHA	CN	CV	CA	CLM	CYN	CEL	CL	CD	L/D
	-.112	-.04825	-.02252	.16104	.02995	.00680	-.00354	-.04894	.16113	-.28510
	5.165	.04996	-.02127	.14100	.00401	.00596	-.00323	.03696	.14491	.25905
	9.681	.14177	-.01889	.12640	-.01878	.00358	-.00278	.11834	.14840	.79664
	14.967	.23819	-.01552	.11682	-.05001	.00129	-.00272	.21741	.17873	1.21839
	20.034	.37997	-.01807	.11308	-.08863	.00274	-.00367	.31816	.25632	1.34316
	30.060	.73437	-.01684	.11291	-.25722	.00446	-.00484	.57030	.46572	1.24389
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

MACH 7.320  
 CLMFO .160811  
 CYNFO -.044284  
 CLMFO .028436

DATE 03 DEC 75

TABULATED SOURCE ARC 3.5-100, 1A10

PAGE 5

ANES 3.5-100 1A10 CD T10 ATE PLANE OFF

(997004) ( 28 NOV 75 )

REFERENCE DATA

SPOT = 2000.0000 90.FT. 1000 = 1076.4000 IN.  
LWSP = 1250.0000 IN. 1000 = .0000 IN.  
SPOT = 506.0000 IN. 2000 = 400.0000 IN.  
SCALE = .0100

PARAMETRIC DATA

BETA = .000 AILRON = .000  
ELEVON = .000 DOFLAP = .000  
SPDRK = .000 RUDDER = .000

RAVL = 4.78 GRADIENT INTERVAL = -5.00/ 5.00

MACH = 7.350  
ALPHA  
-112  
5.165  
9.661  
14.967  
20.054  
30.060  
GRADIENT  
CLM/A  
-00403  
-00498  
-00577  
-00614  
-01012  
-01173  
00000  
CMA  
01764  
01937  
02113  
02209  
02621  
03607  
00000  
MAC/L  
-22955  
-25715  
-27306  
-27769  
-33679  
-44065  
00000

DATE 05 DEC 75

TABULATED SOURCE ARC 3.5-169, 1A10

AVES 3.5-169 1A10 CO T10 ATC PLANE OFF

00370005 ( 28 NOV 75 )

PAGE 6

## REFERENCE DATA

SREF = 2880.0000 50.FT. XREF = 1076.4800 IN.  
 LREF = 1280.0000 IN. YREF = .0000 IN.  
 BREF = 936.6800 IN. ZREF = 400.0000 IN.  
 SCALE = .0100

BETA = .000 AILRON = 10.000  
 ELEVON = .000 BOFLAP = .000  
 SPDRK = .000 RUDER = .000

## PARAMETRIC DATA

WACH	ALPHA	CN	CY	CA	CLN	CYN	CEL	CL	CD	L/D
5.280	-0.075	-0.0148	-0.01354	.1774	.03667	.00417	-.00006	-.05125	.17781	-.28821
5.280	5.130	.06776	-0.01084	.16366	-.00348	.00109	.00083	.05286	.16906	.31264
5.280	9.667	.17475	-0.04339	.14756	-.04086	-.00166	.00184	.14747	.17481	.84360
5.280	14.934	.30368	-0.04443	.13533	-.08218	-.00196	.00239	.28068	.20958	1.24380
5.280	20.081	.45756	-0.05379	.12565	-.12827	-.00247	.00327	.36782	.26824	1.57124
5.280	30.068	.77857	.00434	.11607	-.27694	-.00631	.00672	.61564	.49054	1.25303
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 0/ 0 RVAL = 1.92 GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	CN	CY	CA	CLN	CYN	CEL	CL	CD	L/D
7.320	-0.083	-0.04881	-0.02885	.16079	.03127	.00819	-.00221	-.04953	.18767	-.30836
7.320	5.200	.05105	-0.02459	.14027	.00323	.00639	-.00169	.05812	.14432	.26417
7.320	9.771	.14258	-0.02120	.12582	-.01941	.00368	-.00078	.11916	.14819	.80410
7.320	15.011	.25970	-0.01855	.11865	-.03202	.00110	-.00001	.22162	.17994	1.22611
7.320	20.115	.39096	-0.01480	.11265	-.09331	-.00042	.00020	.32799	.24010	1.36608
7.320	30.100	.74748	-0.01681	.11329	-.24431	.00207	.00217	.59987	.47289	1.24758
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 0/ 0 RVAL = 5.78 GRADIENT INTERVAL = -5.00/ 5.00

WACH CAAPD CMAPD CLMAPD  
 5.280 .17758 -.048739 .036104  
 7.320 .16320 -.048045 .030820

DATE 03 DEC 73

TABULATED SOURCE ARC 3.5-100. 1A10

ARC 3.5-100 1A10 C9 T10 AT2 PLANE OFF

(P87005) ( 28 NOV 73 )

REFERENCE DATA

REF = 2000.0000 20.0 FT.  
 LREF = 1290.0000 IN.  
 REF = 936.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

BETA = .000  
 ELEVON = .000  
 SPDRK = .000  
 ATLRON = 10.000  
 BDFLAP = .000  
 RUDDER = .000

RUN NO. 0/ 0 RWL = 1.92 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLWA	CWA	YAC/L
5.260	-.075	-.00754	.02283	-.33016
5.260	5.130	-.00807	.02307	-.34660
5.260	9.667	-.00811	.02437	-.33260
5.260	14.934	-.00762	.02489	-.31675
5.260	20.081	-.01102	.02829	-.36942
5.260	30.068	-.01682	.03707	-.45372
	GRADIENT	.00000	.00000	.00000

RUN NO. 0/ 0 RWL = 5.78 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CLWA	CWA	YAC/L
7.320	-.093	-.00482	.01691	-.23431
7.320	5.200	-.00511	.01933	-.26449
7.320	9.771	-.00570	.02104	-.27066
7.320	15.011	-.00680	.02343	-.29454
7.320	20.115	-.01084	.02924	-.37069
7.320	30.101	-.01696	.03900	-.43496
	GRADIENT	.00000	.00000	.00000

DATE 05 DEC 75

TABULATED SOURCE ARC 3.5-100, 1A10

PAGE 8

AVES 3.5-100 1A10 CO T10 A72 PLUME OFF

087064 ( 28 NOV 75 )

## REFERENCE DATA

REF = 2400.0000 58.17. 100P = 1076.4800 IN.  
 LREF = 1250.0000 IN. 700P = .0000 IN.  
 SREF = 536.6000 IN. 200P = 400.0000 IN.  
 SCALE = .0100

## PARAMETRIC DATA

ALPHA = .000 AILRON = .000  
 ELEVON = .000 BOFLAP = .000  
 SPOBRK = .000 RUDDER = .000

RUN NO. 0/ 0 RVAL = 1.85 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.200	-9.500	-0.0815	.27782	.19193	.03280	-.10061	.03823	-.05815	.19193	-.24006
5.200	-7.450	-.06176	.20729	.18796	.03631	-.07461	.02826	-.06176	.18796	-.20959
5.200	-5.475	-.04470	.14449	.18393	.03944	-.05278	.01923	-.06470	.18393	-.32083
5.200	-3.470	-.06357	.07971	.17943	.04115	-.02991	.01083	-.06357	.17943	-.35649
5.200	-1.680	-.04811	.02447	.17706	.04246	-.01167	.00354	-.06811	.17706	-.38331
5.200	.161	-.06722	-.02391	.17466	.04163	-.00616	-.02334	-.06722	.17466	-.38080
5.200	2.272	-.06486	-.05389	.16966	.04134	.02963	-.01187	-.06486	.16966	-.34662
5.200	4.253	-.06145	-.19217	.16332	.03868	.03238	-.01944	-.06145	.16332	-.31146
5.200	6.343	-.05677	-.21444	.15515	.03570	.07940	-.02929	-.05677	.15515	-.28074
5.200	8.446	-.05397	-.28402	.15192	.03363	.09981	-.03976	-.05397	.15192	-.23136
5.200	10.502	-.05367	-.34947	.14866	.03123	.12466	-.04928	-.05367	.14866	-.20673
GRADIENT		.00061	.02990	.00110	-.00030	.01064	-.00396	.02161	.00110	.00674

RUN NO. 0/ 0 RVAL = 3.88 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.350	-9.505	-.04841	.28068	.18225	.02364	-.08530	.03391	-.04841	.18025	-.22343
7.350	-7.532	-.05042	.19550	.17431	.02715	-.08260	.02321	-.05042	.17431	-.25510
7.350	-5.603	-.05227	.13624	.16876	.02884	-.04297	.01706	-.05227	.16876	-.28837
7.350	-3.482	-.05291	.08432	.16467	.02926	-.02815	.00992	-.05291	.16467	-.31222
7.350	-1.880	-.05777	.03428	.16108	.03203	-.01120	.00399	-.05777	.16108	-.35660
7.350	.152	-.05882	-.00063	.16103	.03226	.00106	-.02122	-.05882	.16103	-.36321
7.350	2.296	-.05409	-.06309	.16295	.03031	.01716	-.00783	-.05409	.16295	-.32715
7.350	4.248	-.05087	-.11638	.16350	.02854	.03412	-.01444	-.05087	.16350	-.28831
7.350	6.347	-.05268	-.17104	.17395	.02832	.05232	-.02213	-.05268	.17395	-.27457
7.350	8.485	-.04884	-.22353	.17923	.02811	.07055	-.02978	-.04884	.17923	-.23195
7.350	10.503	-.05039	-.28432	.18695	.02513	.09128	-.03900	-.05039	.18695	-.21363
GRADIENT		.00043	-.02574	.00049	-.00018	.02769	-.00311	.00743	.00049	.00427

MACH YAC/L  
 5.200 17.457196  
 7.350 17.821884

DATE 05 DEC 73

TABULATED SOURCE ARC 3.5-16.1, 1A10

ARC 3.5-16.1 1A10 ON T10 AT2 PLANE OFF

06070077 ( 25 NOV 73 )

REFERENCE DATA

SRCP = 2880.0000 SQ.FT. WARP = 1076.4000 IN.  
 LREF = 1290.0000 IN. WARP = .0000 IN.  
 SRCP = 936.6000 IN. ZARP = 400.0000 IN.  
 SCALE = .0100

PARAMETRIC DATA

ALPHA = .000 AILRON = .000  
 ELEVON = .000 BOFLAP = .000  
 SPURK = .000 RUDDER = 10.000

RUN NO. D/ O RWL = 2.61 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
5.280	-9.561	-0.0859	.28229	.18631	.03378	-.10661	.04007	-.05899	.18631	-.25140
5.280	-7.472	-.05860	.20961	.18279	.03994	-.07904	.02938	-.05899	.18279	-.28722
5.280	-5.426	-.06377	.14580	.17957	.03916	-.05644	.02007	-.06377	.17957	-.33121
5.280	-3.487	-.06545	.08021	.17450	.04102	-.03318	.01183	-.06545	.17450	-.36554
5.280	-1.526	-.06381	.02840	.17086	.04177	-.01509	.00460	-.06381	.17086	-.38361
5.280	.166	-.06705	-.01997	.16823	.04190	.00270	-.02212	-.06705	.16823	-.39879
5.280	2.251	-.06342	-.08203	.17421	.03980	.02492	-.01018	-.06342	.17421	-.35771
5.280	4.266	-.06157	-.14240	.17688	.03851	.04670	-.01800	-.06157	.17688	-.32930
5.280	6.433	-.06684	-.20335	.18209	.03502	.06811	-.02721	-.05684	.18209	-.27940
5.280	8.454	-.05453	-.26253	.18645	.03242	.08275	-.03613	-.05453	.18645	-.24450
5.280	10.497	-.05381	-.32705	.19250	.03054	.11791	-.04575	-.05381	.19250	-.21623
GRADIENT		.00054	-.02883	.00044	-.00037	.01036	-.00386	.00054	.00044	.00527

RUN NO. D/ O RWL = 4.24 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
7.320	-9.595	-.05069	.26356	.17937	.02710	-.08798	.03549	-.05069	.17937	-.22962
7.320	-7.596	-.05413	.20056	.17374	.02927	-.06598	.02675	-.05413	.17374	-.27242
7.320	-5.481	-.05496	.14051	.16809	.03069	-.04575	.01817	-.05496	.16809	-.30415
7.320	-3.504	-.05712	.08468	.16353	.03180	-.02737	.01074	-.05712	.16353	-.33920
7.320	-1.656	-.06059	.03355	.16020	.03357	-.01225	.00474	-.06059	.16020	-.37599
7.320	.110	-.05965	-.00533	.15951	.03311	.00038	-.00037	-.05965	.15951	-.36768
7.320	2.288	-.05648	-.06231	.16233	.03145	.01666	-.00723	-.05648	.16233	-.34300
7.320	4.266	-.05394	-.11280	.16637	.03034	.03004	-.01374	-.05394	.16637	-.30947
7.320	6.367	-.05142	-.16579	.17194	.02830	.08032	-.02134	-.05142	.17194	-.27168
7.320	8.413	-.05190	-.22132	.17650	.02743	.08648	-.02886	-.05190	.17650	-.24645
7.320	10.495	-.05125	-.27949	.18392	.02638	.08941	-.03707	-.05125	.18392	-.21929
GRADIENT		.00055	-.02332	.00043	-.00027	.00770	-.00313	.00055	.00043	.00495

MACH YAC/L  
 5.280 19.216957  
 7.320 13.944568

(087008) ( 21 SEP 73 )

AVES 3.5-169 1A10 09 T10 ATE PLUME ON

## REFERENCE DATA

SHRP = 2660.0000 99.97.  
 LREF = 1250.0000 IN.  
 BREF = 936.0000 IN.  
 SCALE = .0100

ALPHA = .000  
 ELEVON = .000  
 SPDRK = .000

AILCON = .070  
 ROP LAP = .000  
 RUDDER = 10.000

## PARAMETRIC DATA

RUN NO. 0/ 0 RVAL = 2.05 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CT	CA	CLM	CYM	CBL	CL	CD	L/D
5.260	-0.6274	-0.06274	.26042	.17423	.03394	-.09826	.03687	-.06274	.17423	-.28712
5.260	-0.390	-0.0682	.20395	.17045	.03736	-.07012	.02604	-.06882	.17045	-.34174
5.260	-7.477	-0.06763	.13920	.16673	.03636	-.04639	.01677	-.06763	.16673	-.37826
5.260	-5.439	-0.06725	.07194	.16172	.03682	-.02206	.00725	-.06725	.16172	-.40557
5.260	-3.312	-0.06843	.01131	.15801	.03611	.00003	-.00146	-.06843	.15801	-.41979
5.260	-1.582	-0.06542	-.03457	.15752	.03699	.01320	-.00746	-.06542	.15702	-.41638
5.260	.167	-0.06504	-.06331	.16236	.03780	.02427	-.00996	-.06504	.16236	-.39287
5.260	2.289	-0.06236	-.13564	.16877	.03753	.03927	-.01486	-.06236	.16877	-.34961
5.260	4.251	-0.05630	-.20393	.17540	.03665	.06475	-.02436	-.05630	.17540	-.28522
5.260	6.426	-0.05708	-.26890	.18110	.03301	.08414	-.03419	-.05708	.18110	-.26101
5.260	10.313	-0.06010	-.35636	.18920	.03270	.11583	-.04396	-.06010	.18920	-.24295
GRADIENT		.00358	-.02626	.03788	-.00114	.00755	-.00275	.00358	.00098	.00727

RUN NO. 0/ 0 RVAL = 5.24 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CT	CA	CLM	CYM	CBL	CL	CD	L/D
7.320	-9.574	-.05725	.26295	.16420	.02943	-.08904	.03317	-.05725	.16420	-.27843
7.320	-7.485	-.06213	.19089	.16036	.03216	-.06175	.02434	-.06213	.16036	-.33655
7.320	-5.422	-.06244	.13650	.15494	.03250	-.04145	.01583	-.06244	.15494	-.37956
7.320	-3.310	-.06375	.08026	.15095	.03296	-.02219	.00613	-.06375	.15095	-.40974
7.320	-1.632	-.06543	.03018	.14676	.03371	-.00963	.00163	-.06543	.14676	-.43748
7.320	.151	-.06392	-.01397	.14749	.03254	.00797	-.00369	-.06392	.14749	-.43053
7.320	2.285	-.06105	-.06195	.14865	.03152	.01673	-.00709	-.06105	.14865	-.40426
7.320	4.229	-.05948	-.10757	.15456	.03064	.02682	-.01091	-.05948	.15456	-.36703
7.320	6.334	-.05912	-.16430	.15981	.02974	.04897	-.01885	-.05912	.15981	-.33400
7.320	8.471	-.05911	-.21746	.16312	.02920	.06382	-.02651	-.05911	.16312	-.30259
7.320	10.477	-.05931	-.27674	.17127	.02828	.08426	-.03443	-.05931	.17127	-.26638
GRADIENT		.00367	-.02410	.00358	-.00036	.00620	-.00245	.00367	.00156	.00825

DATE 05 DEC 73

TABULATED SOURCE ARC 3.5-169, 1A1D  
AVES 3.5-169 1A1D C8 T10 AT2 FLUME ON

(087009) ( 29 NOV 73 )

PARAMETRIC DATA

ALPHA = .000 AIRCON = .000  
ELEVON = .000 BOFLAP = .000  
SPDRK = .000 RUDDER = .000

REFERENCE DATA

SREF = 2480.0000 98.17.  
LREF = 1290.0000 IN.  
BREF = 936.6900 IN.  
SCALE = .0100

W/ O RWL = 3.54 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CBL	CL	CD	L/D
5.260	-9.595	-.06042	.26775	.16777	.03212	-.09160	.03417	-.06042	.16777	-.28764
5.260	-7.476	-.06036	.19641	.16341	.03605	-.06495	.02392	-.06036	.16341	-.35360
5.260	-5.407	-.06300	.13197	.15959	.03677	-.04373	.01538	-.06300	.15959	-.37944
5.260	-3.520	-.06323	.07171	.15682	.03675	-.02068	.00634	-.06323	.15682	-.39366
5.260	-1.675	-.06407	.01806	.15352	.03783	-.00185	-.00084	-.06407	.15352	-.41628
5.260	.148	-.06875	-.02335	.15155	.03930	.00617	-.00268	-.06875	.15155	-.44529
5.260	2.268	-.06802	-.07430	.15590	.03952	.01730	-.00990	-.06802	.15590	-.47969
5.260	4.314	-.06062	-.13384	.16071	.03536	.03979	-.01466	-.06062	.16071	-.50708
5.260	6.440	-.05763	-.20006	.16646	.03263	.06975	-.02437	-.05763	.16646	-.53677
5.260	8.325	-.05628	-.26727	.17529	.03298	.08975	-.03409	-.05628	.17529	-.56515
5.260	10.489	-.05571	-.32708	.17574	.03045	.11424	-.04353	-.05571	.17574	-.59376
GRADIENT		.07722	-.02567	.00358	-.07012	.00715	-.00242	.00050		.00434

W/ O RWL = 4.74 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ON	CY	CA	CLM	CYN	CBL	CL	CD	L/D
5.260	-9.580	-.05597	.25031	.16338	.02894	-.07600	.03115	-.05597	.16338	-.27575
7.320	-7.456	-.05920	.16795	.15926	.03058	-.05711	.02278	-.05920	.15926	-.32509
7.320	-5.399	-.05922	.12786	.15407	.03104	-.03662	.01421	-.05922	.15407	-.35873
7.320	-3.476	-.06042	.07585	.14999	.03178	-.01984	.00731	-.06042	.14999	-.39153
7.320	-1.567	-.06055	.02616	.14729	.03183	-.00481	.00162	-.06055	.14729	-.40928
7.320	.159	-.06108	-.01339	.14607	.03163	.00488	-.00237	-.06108	.14607	-.41801
7.320	2.270	-.06034	-.05772	.14846	.03138	.01195	-.00491	-.06034	.14846	-.42557
7.320	4.307	-.05669	-.11107	.15285	.02943	.02900	-.01133	-.05669	.15285	-.45266
7.320	6.406	-.05662	-.16677	.15922	.02911	.04720	-.01913	-.05662	.15922	-.48020
7.320	8.489	-.05552	-.22103	.16443	.02741	.06812	-.02667	-.05552	.16443	-.50435
7.320	10.327	-.05530	-.27707	.17018	.02666	.08522	-.03472	-.05530	.17018	-.52375
GRADIENT		.00041	-.02359	.00038	-.00027	.00589	-.00225	.00041		.00465

MACH YACA  
5.260 32.802392  
7.320 14.902700



REFERENCE DATA

SREF = 2880.0000 86.FT. XGRP = 1076.4800 IN.  
LREF = 1790.0000 IN. YGRP = .0000 IN.  
BREF = 936.8800 IN. ZGRP = 400.0000 IN.  
SCALE = .0100

PARAMETRIC DATA

BETA = .000 AILRON = .000  
ELEVON = .000 BOFLAP = .000  
SPDRK = .000 RUDDER = .000

RUN NO. 0/ 0 RVAL = 9.64 GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	ON	CY	CA	CLM	CTN	CBL	CL	CD	L/D
5.300	-9.539	-28156	-.02886	.18946	.10885	.00736	-.01253	-.24827	.23350	-1.05469
5.300	-7.395	-23531	-.03017	.18267	.09608	.01054	-.00337	-.20985	.21144	-.99247
5.300	-5.365	-18837	-.03350	.17988	.08006	.01470	-.01826	-.16929	.19064	-.86798
5.300	-3.402	-13802	-.02742	.16474	.06229	.01014	-.00382	-.12800	.17264	-.74143
5.300	-1.510	-.09254	-.02394	.15962	.04566	.00743	-.00282	-.08441	.15800	-.59955
5.300	.175	-.05368	-.02341	.14848	.03178	.00760	-.00305	-.05413	.14831	-.36497
5.300	2.258	-.04039	-.02206	.14194	.01498	.00700	-.00301	-.00997	.14165	-.07738
5.300	4.354	.03999	-.02333	.13660	-.00002	.00849	-.00342	.02951	.13924	.21192
5.300	6.517	.08616	-.02126	.13323	-.01434	.00743	-.00317	.07048	.14215	.49580
5.300	8.547	.12475	-.01766	.12726	-.02596	.00541	-.00255	.10445	.14439	.72340
5.300	10.551	.16821	-.01365	.12136	-.04044	.00224	-.00125	.14900	.15003	.95319
GRADIENT		.02303	.00031	-.00361	-.00805	-.00018	.00003	.02040	-.00427	.12460

RUN NO. 0/ 0 RVAL = 5.55 GRADIENT INTERVAL = -5.00/ 5.00

WACH	ALPHA	ON	CY	CA	CLM	CTN	CBL	CL	CD	L/D
7.320	-9.586	-26159	-.02893	.19065	.09310	.01137	-.00440	-.22569	.23451	-.96259
7.320	-7.396	-21882	-.02858	.18314	.08461	.00882	-.00346	-.19342	.20978	-.92201
7.320	-5.389	-17778	-.02774	.17453	.07221	.01053	-.00425	-.16060	.19545	-.84326
7.320	-3.418	-13841	-.02418	.16323	.05964	.00844	-.00364	-.12643	.17107	-.73976
7.320	-1.486	-.09476	-.02272	.15316	.04459	.01016	-.00447	-.09075	.15557	-.58338
7.320	.199	-.05938	-.02365	.14561	.03323	.00743	-.00334	-.05989	.14540	-.41187
7.320	2.285	-.02059	-.01549	.13776	.02332	.00181	-.00134	-.02807	.13683	-.19052
7.320	4.323	.01297	-.01814	.13309	.01475	.00385	-.00187	.02290	.13369	.02172
7.320	6.486	.04988	-.01649	.12885	.00648	.00281	-.00162	.03900	.13366	.28186
7.320	8.519	.08747	-.01587	.12231	-.00388	.00175	-.00113	.06838	.13391	.51068
7.320	10.541	.12861	-.01555	.11727	-.01540	.00385	-.00254	.10499	.13482	.75631
GRADIENT		.01934	.00124	-.00392	-.00575	-.00091	.00035	.01677	-.00482	.09952

WACH CJAFT CNAFT CLMAFD YAC/L  
5.300 .149138 -.057781 .033211 -.349371  
7.320 .146459 -.063408 .034393 -.297275

REFERENCE DATA

SREF = 2000.0000 50. FT.    WARP = 1076.4000 IN.  
 YREF = 1200.0000 IN.    YWAP = .0000 IN.  
 ZREF = 936.6000 IN.    ZWAP = 400.0000 IN.  
 SCALE = .0100

WAOH = 5.260

RNVL = 2.50    GRADIENT INTERVAL = -5.00/ 5.00

PARAMETRIC DATA

BETA = .000    AILRON = .000  
 ELEVON = .000    BOPFLAP = .000  
 SPDBRK = .000    RUDDER = .000

	ON	CY	CA	CLM	CYN	CEL	CL	CD	L/D
ALPHA	-23008	-01754	.21431	.10419	.00580	-.00214	-.20713	.24435	-.84767
-7.722	-19111	-01290	.20318	.08821	.00439	-.00202	-.16976	.22133	-.76700
-5.757	-14207	-01157	.19065	.07084	.00482	-.00197	-.12916	.19962	-.64702
-3.789	-10784	-01157	.18241	.05688	.00460	-.00191	-.09990	.18688	-.53456
-2.465	-04772	-01622	.16839	.03409	.00430	-.00173	-.04752	.16845	-.28213
-0.66	-00098	-01091	.16100	.01817	.00217	-.00147	-.00692	.16285	-.04301
2.114	.03956	-00974	.15637	.00509	.00118	-.00120	.02807	.19903	.17680
4.171	.06372	-00693	.15098	-.00368	.00007	-.00112	.06717	.15904	.42277
6.112	.12691	-01001	.14430	-.02448	.00098	-.00110	.10538	.16062	.63610
8.039	.16953	-00700	.13875	-.03965	-.00058	-.00075	.14351	.16552	.86704
9.774	.20665	-00307	.12865	-.06180	-.00078	-.00100	.25506	.20171	1.26448
15.033	.43124	-00552	.12032	-.12692	-.00036	-.00112	.36339	.26160	1.36914
27.135	.76827	-00344	.11322	-.26955	-.00254	-.00080	.60713	.48419	1.25390
30.190	.02289	.00082	-.00434	-.00827	-.00048	.00010	.01984	-.00317	.10458
GRADIENT									

WAOH    CAFTU    ONAFT    CLMFT    XAC/L  
 5.260    .169085    -.046180    .033546    -.361308

(JB7007) ( 25 OCT 73 )

AMES 3.5-169 1A10 09 T10 AT2 PLUME OFF

## REFERENCE DATA

SRP = 2690.0000 98.00 FT. 104RP = 1076.4600 IN.  
 LRP = 1290.0000 IN. 144RP = .0000 IN.  
 DRP = 936.6600 IN. 244RP = 400.0000 IN.  
 SCALE = .0100

RVL = 2.64 GRADIENT INTERVAL = -5.00/ 5.00

MAOH = 5.300

BETA	DOVOR	DCY/DR	DCA/DR	DCL/DR	DCV/DR	DOBLDR
-10.000	-.0011	.00093	-.00031	.00012	-.00065	.00021
-8.000	.00014	.00027	-.00050	-.00003	-.00045	.00012
-6.000	.00014	.00031	-.00044	-.00003	-.00043	.00011
-4.000	-.00002	-.00002	-.00046	.00000	-.00031	.00009
-2.000	.00021	.00063	-.00057	-.00007	-.00044	.00015
.000	.00003	.00043	-.00070	.00003	-.00036	.00013
2.000	.00013	.00072	-.00081	-.00014	-.00042	.00015
4.000	.00000	.00099	-.00093	-.00003	-.00036	.00018
6.000	-.00005	.00124	-.00124	-.00004	-.00061	.00025
8.000	.00010	.00200	-.00149	-.00011	-.00067	.00029
10.000	.00017	.00223	-.00160	-.00009	-.00067	.00033
GRADIENT	-.00000	.00010	-.00006	-.00001	-.00002	.00001

RVL = 4.22 GRADIENT INTERVAL = -5.00/ 5.00

MAOH = 7.300

BETA	DOVOR	DCY/DR	DCA/DR	DCL/DR	DCV/DR	DOBLDR
-10.000	-.00006	.00033	-.00009	.00012	-.00032	.00017
-8.000	-.00033	.00025	-.00009	.00021	-.00026	.00013
-6.000	-.00027	.00064	-.00003	.00018	-.00042	.00017
-4.000	-.00039	.00012	-.00011	.00025	-.00016	.00009
-2.000	-.00033	.00026	-.00009	.00018	-.00009	.00007
.000	.00001	.00024	-.00015	.00018	-.00012	.00008
2.000	-.00018	.00000	-.00007	.00010	-.00003	.00003
4.000	-.00034	.00035	-.00019	.00019	-.00011	.00007
6.000	.00010	.00061	-.00023	.00002	-.00022	.00009
8.000	-.00018	.00030	-.00009	.00009	-.00016	.00007
10.000	-.00018	.00035	-.00009	.00014	-.00016	.00008
GRADIENT	.00001	.00002	-.00001	-.00001	.00001	-.00000

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR.

(JBY006) ( 25 OCT 73 )

PARAMETRIC DATA

TABULATED SOURCE ARC 3.5-160, 1A10  
 ANES 3.5-160 1A10 CB T10 A72 PLUME ON

DATE 05 DEC 73

REFERENCE DATA

SWP = 2880.0000 SQ.FT. 100P = 1078.4800 IN.  
 LWP = 1250.0000 IN. 100P = .0000 IN.  
 SWP = 538.8800 IN. 200P = 400.0000 IN.  
 SCALE = .0100

RM/L = 2.11 GRADIENT INTERVAL = -5.00/ 5.00

WACH = 5.300	BETA	DOV/DX	DCY/DX	DCA/DX	DCL/DX	DCV/DX	DCBLR
	-10.000	-.00028	.00137	.00082	.00018	-.00082	.00029
	-8.000	-.00006	.00069	.00068	.00014	-.00057	.00023
	-6.000	-.00019	.00035	.00072	.00014	-.00056	.00015
	-4.000	-.00039	.00010	.00056	.00021	-.00018	.00008
	-2.000	-.00030	-.00027	.00045	.00009	-.00000	.00002
	.000	.00011	-.00102	.00053	-.00022	.00083	-.00044
	2.000	.00003	-.00091	.00061	-.00011	.00077	-.00044
	4.000	-.00015	-.00040	.00079	.00019	.00007	-.00008
	6.000	.00007	-.00041	.00066	.00012	.00006	.00001
	8.000	.00016	-.00033	.00102	-.00000	.00012	-.00003
	10.000	-.00027	-.00067	.00125	.00015	.00012	-.00003
	GRADIENT	.00004	-.00008	.00003	-.00001	.00006	-.00004

RM/L = 5.21 GRADIENT INTERVAL = -5.00/ 5.00

WACH = 7.300	BETA	DOV/DX	DCY/DX	DCA/DX	DCL/DX	DCV/DX	DCBLR
	-10.000	-.00009	.00140	.00005	.00002	-.00079	.00022
	-8.000	-.00026	.00065	.00012	.00014	-.00047	.00015
	-6.000	-.00033	.00082	.00010	.00014	-.00047	.00016
	-4.000	-.00031	.00046	.00009	.00011	-.00028	.00010
	-2.000	-.00048	.00025	.00013	.00018	-.00007	.00000
	.000	-.00027	-.00005	.00015	.00010	.00023	-.00013
	2.000	-.00006	-.00041	.00002	.00001	.00031	-.00023
	4.000	-.00025	.00008	.00019	.00011	-.00008	-.00001
	6.000	-.00025	.00011	.00011	.00007	-.00009	.00002
	8.000	-.00033	.00028	.00007	.00015	-.00009	.00001
	10.000	-.00033	.00003	.00012	.00018	-.00007	.00001
	GRADIENT	.00003	-.00007	.00000	-.00001	.00005	-.00002